

Distributed Neurodynamics-Based Backstepping Optimal Control for Robust Constrained Consensus of Underactuated Underwater Vehicles Fleet

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Abstract—Robust constrained formation tracking control of underactuated underwater vehicles (UUVs) fleet in 3-D space is a challenging but practical problem. To address this problem, this article develops a novel consensus-based optimal coordination protocol and a robust controller, which adopts a hierarchical architecture. On the top layer, the spherical coordinate transform is introduced to tackle the nonholonomic constraint, and then a distributed optimal motion coordination strategy is developed. As a result, the optimal formation tracking of UUVs fleet can be achieved, and the constraints are fulfilled. To realize the generated optimal commands better and, meanwhile, deal with the underactuation, at the lower-level control loop a neurodynamics-based robust backstepping controller is designed, and in particular, the issue of “explosion of terms” appearing in conventional backstepping-based controllers is avoided and control activities are improved. The stability of the overall UUVs formation system is established to ensure that all the states of the UUVs are uniformly ultimately bounded in the presence of unknown disturbances. Finally, extensive simulation comparisons are made to illustrate the superiority and effectiveness of the derived optimal formation tracking protocol.

Index Terms—Backstepping control, distributed optimal motion coordination, neurodynamics-based control, robust constrained consensus formation tracking control, underactuated underwater vehicles (UUVs) fleet.

I. INTRODUCTION

AUTONOMOUS underwater vehicle (AUV) is a sort of marine mechatronics systems, and has been used to perform various underwater missions without human intervention [1], [2], [3]. Recently, employing a group of AUVs has attracted growing attention as multiagent systems are proved to be more efficient, flexible, and cost-effective compared to a single AUV, and appear also to be more robust when faced

with disturbances or even faults. The main technical problems of this type of system lie in designing effective and efficient coordination protocols to make teams of AUVs perform tasks together in complicated marine conditions. In particular, formation control has increasingly become a focus in multiple AUVs coordination, considering its wide applications in practice. However, formation control of a group of AUVs is barely an easy thing to do, due to the nonlinear, uncertain and underactuated characteristics of dynamics, communication constraints as well as detrimental marine environments. Therefore, it is still an open and pressing problem for both societies of control and ocean engineering [4], [5], [6].

Formation control of an AUVs fleet can be roughly separated into two major portions, that is, motion coordination and control. For the former, there are several structures and strategies commonly used to coordinate the motion of multiple AUVs, such as leader-following structure [7], [8], virtual structure [9], [10], artificial potential field approach [11], etc. Besides coordination, efficient formation controllers are also key to achieving coordinated motions successfully. Following this procedure, extensive research efforts have been made in recent decades in order to synthesize effective and practical formation control protocols for AUVs fleet. In [12], an H_2/H_∞ control scheme was proposed based on the leader-following structure to ensure the optimal formation performance when disturbances and communication delays may happen. While the linear quadratic-based optimal control is fairly efficient at specific operating points, it may become restrictive when a wide range of operations is required, such as following a time-varying dynamic trajectory. In this respect, nonlinear control techniques have played an important role in the design of high-performance AUVs formation controllers and have been widely applied [13], [14], [15]. To handle the nonlinearity and underactuation, an adaptive backstepping controller (BC) was synthesized with neural network approximation to drive a group of underactuated underwater vehicles (UUVs) to create formation via a leader-following structure [13]. By incorporating a data-driven predictor, the resulting formation control strategy also addressed the communication delay [16]. System constraints fulfillment is another critical concern in designing practical controllers. As such, the model predictive control (MPC) method, as one of the optimal control techniques, was applied to resolve the AUV trajectory tracking problem subject to constraints [14]. In their studies, a

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Lyapunov-based backstepping nonlinear MPC algorithm was proposed with stability and feasibility guarantees. Based on a similar idea, receding horizon formation tracking of multiple UUVs with input limitation were addressed [15].

In addition to the nonlinearity and underactuation handling, the marine disturbances (e.g., ocean currents, waves, and winds) as well as hydrodynamic effects have significant impacts on the acquirement of robust formation performance. Toward this end, a good many research works take advantage of the sliding mode control (SMC) method due to its great robustness in tackling any matched and bounded disturbances [17], [18], [19], [20], [21]. Das et al. [17] presented an adaptive sliding mode formation control scheme to address issues of the variable added mass and communication constraints, and the overall closed-loop stability was analyzed using the Lyapunov theory. To pursue a fast transient performance, a terminal SMC method was adopted for the tracking control of UUVs [18]. While SMC-based approaches are expected to obtain good robustness against the disturbances, the chattering phenomenon stops their applications from real AUV control implementation. To overcome this drawback, a higher order SMC method was proposed for chattering-free trajectory tracking control of AUVs [19]. With the integration of a neurodynamics model, a distributed bioinspired SMC scheme was proposed to address the robust formation tracking of a fleet of fully actuated AUVs [22]. Other than SMC strategies, observer techniques are other effective alternatives to the improvement of system robustness [23]. In [20], considering both unknown disturbances and uncertain nonlinearity, an extended state observer-based integral SMC scheme was proposed for an underwater robot, and real-world experiments verified its effectiveness. Employing a similar technique, active disturbance rejection control was used in the dynamic controller design of multiple AUVs formation [24]. Cheng et al. [21] addressed robust finite-time consensus formation control of nonholonomic wheeled mobile robots, in which a finite-time observer was designed to estimate both velocities and disturbances, followed by an integral SMC controller. Likewise, based on a disturbance observer, distributed formation tracking for a group of underactuated AUVs in the horizontal plane was studied [25].

While a vital amount of research results as mentioned above have been attained to study the formation control of AUVs fleet, there still are several aspects not well considered from a practical control point of view. Most of the existing AUVs formation protocol adopts a leader-following structure [12], [13], [15], [24], [25]. In such an approach, it is assumed that all the vehicles can have access to the leaders' information, which would be rather restricted in reality. Besides, the robustness analysis in their methods is usually neglected, but it is quite crucial to maintain the feasibility of a method when faced with uncertainties. To handle the underactuation, many existing works follow a backstepping control design procedure, whereas such a method necessitates the derivative of designed virtual commands which is hard to obtain and its robustness is also limited. In terms of disturbance rejection, while SMC behaves well for certain bounded disturbances [17], [18], [19], such a method

essentially employs a high-gain strategy, thus intrinsically sensitive to the noise. The observer technique acts as an active disturbance compensation [23], yet its performance relies closely on the accurate modeling of particular disturbances, which is almost impossible for the marine situation. On the other hand, system constraints handling and performance optimization are also significant dimensions in the control design of real mechatronic systems, but barely resolved in the existing formation control literature.

Motivated by the above observations, this article is concerned with the UUVs optimal formation tracking control with unknown disturbances as well as system constraints in 3-D space. Such a problem, clearly, is of practical interest but more challenging, and has not been well studied yet. The main contributions and novelties of this article are detailed below.

- 1) A distributed robust optimal protocol is developed for the consensus formation tracking of a fleet of underwater autonomous vehicles in 3-D space. The controlled plant is subject to velocity constraints, underactuation, and unknown disturbances.
- 2) To deal with the underactuation, a spherical coordinate transformation is used, followed by a consensus-based formation tracking design. Furthermore, to achieve optimal coordination and meanwhile fulfill the constraints, an on-line motion optimization procedure is developed, and the stability, feasibility, and real-time applicability are discussed.
- 3) To realize the planned optimal commands efficiently and robustly, a neurodynamics-based BC is designed, in which the issue of "explosion of terms" is avoided and control performance is improved. Moreover, the stability and robustness properties are analyzed.
- 4) The overall stability result of the proposed UUVs formation system is derived, which shows that under some moderate conditions, all the states of the UUVs in the fleet can be steered into an ultimate bound even when faced with unknown disturbances.

The remainder of this article is arranged as follows. Some preliminaries are presented in Section II. Section III addresses the constrained consensus formation tracking problem. Neurodynamics-based robust BC shall be designed and analyzed in Section IV. Section V provides extensive numerical simulations. The conclusion is made in Section VI.

II. PRELIMINARY AND PROBLEM FORMULATION

In this section, some basic knowledge regarding the graph theory is presented, mathematical models of UUVs are described, and moreover the objective of formation tracking control of UUVs fleet is formulated.

A. Preliminary on Graph Theory

The communication topology of a UUVs fleet can be modeled by a weighted directed graph $G = \{V, E, A\}$, and each vehicle in such a system can be treated as a node. As for a simple time-invariant graph G , it can be described by the vertex set $V = \{v_1, v_2, \dots, v_N\}$, edge set $E \subseteq V \times V$, and weighted adjacency matrix $A = [a_{ij}] \in \mathbb{R}^{N \times N}$. The element v_i

in vertex set V denotes i th UUV, and the index i belongs to an index set $\Gamma = \{1, \dots, N\}$. If v_i is able to receive messages from v_j ($i \neq j$), then, say, there exists an edge pointing from v_j to v_i , that is, $(v_i, v_j) \in E$, and $a_{ij} > 0$; particularly, we call v_j a neighbor of v_i , and all such v_j form the set of neighbors of v_i , denoted by $N_i = \{j | (v_i, v_j) \in E\}$. Otherwise, there is no edge from v_j to v_i , and $a_{ij} = 0$. Moreover, we define $a_{ii} = 0$ for all $i \in \Gamma$, and out-degree $d_i = \sum_{j \in N_i} a_{ij}$ associated with the node i . Then, the degree matrix and Laplacian matrix of graph G are defined as $D = \text{diag}\{d_1, \dots, d_N\} \in \mathbb{R}^{N \times N}$ and $L = D - A$, respectively. A path in G is defined by a set of successive adjacent nodes, starting from any v_i and ending at v_j . If there is at least one path on any two nodes in graph G , then, say, graph G is connected.

In order to make the UUVs fleet move along with a prescribed trajectory together, a reference must be defined ahead of time. The availability to the information of reference trajectory for i th UUV is indicated by a parameter b_i ; that is, if UUV i have access to this information, then $b_i > 0$; otherwise, $b_i = 0$, and define matrix $B = \text{diag}(b_1, \dots, b_N)$.

Assumption 1: For the considered multi-UUV formation control network, graph G is connected, and moreover there is at least one UUV able to receive the information of reference trajectory, that is, the elements of matrix B are not all equal to zero.

Lemma 1: If Assumption 1 holds, then matrix $L + B$ is positive definite.

B. Dynamic Model and Problem Formulation

The distributed robust constrained formation tracking control of fleets of underactuated AUVs in 3-D space is addressed in this article. First, following the work of Qi and Cai [26], the kinematics of each underwater vehicle are described as follows:

$$\begin{aligned} \dot{x}_i &= \cos \theta_i \cos \psi_i u_i - \sin \psi_i v_i + \sin \theta_i \cos \psi_i w_i \\ \dot{y}_i &= \cos \theta_i \sin \psi_i u_i + \cos \psi_i v_i + \sin \theta_i \sin \psi_i w_i \\ \dot{z}_i &= -\sin \theta_i u_i + \cos \theta_i w_i \\ \dot{\theta}_i &= q_i \\ \dot{\psi}_i &= \frac{1}{\cos \theta_i} r_i \end{aligned} \quad (1)$$

where $\eta_{i1} = [x_i, y_i, z_i]^T \in \mathbb{R}^3$ and $\eta_{i2} = [\theta_i, \psi_i]^T \in \mathbb{R}^2$ represent the location and orientation of the i th vehicle ($i \in \Gamma$), respectively, expressed in the Earth-fixed frame $E^I = \{e_o^I, e_x^I, e_y^I, e_z^I\}$, and $v_{i1} = [u_i, v_i, w_i]^T \in \mathbb{R}^3$ and $v_{i2} = [q_i, r_i]^T \in \mathbb{R}^2$ are the linear and angular velocities, respectively, which is expressed in the body-fixed frame $E^B = \{e_{o,i}^B, e_{x,i}^B, e_{y,i}^B, e_{z,i}^B\}$, as shown in Fig. 1.

The dynamics of the i th vehicle are modeled by

$$\begin{aligned} m_{i1} \dot{u}_i &= m_{i2} v_i r_i - m_{i3} w_i q_i - \beta_{ui} u_i + \tau_{i1} + d_{i1} \\ m_{i2} \dot{v}_i &= -m_{i1} u_i r_i - \beta_{vi} v_i + d_{i2} \\ m_{i3} \dot{w}_i &= m_{i1} u_i q_i - \beta_{wi} w_i + d_{i3} \\ m_{i4} \dot{q}_i &= (m_{i3} - m_{i1}) u_i w_i - \beta_{qi} q_i - \beta_{bi} \sin \theta_i + \tau_{i2} + d_{i4} \\ m_{i5} \dot{r}_i &= (m_{i1} - m_{i2}) u_i v_i - \beta_{ri} r_i + \tau_{i3} + d_{i5} \end{aligned} \quad (2)$$

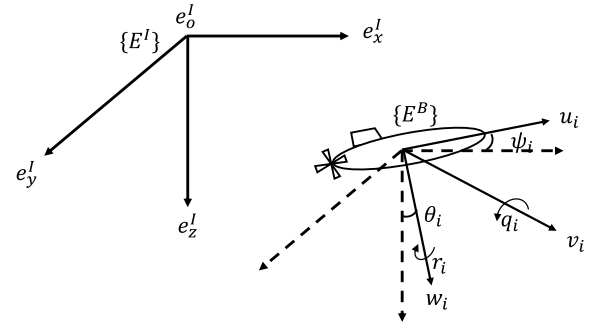


Fig. 1. Schematic of i th UUV.

where $m_{i1} = m_i - \beta_{ui}$, $m_{i2} = m_i - \beta_{vi}$, $m_{i3} = m_i - \beta_{wi}$, $m_{i4} = I_{yi} - \beta_{qi}$, and $m_{i5} = I_{zi} - \beta_{ri}$; m_i is the mass of the i th vehicle; I_{yi} and I_{zi} are the moments of inertia around the axes of $e_{y,i}^B$ and $e_{z,i}^B$, respectively; $\beta_{(\cdot)}$ is a set of hydrodynamics related terms associated with the i th vehicle. $\tau_i = [\tau_{i1}, \tau_{i2}, \tau_{i3}]^T \in \mathbb{R}^3$ is the control input, and $d_i = [d_{i1}, d_{i2}, d_{i3}, d_{i4}, d_{i5}]^T \in \mathbb{R}^5$ is the unknown disturbance acting on the i th vehicle.

Remark 1: It can be clearly seen from (2) that the velocities in sway and heave directions are underactuated; that is, these two degrees of freedom cannot be manipulated directly, and thus the control of such a system can be more challenging.

To deal with the underactuated constraint, a spherical coordinate transformation [27] is introduced as follows for i th UUV:

$$\begin{aligned} u_{ia} &= \sqrt{u_i^2 + v_i^2 + w_i^2} \\ \theta_{ia} &= \theta_i + \theta'_i \\ \psi_{ia} &= \psi_i + \psi'_i \end{aligned} \quad (3)$$

with

$$\begin{aligned} \theta'_i &= \arctan\left(-w_i / \sqrt{u_i^2 + v_i^2}\right) \\ \psi'_i &= \arctan(v_i / u_i). \end{aligned} \quad (4)$$

Since u_{ia} is positive, it is easy to verify that θ'_i and ψ'_i are both well defined in the open interval $(-\pi/2, \pi/2)$. Applying the above transformation, the kinematics of UUV i in (1) becomes

$$\begin{aligned} \dot{x}_i &= u_{ia} \cos \theta_{ia} \cos \psi_{ia} \\ \dot{y}_i &= u_{ia} \cos \theta_{ia} \sin \psi_{ia} \\ \dot{z}_i &= -u_{ia} \sin \theta_{ia} \\ \dot{\theta}_{ia} &= q_i + \dot{\theta}'_i \\ \dot{\psi}_{ia} &= r_i / \cos \theta_i + \dot{\psi}'_i. \end{aligned} \quad (5)$$

Clearly, the transformed variables $(u_{ia}, \theta_{ia}, \psi_{ia})$ now are all fully actuated.

In the considered formation tracking problem, the desired geometric formation shape of the UUVs fleet can be determined by a set of relative deviations between the vehicles i and j ($i, j \in \Gamma$), and denote as $\Delta_{ij} = [\delta_{x,ij}, \delta_{y,ij}, \delta_{z,ij}]^T \in \mathbb{R}^3$ where $\delta_{(\cdot),ij}$ is the relative deviation in a particular direction. In addition to the formation keeping, a prescribed reference trajectory requires to be tracked by the UUVs. Denote by

$\eta_{i1}^d = [x_i^d, y_i^d, z_i^d]^T \in \mathbb{R}^3$ the desired 3-D trajectory for each vehicle to track. Both variables Δ_{ij} and η_{i1}^d will be given for a particular formation tracking task. We may have the following assumptions.

Assumption 2: The reference signals, that is, $\eta_{ij}^d = [x_i^d, y_i^d, z_i^d]^T$ and its first and second derivatives $\dot{\eta}_{ij}^d = [\dot{x}_i^d, \dot{y}_i^d, \dot{z}_i^d]^T$ and $\ddot{\eta}_{ij}^d = [\ddot{x}_i^d, \ddot{y}_i^d, \ddot{z}_i^d]^T$, are all bounded for all time ($i \in \Gamma$).

Assumption 3: There is some positive constant α_1 such that the unknown disturbance d_i enforced on i th UUV is bounded by $\|d_i\| \leq \alpha_1$.

Consider a multiple UUVs formation system where the motion of each individual vehicle is described by (1) and (2), satisfying Assumptions 2 and 3. The control objective of this article is to provide a distributed robust constrained solution for each vehicle such that the following coordination motion can be achieved.

- 1) The desired formation shape (i.e., desired deviations Δ_{ij}) can be formed and maintained by UUVs.
- 2) Besides, the UUV fleet can track a predefined trajectory together even in the presence of disturbances.
- 3) The restrictions in velocities and control inputs should be realized.

III. CONSTRAINED FORMATION TRACKING CONTROL PROTOCOL DESIGN

To achieve the preceding control requirements, this section addresses the consensus-based formation tracking control problem for a fleet of UUVs, in which all neighbors' information is considered. In particular, the control commands of each vehicle are optimized through an on-line motion optimization procedure so that the planned maneuvering actions could be ensured within a practical range, meanwhile realizing the required specifications.

A. Distributed Formation Tracking Controller

Let us first define the consensus formation tracking error for i th UUV ($i \in \Gamma$) as follows:

$$e_i = \sum_{j \in N_i} a_{ij}(\eta_{i1} - \eta_{j1} - \Delta_{ij}) + b_i(\eta_{i1} - \eta_{i1}^d) \quad (6)$$

where the non-negative indicator a_{ij} shows the information interactions between vehicle i and its neighbors $j \in N_i$, and non-negative constant b_i indicates whether or not the i th vehicle can access the information of the reference trajectory. Δ_{ij} denotes the desired constant relative position between vehicles i and j . Taking the time derivative of (6), yield

$$\dot{e}_i = \sum_{j \in N_i} a_{ij}(\dot{\eta}_{i1} - \dot{\eta}_{j1}) + b_i(\dot{\eta}_{i1} - \dot{\eta}_{i1}^d). \quad (7)$$

Based on the above consensus error defined for each individual, we denote with $e = [e_1^T, e_2^T, \dots, e_N^T]^T$ the consensus formation tracking error of overall UUVs formation system and its time derivative $\dot{e} = [\dot{e}_1^T, \dot{e}_2^T, \dots, \dot{e}_N^T]^T$. Since UUVs fleet moves as a whole, the desired velocities are the same (i.e., $\dot{\eta}_{i1}^d = \dot{\eta}_1^d$ for all $i \in \Gamma$). Then, a set of (7) can be arranged

into a compact form

$$\dot{e} = (L + B)(\dot{\eta} - 1_N \dot{\eta}_1^d) \quad (8)$$

where $\dot{\eta}_1 = [\dot{\eta}_{11}^T, \dots, \dot{\eta}_{N1}^T]^T$, $1_N = [1, \dots, 1]^T$, and matrices L and B describing the communication topology of the considered formation system are detailed in the previous section (see Section II-A).

Recall that after transformation the η_{i1} -dynamics is governed by (5), and it is desirable to design control commands driving the consensus error e to zero. To this end, define

$$\rho_i = \begin{bmatrix} \rho_{ix} \\ \rho_{iy} \\ \rho_{iz} \end{bmatrix} = \begin{bmatrix} u_{ia}^{cmd} \cos \theta_{ia}^{cmd} \cos \psi_{ia}^{cmd} \\ u_{ia}^{cmd} \cos \theta_{ia}^{cmd} \sin \psi_{ia}^{cmd} \\ -u_{ia}^{cmd} \sin \theta_{ia}^{cmd} \end{bmatrix} \quad (9)$$

and a virtual control law is proposed for UUV i as follows:

$$\rho_i = -K_{i1} e_i + \dot{\eta}_1^d \quad (10)$$

where $K_{i1} \in \mathbb{R}^{3 \times 3}$ is a diagonal positive definite matrix. By (9), we then get the following control commands for UUV i :

$$\begin{aligned} u_{ia}^{cmd} &= \sqrt{\rho_{ix}^2 + \rho_{iy}^2 + \rho_{iz}^2} \\ \theta_{ia}^{cmd} &= -\arcsin\left(\frac{\rho_{iz}}{u_{ia}^{cmd}}\right) \\ \psi_{ia}^{cmd} &= \arcsin\left(\frac{\rho_{iy}}{u_{ia}^{cmd} \cos \theta_{ia}^{cmd}}\right). \end{aligned} \quad (11)$$

There may exist multiple results for the above angle commands. Observe that in many practical tasks it is unlikely to specify motions for UUVs such that the θ_i goes beyond $\pm\pi/2$, whereby the angle command θ_{ia}^{cmd} can select those values within the interval $(-\pi/2, \pi/2)$. For the determination of ψ_{ia}^{cmd} , based on the kinematic relation (9), we also have the expression $\tan(\psi_{ia}^{cmd}) = \rho_{iy}/\rho_{ix}$, by which the unique solution for ψ_{ia}^{cmd} can be determined.

On the basis of the virtual controller as proposed in (9)–(11), the η_{i1} -dynamics can be rewritten as follows:

$$\dot{\eta}_{i1} = -K_{i1} e_i + \dot{\eta}_1^d + \sigma_i \quad (12)$$

with

$$\sigma_i = \begin{bmatrix} u_{ai} \cos \theta_{ai} \cos \psi_{ai} - u_{ai}^{cmd} \cos \theta_{ai}^{cmd} \cos \psi_{ai}^{cmd} \\ u_{ai} \cos \theta_{ai} \sin \psi_{ai} - u_{ai}^{cmd} \cos \theta_{ai}^{cmd} \sin \psi_{ai}^{cmd} \\ -u_{ai} \sin \theta_{ai} + u_{ai}^{cmd} \sin \theta_{ai}^{cmd} \end{bmatrix}. \quad (13)$$

Substituting (12) into (8), yield

$$\dot{e} = (L + B)(-K_1 e + \sigma) \quad (14)$$

where $K_1 = \text{diag}(K_{11}, \dots, K_{N1})$ and $\sigma = [\sigma_1, \dots, \sigma_N]^T$.

We then state the following stability properties.

Lemma 2: The overall error dynamics of multiple UUVs consensus formation tracking as described in (14), obtained by applying the protocol (10) and (11), is input-to-state stable with respect to σ .

Proof: Propose the following Lyapunov function candidate:

$$V_1 = \frac{1}{2} e^T (L + B)^{-1} e. \quad (15)$$

It is clear from Lemma 1 that the proposed Lyapunov function is valid. Then, taking the time derivative of V_1 , along the error dynamics (14), and applying again Lemma 1 we obtain

$$\begin{aligned}\dot{V}_1 &= -e^T K_1 e + e^T \sigma \\ &\leq -\underline{k}_1 \|e\|^2 + \|e\| \|\sigma\| \\ &\leq -(\underline{k}_1 - \varepsilon_1) \|e\|^2 \quad \text{whenever} \quad \|e\| \geq \mu_1\end{aligned}\quad (16)$$

where \underline{k}_1 is the minimum eigenvalue of K_1 , ε_1 an arbitrary number within the interval $(0, \underline{k}_1)$, and $\mu_1 = (1/\varepsilon_1) \|\sigma\|$.

It is immediate from (16) that the error dynamics (14) is input-to-state stable with respect to the input σ , and in particular if $\|\sigma\| \rightarrow 0$ as $t \rightarrow \infty$, then the origin of the error system is asymptotically stable. This completes the proof. ■

Remark 2: It is noted that the virtual control law ρ_i designed for UUV i , as seen in (10), uses simply the information from its neighbors, and therefore the resulting formation protocol is said to be fully distributed.

B. On-Line Motion Optimization Procedure

It is shown above that the virtual control law proposed can lead to stable consensus formation tracking so long as the error of control commands can be made bounded. It is, however, inevitable to tune the virtual control gain K_i carefully so as to meet a satisfactory consensus formation tracking performance, and besides once the control gain is determined it cannot be changed in all future time. These features may greatly restrict the performance of the present formation plan in reality. To relax it, this section develops an on-line optimization procedure such that the control gain K_i can be optimized automatically with respect to a certain performance index; meanwhile, the constraints on UUVs' velocities can be fulfilled to improve the efficacy and security of the resulting optimal control actions.

The optimal virtual control gain of i th UUV can be obtained by solving the following constrained minimization problem at sampling time instant t_k , ($t_k > 0$):

$$\begin{aligned}\min_{K_{i,1}} J_i &= e_{i,1}^T Q e_{i,1} + \rho_{i,1}^T R_1 \rho_{i,1} + p_1 (k_{i1,1} - k_{i1,0})^2 \\ &\quad + p_2 (k_{i2,1} - k_{i2,0})^2 + p_3 (k_{i3,1} - k_{i3,0})^2 \\ &\quad + (\rho_{i,1} - \rho_{i,0})^T R_2 (\rho_{i,1} - \rho_{i,0})\end{aligned}\quad (17)$$

$$\begin{aligned}\text{s.t.} \quad e_{i,1} &= \sum_{j \in N_i} (\eta_{i1,1} - \eta_{j1,1} - \Delta_{ij}) \\ &\quad + b_i (\eta_{i1,1} - \eta_{i1,1}^d)\end{aligned}\quad (18)$$

$$\eta_{i1,1} = \eta_{i,0} + \rho_{i,1} \Delta t \quad (19)$$

$$\rho_{i,1} = -K_{i1,1} e_{i,0} + \dot{\eta}_{i,0}^d \quad (20)$$

$$\underline{\rho}_i \leq \rho_{i,1} \leq \bar{\rho}_i \quad (21)$$

$$-K_{i1,1} < 0 \quad (22)$$

$$\eta_{i1,0} = \eta_{i1}(t_k), \dot{\eta}_{i1,0}^d = \dot{\eta}_{i1}^d(t_k), \eta_{j1,1} = \eta_{j1}(t_k)$$

$$e_{i,0} = e_i(t_k), \eta_{i1,1}^d = \eta_{i1}^d(t_k), \rho_{i,0} = \rho_i(t_k)$$

$$k_{ij,0} = k_{ij}(t_k), \quad (j = 1, 2, 3) \quad (23)$$

where $Q, R_1, R_2 \in \mathbb{R}^{3 \times 3}$ and $p_1, p_2, p_3 > 0$ are weighting parameters of the suggested quadratic objective function J_i ;

$K_{i1,1} = \text{diag}(k_{i1,1}, k_{i2,1}, k_{i3,1})$ is the virtual control gain to be solved; Δt is the sampling period; $\underline{\rho}_i, \bar{\rho}_i \in \mathbb{R}^3$ are constant vectors used to restrict the speed of the vehicle; $\eta_{i1}(t_k), \dot{\eta}_{i1}^d(t_k), \eta_{j1}(t_k), e_i(t_k), \eta_{i1}^d(t_k), \rho_i(t_k)$, and $k_{ij}(t_k)$ are the initial values of the optimization problem, all of which are sampled at the time instant t_k . Notice that in this procedure some additional terms can be tailored in the objective function J_i to achieve extra goals like obstacles and collision avoidance; for example, such a term can be designed as the reciprocal of the distances between UUV and obstacles.

Remark 3: It should be noted that the above-constrained minimization problem as in (17)–(23) is implemented in a real-time manner. That is, at each sampling time instant t_k ($k > 0$), based on the measured current states and last optimizing results, the minimization problem is solved independently by each vehicle to yield the optimal solution $K_{i1,1}^*$. Using this gain, the current optimal control commands can be acquired by applying (10) and (11). At the next sampling time instant t_{k+1} , this procedure is repeated, and a new optimal solution will be calculated. This way, the control policy could be optimized dynamically in order to reach the best performance.

Remark 4: In designed constrained motion optimization, an approximate predictive model, shown in (19) with sampling time period $\Delta t > 0$, is used so as to be able to generate the one-step ahead predicted trajectories which are being optimized in terms of the consensus error, energy consumption as well as the smoothness of both processes of control and dynamic optimization, as the performance index suggested. As such, the resulting control policy could be more efficient, adaptive, and consistent than the previous one (10) where the control gain K_{i1} is commonly determined by trial-and-error. Besides, the restrictions on UUV's velocities can be fulfilled as well by the constraint (21), which effectively ensures the practicality and security of the proposed scheme.

Remark 5: The feasibility of the constrained optimization problem (17)–(23) is straightforward, as there are no state constraints imposed on UUVs. It is also clear that the stability of the resulting optimized control commands can be guaranteed, which is attributed to the introduced constraint (22). This follows directly from the result of Lemma 2. Most importantly, unlike the framework [14], since the presented problem is fully convex (i.e., both objective function and constraints are convex), there exist highly efficient programming methods (e.g., interior point methods) to solve it without affecting the real-time capability.

IV. BIOINSPIRED ROBUST CONTROLLER DESIGN

The constrained consensus formation tracking problem of UUVs fleet is addressed in the previous section. In particular, distributed optimal control commands are derived for each UUV by on-line solving a constrained minimization problem, and the stability and flexibility of the developed optimization problem are clarified. This section investigates the robust dynamic control of UUVs so that the derived optimal control commands can be realized effectively even in the presence of various unknown marine disturbances.

A. Backstepping Design Procedure

To address the underactuation issue, this section employs the backstepping design procedure, in which two auxiliary virtual controllers are defined to help design the final control laws. Before the derivations, we define the following error variables:

$$\tilde{q}_i = q_i - q_i^{cmd} \quad (24)$$

$$\tilde{r}_i = r_i - r_i^{cmd} \quad (25)$$

where q_i and r_i are the actual body frame angular velocities of UUVs in the yaw and pitch directions, respectively, and q_i^{cmd} and r_i^{cmd} are corresponding two virtual control commands to be designed.

Based on the relations introduced in (24) and (25), the (θ_{ia}, ψ_{ia}) -dynamics in (5), become

$$\dot{\theta}_{ia} = \dot{q}_i^{cmd} + \tilde{q}_i + \dot{\theta}'_i \quad (26)$$

$$\dot{\psi}_{ia} = (\dot{r}_i^{cmd} + \tilde{r}_i) / \cos \theta_i + \dot{\psi}'_i. \quad (27)$$

The virtual controllers now are designed as follows:

$$q_i^{cmd} = -k_{i\theta}(\theta_{ia} - \theta_{ia}^{cmd}) - \dot{\theta}'_i + \dot{\theta}_{ia}^{cmd}, \quad (28)$$

$$r_i^{cmd} = \cos \theta_i \left[-k_{i\psi}(\psi_{ia} - \psi_{ia}^{cmd}) - \dot{\psi}'_i + \dot{\psi}_{ia}^{cmd} \right] \quad (29)$$

where $k_{i\theta}$ and $k_{i\psi}$ are some positive constants. Let $\tilde{\theta}_{ia} = \theta_{ia} - \theta_{ia}^{cmd}$, $\tilde{\theta}'_{ia} = \dot{\theta}_{ia} - \dot{\theta}_{ia}^{cmd}$, $\tilde{\psi}_{ia} = \psi_{ia} - \psi_{ia}^{cmd}$, and $\tilde{\psi}'_{ia} = \dot{\psi}_{ia} - \dot{\psi}_{ia}^{cmd}$. Substituting the virtual control laws (28) and (29) into (26) and (27), respectively, yield

$$\dot{\tilde{\theta}}_{ia} = -k_{i\theta}\tilde{\theta}_{ia} + \tilde{q}_i \quad (30)$$

$$\dot{\tilde{\psi}}_{ia} = -k_{i\psi}\tilde{\psi}_{ia} + \tilde{r}_i / \cos \theta_i. \quad (31)$$

Taking the time derivatives in (24) and (25), together with the dynamics of q_i and r_i shown in (2), we obtain

$$\dot{\tilde{q}}_i = \left(\frac{1}{m_{i4}} \right) [(m_{i3} - m_{i1})u_i w_i - \beta_{qi}q_i - \beta_{bi} \sin \theta_i + \tau_{i2} + d_{i4}] - \dot{q}_i^{cmd} \quad (32)$$

$$\dot{\tilde{r}}_i = \left(\frac{1}{m_{i5}} \right) [(m_{i1} - m_{i2})u_i v_i - \beta_{ri}r_i + \tau_{i3} + d_{i5}] - \dot{r}_i^{cmd}. \quad (33)$$

Let $\tilde{u}_{ia} = u_{ia} - u_{ia}^{cmd}$ and $\dot{\tilde{u}}_{ia} = \dot{u}_{ia} - \dot{u}_{ia}^{cmd}$. Due to the (u_i, v_i, w_i) -dynamics in (2) as well as transformation (3), we have

$$\begin{aligned} \dot{\tilde{u}}_{ia} = & \frac{\cos \theta'_i \cos \psi'_i}{m_{i1}} (m_{i2}v_i r_i - m_{i3}w_i q_i - \beta_{ui}u_i + \tau_{i1}) \\ & - \frac{\cos \theta'_i \sin \psi'_i}{m_{i2}} (m_{i1}u_i r_i + \beta_{vi}v_i) \\ & - \frac{\sin \theta'_i}{m_{i3}} (m_{i1}u_i q_i - \beta_{wi}w_i) - \dot{u}_{ia}^{cmd} + \bar{d}_{i1} \end{aligned} \quad (34)$$

where

$$\bar{d}_{i1} = \frac{\cos \theta'_i \cos \psi'_i}{m_{i1}} d_{i1} + \frac{\cos \theta'_i \sin \psi'_i}{m_{i2}} d_{i2} + \frac{\sin \theta'_i}{m_{i3}} d_{i3}.$$

The goal now is to seek control laws for τ_{i1} , τ_{i2} , and τ_{i3} such that the error variables $\tilde{\theta}_{ia}$, $\tilde{\psi}_{ia}$, \tilde{q}_i , \tilde{r}_i , and \tilde{u}_{ia} , governed

by (30)–(34), can be brought to the origins. We provide the following lemma to achieve this purpose.

Lemma 3: Consider the error system described by (30)–(34) with Assumption 3 being satisfied. The resulting closed-loop error system is input-to-state stable, if the following control laws are employed:

$$\begin{aligned} \tau_{i1} = & -m_{i2}v_i r_i + m_{i3}w_i q_i + \beta_{ui}u_i + \frac{m_{i1}}{\cos \theta'_i \cos \psi'_i} \left(-k_{iu}\tilde{u}_{ia} \right. \\ & \left. + \dot{u}_{ia}^{cmd} + \tau_{i1}^* \right) \end{aligned} \quad (35)$$

$$\begin{aligned} \tau_{i1}^* = & \frac{\cos \theta'_i \sin \psi'_i}{m_{i2}} (m_{i1}u_i r_i + \beta_{vi}v_i) + \frac{\sin \theta'_i}{m_{i3}} (m_{i1}u_i q_i \\ & - \beta_{wi}w_i) \end{aligned} \quad (36)$$

$$\begin{aligned} \tau_{i2} = & -(m_{i3} - m_{i1})u_i w_i + \beta_{qi}q_i + \beta_{bi} \sin \theta_i - k_{iq}m_{i4}\tilde{q}_i \\ & - m_{i4}\tilde{\theta}_{ia} + m_{i4}\dot{q}_i^{cmd} \end{aligned} \quad (37)$$

$$\begin{aligned} \tau_{i3} = & -(m_{i1} - m_{i2})u_i v_i + \beta_{ri}r_i - k_{ir}m_{i5}\tilde{r}_i \\ & - m_{i5}\tilde{\psi}_{ia} / \cos \theta_i + m_{i5}\dot{r}_i^{cmd} \end{aligned} \quad (38)$$

with k_{iu} , k_{iq} , and k_{ir} being some positive constants.

Proof: Plugging the proposed control laws in (35)–(38) into (32)–(34), respectively, result in the below $(\tilde{q}_i, \tilde{r}_i, \tilde{u}_{ia})$ -error dynamic system

$$\dot{\tilde{q}}_i = -k_{iq}\tilde{q}_i - \tilde{\theta}_{ia} + \frac{1}{m_{i4}}d_{i4} \quad (39)$$

$$\dot{\tilde{r}}_i = -k_{ir}\tilde{r}_i - \tilde{\psi}_{ia} / \cos \theta_i + \frac{1}{m_{i5}}d_{i5} \quad (40)$$

$$\dot{\tilde{u}}_{ia} = -k_{iu}\tilde{u}_{ia} + \bar{d}_{i1}. \quad (41)$$

We propose the Lyapunov function candidate as follows:

$$V_{i2} = \frac{1}{2}\tilde{\theta}_{ia}^2 + \frac{1}{2}\tilde{\psi}_{ia}^2 + \frac{1}{2}\tilde{r}_i^2 + \frac{1}{2}\tilde{q}_i^2 + \frac{1}{2}\tilde{u}_{ia}^2 \quad (42)$$

and employing the time derivative of V_{i2} along the trajectories of error system in (30), (31), and (39)–(41) yields

$$\begin{aligned} \dot{V}_{i2} = & -k_{i\theta}\tilde{\theta}_{ia}^2 - k_{i\psi}\tilde{\psi}_{ia}^2 - k_{ir}\tilde{r}_i^2 - k_{iq}\tilde{q}_i^2 - k_{iu}\tilde{u}_{ia}^2 \\ & + \frac{1}{m_{i4}}\tilde{q}_i d_{i4} + \frac{1}{m_{i5}}\tilde{r}_i d_{i5} + \tilde{u}_{ia}\bar{d}_{i1}. \end{aligned} \quad (43)$$

Let $\zeta_i = [\tilde{\theta}_{ia}, \tilde{\psi}_{ia}, \tilde{r}_i, \tilde{q}_i, \tilde{u}_{ia}]^T$ and $\bar{d}_i = [0, 0, d_{i4}, d_{i5}, \bar{d}_{i1}]^T$. Equation (43) becomes

$$\dot{V}_{i2} \leq -\underline{k}_i \|\zeta_i\|^2 + \gamma_i \|\zeta_i\| \|\bar{d}_i\| \quad (44)$$

with

$$\begin{aligned} \underline{k}_i = & \inf \{ k_{i\theta}, k_{i\psi}, k_{ir}, k_{iq}, k_{iu} \} \\ \gamma_i = & \sup \left\{ \frac{1}{m_{i4}}, \frac{1}{m_{i5}}, 1 \right\}. \end{aligned}$$

Applying Assumption 3, obtain

$$\begin{aligned} \dot{V}_{i2} \leq & -\underline{k}_i \|\zeta_i\|^2 + \gamma_i \alpha_1 \|\zeta_i\| \\ \leq & -(\underline{k}_i - \lambda_1) \|\zeta_i\|^2 \quad \text{whenever} \quad \|\zeta_i\| \geq \frac{\gamma_i \alpha_1}{\lambda_1} \end{aligned} \quad (45)$$

where $0 < \lambda_1 < \underline{k}_i$. This completes the proof. ■

Remark 6: It is observed that in standard backstepping procedure while the stability of the closed-loop system can be guaranteed by Lemma 3, the derived controllers (35)–(38) rely on the time derivatives of the virtual commands introduced

in (28) and (29). Therefore, the implementation of such control laws may become much complicated with many terms resulting from the differential operation, also known as the issue of “explosion of terms,” which poses a great difficulty on applications of backstepping control design.

Remark 7: In many simulation studies, it is not uncommon to use a numerical difference to approximate the analytical one to simplify the backstepping control realization. However, such a numerical operation can be rather vulnerable to the noise which exists ubiquitously in real processes and, thus, may lead to undesired behavior or even instability of the overall formation system.

Remark 8: It is also clear from the properties of \dot{V}_{i2} as in (45) that the robustness of the BC against the disturbances is mainly dependent of the selection of the corresponding control gains (i.e., $k_{i\theta}$, $k_{i\psi}$, k_{iu} , k_{ir} , and k_{iq}). Consequently, due to the cascade connection generated by the backstepping procedure, it is easier to result in a high-gain controller, and then more likely to wind up the actuators in practice.

Based on the observations, it is necessary to remedy the above backstepping control laws so that more practical and efficient controllers can be synthesized while the nice robustness properties can be obtained without the employment of such high-gain control laws.

B. Neural Dynamics-Based Robust Control Design

To overcome the aforementioned challenges, a bioinspired solution is provided to improve the robustness properties of the conventional BCs, and it is noticed that in order to avoid the “explosion of terms,” the time derivatives of the auxiliary variables, that is, \dot{q}_i^{cmd} and \dot{r}_i^{cmd} , are regarded as the disturbances in the sequel and counteracted by the introduced neurodynamics model.

The shunting model as one of the biologically inspired neurodynamics models was initially proposed to describe the behavior of neurons in membrane with stimulus. By virtue of its beneficial properties, it has been extensively employed to develop bio-driven autonomous systems. The original shunting model of i th neuron can be described by the following switching nonlinear differential equation:

$$\dot{x}_i = -a_i x_i + (b_i - x_i) s_i^+ - (b'_i + x_i) s_i^- \quad (46)$$

where x_i represents the i th neuron activities, s_i^+ and s_i^- capture the environmental excitatory and inhibitory inputs, respectively, and a_i , b_i , and b'_i are some positive coefficients. It is noted that when the input s_i to (46) is non-negative, $s_i^+ = s_i$ and $s_i^- = 0$; otherwise, $s_i^+ = 0$ and $s_i^- = -s_i$. On this basis, shunting model (46) can be rewritten as follows:

$$\dot{x}_i = -(a_i + |s_i|)x_i + b_i s_i^+ - b'_i s_i^- \quad (47)$$

Integrated with the above model, the bioinspired robust control laws are then designed as follows:

$$\tau_{i1} = -m_{i2} v_i r_i + m_{i3} w_i q_i + \beta_{ui} u_i + \frac{m_{i1}}{\cos \theta'_i \cos \psi'_i} (-k_{iu} x_{i1} + \tau_{i1}^*) \quad (48)$$

$$\dot{x}_{i1} = -(a_{i1} + |\tilde{u}_{ia}|)x_{i1} + g_{i1}(\tilde{u}_{ia}) \quad (49)$$

$$\tau_{i2} = -(m_{i3} - m_{i1})u_i w_i + \beta_{qi} q_i + \beta_{bi} \sin \theta_i - k_{iq} m_{i4} x_{i2} - m_{i4} \tilde{\theta}_{ia} \quad (50)$$

$$\dot{x}_{i2} = -(a_{i2} + |\tilde{q}_i|)x_{i2} + g_{i2}(\tilde{q}_i) \quad (51)$$

$$\tau_{i3} = -(m_{i1} - m_{i2})u_i v_i + \beta_{ri} r_i - k_{ir} m_{i5} x_{i3} - m_{i5} \tilde{\psi}_{ia} / \cos \theta_i \quad (52)$$

$$\dot{x}_{i3} = -(a_{i3} + |\tilde{r}_i|)x_{i3} + g_{i3}(\tilde{r}_i) \quad (53)$$

where a_{i1} , a_{i2} , and a_{i3} are the positive constants to be designed, and functions g_{ij} , ($j = 1, 2, 3$) are defined as follows:

$$g_{ij}(y) = \begin{cases} b_{ij} y, & y \geq 0 \\ b'_{ij} y, & y < 0 \end{cases}$$

with b_{ij} and b'_{ij} being some positive constants. Notice that for ease of analysis in the sequel we rewrite the expression of the shunting model in our designed controller, and it is easy to verify that (49), (51), and (53) are equivalent to the primal form (46).

C. Stability Analysis

Substituting into (32)–(34) the neuro-dynamics-based BCs in (48)–(53), we obtain the following modified error subsystems of \tilde{u}_{ia} , \tilde{q}_i , and \tilde{r}_i :

$$\dot{\tilde{u}}_{ia} = -k_{iu} x_{i1} + \bar{d}'_{i1} \quad (54)$$

$$\dot{\tilde{q}}_i = -k_{iq} x_{i2} - \tilde{\theta}_{ia} + \bar{d}_{i4} \quad (55)$$

$$\dot{\tilde{r}}_i = -k_{ir} x_{i3} - \tilde{\psi}_{ia} / \cos \theta_i + \bar{d}_{i5} \quad (56)$$

Here, x_{i1} , x_{i2} , and x_{i3} are the extended states, due to the introduced shunting compensators, and are governed by (49), (51), and (53), respectively; $\bar{d}'_{i1} = \bar{d}_{i1} - \dot{u}_{ia}^{cmd}$, $\bar{d}_{i4} = \bar{d}_{i4} / m_{i4} - \dot{q}_i^{cmd}$, and $\bar{d}_{i5} = \bar{d}_{i5} / m_{i5} - \dot{r}_i^{cmd}$ are the lumped disturbances.

We shall provide a theorem to establish the stability properties of the closed-loop system with all error states ($\tilde{\theta}_{ia}$, $\tilde{\psi}_{ia}$, \tilde{u}_{ia} , \tilde{q}_i , \tilde{r}_i , x_{i1} , x_{i2} , x_{i3}) given the proposed controller (48)–(53). To facilitate the analysis, we shall follow a 2-step demonstration.

Step 1 (Input-to-State Stability of $(\tilde{\theta}_{ia}, \tilde{\psi}_{ia})$ -Subsystem): The $(\tilde{\theta}_{ia}, \tilde{\psi}_{ia})$ -dynamics are given in (30) and (31). It is clear by control theory that as long as the control gains $k_{i\theta}$ and $k_{i\psi}$ are chosen as the positive numbers, the origin of the resulting system is input-to-state stable with respect to the inputs \tilde{q}_i and $\tilde{r}_i / \cos \theta_i$. In particular, the following inequalities hold:

$$\|\tilde{\theta}_{ia}\| \leq e^{-k_{i\theta} t} \|\tilde{\theta}_{ia}(0)\| + \frac{1}{k_{i\theta}} \sup_{0 \leq \tau \leq t} \|\tilde{q}_i(\tau)\| \quad (57)$$

$$\|\tilde{\psi}_{ia}\| \leq e^{-k_{i\psi} t} \|\tilde{\psi}_{ia}(0)\| + \frac{1}{k_{i\psi}} \sup_{0 \leq \tau \leq t} \left\| \frac{\tilde{r}_i(\tau)}{\cos \theta_i(\tau)} \right\|. \quad (58)$$

Step 2 (Input-to-State Stability of $(\tilde{u}_{ia}, x_{i1}, \tilde{q}_i, x_{i2}, \tilde{r}_i, x_{i3})$ -Subsystem): Let $\xi_{i1} = [\tilde{u}_{ia}, x_{i1}]^T$, $\xi_{i2} = [\tilde{q}_i, x_{i2}]^T$, and $\xi_{i3} = [\tilde{r}_i, x_{i3}]^T$. Define $\bar{d}'_{i2} = \bar{d}_{i4} - \tilde{\theta}_{ia}$ and $\bar{d}'_{i3} = \bar{d}_{i5} - \tilde{\psi}_{ia} / \cos \theta_i$. The ξ_{ij} -dynamics ($j = 1, 2, 3$) can be rewritten into the following form:

$$\dot{\xi}_{ij} = T_j \xi_{ij} + D_j \bar{d}'_{ij} \quad (59)$$

with

$$T_1 = \begin{bmatrix} 0 & -k_{iu} \\ g_{i1} & -\bar{a}_{i1} \end{bmatrix}, \quad T_2 = \begin{bmatrix} 0 & -k_{iq} \\ g_{i2} & -\bar{a}_{i2} \end{bmatrix}$$

$$T_3 = \begin{bmatrix} 0 & -k_{ir} \\ g_{i3} & -\bar{a}_{i3} \end{bmatrix}, \quad D_1 = D_2 = D_3 = \begin{bmatrix} 1 \\ 0 \end{bmatrix}.$$

Notice that in matrix T_1 the constant g_{i1} takes value of either b_{i1} if $\bar{u}_{ia} \geq 0$, or b'_{i1} otherwise, and accordingly, same as the matrices T_2 and T_3 ; $\bar{a}_{i1} = a_{i1} + |\bar{u}_{ia}|$, $\bar{a}_{i2} = a_{i2} + |\bar{q}_i|$, and $\bar{a}_{i3} = a_{i3} + |\bar{r}_i|$.

Lemma 4: Consider subsystems as in (59). If systems matrices T_1 , T_2 , and T_3 are made Hurwitz by suitably choosing the control design parameters k_{iu} , k_{iq} , k_{ir} , a_{i1} , a_{i2} , a_{i3} , b_{i1} , b_{i2} , b_{i3} , b'_{i1} , b'_{i2} , and b'_{i3} . Then, the above subsystems are all input-to-state stable in $t \in [0, \infty)$ with respect to the inputs \bar{d}'_{i1} , \bar{d}'_{i4} , and \bar{d}'_{i5} , respectively.

Proof: Due to the fact that the subsystems (59) are all linear, their solutions can be readily obtained as follows:

$$\xi_{ij}(t) = e^{T_j t} \xi_{ij}(0) + \int_0^t e^{T_j(t-\tau)} D_j \bar{d}'_{ij} d\tau \quad (j = 1, 2, 3) \quad (60)$$

where j indicates different subsystems. Since T_j is Hurwitz, we have the inequality $\|e^{T_j t}\| \leq c_{1j} e^{-c_{2j} t}$, in which c_{1j} and c_{2j} are some positive constants and, in particular, $-c_{2j}$ is greater than the real part of the maximum eigenvalue of T_j . It is noted that while the matrices T_i may be time-varying, their Hurwitz properties can still be maintained, which can be easily verified by their analytical solutions of the eigenvalues. By this, it then follows from (60) that:

$$\begin{aligned} \|\xi_{ij}(t)\| &\leq c_{1j} e^{-c_{2j} t} \|\xi_{ij}(0)\| + \int_0^t c_{1j} e^{-c_{2j}(t-\tau)} D_j \bar{d}'_{ij} d\tau \\ &\leq c_{1j} e^{-c_{2j} t} \|\xi_{ij}(0)\| + \frac{1}{c_{2j}} \sup_{0 \leq \tau \leq t} \|\bar{d}'_{ij}\|. \end{aligned} \quad (61)$$

It can be concluded from (61) that ξ_{ij} -subsystems are all input-to-state stable with respect to the inputs \bar{d}'_{ij} in $t \in [0, \infty)$. This completes the proof. ■

By far, we have shown in an independent way that both $(\bar{\theta}_{ia}, \bar{\psi}_{ia})$ -subsystem and $(\bar{u}_{ia}, x_{i1}, \bar{q}_i, x_{i2}, \bar{r}_i, x_{i3})$ -subsystem are of input-to-state stability properties. However, it is observed that some of the above subsystems are coupled in their states. Thus, it is necessary to show the stability property of the overall coupled system. We provide the following theorem to establish this.

Theorem 1: Consider together the subsystem (30), (31), (49), and (59). Suppose that there exist some positive constant α_2 such that the lumped disturbances $\|\bar{d}'_{i1}\| + \|\bar{d}'_{i4}\| + \|\bar{d}'_{i5}\| \leq \alpha_2$, and also that the results obtained in step 1 and step 2 hold. The states of all above subsystems are input-to-state stable if $1/c_{22}k_{i\theta} < 1$ and $\alpha_\theta^2/c_{23}k_{i\psi} < 1$, where $\alpha'_\theta = \sup_{t \geq 0} 1/\cos \theta_i$.

Proof: For the ease of illustration, introduce the L_∞ norm for a signal $u(t)$ as follows:

$$\|u(t)\|_{L_\infty} = \sup_{t \geq 0} \|u(t)\|. \quad (62)$$

It is immediate from (61) that

$$\begin{aligned} \|\bar{q}_i(t)\| &\leq \|\xi_{i2}(t)\| \leq \frac{1}{c_{22}} \sup_{0 \leq \tau \leq t} \|\bar{\theta}_{ia}\| + \alpha_q + \frac{1}{c_{22}} \alpha_2 \\ \|\bar{r}_i(t)\| &\leq \|\xi_{i3}(t)\| \leq \frac{1}{c_{23}} \sup_{0 \leq \tau \leq t} \left\| \frac{\bar{\psi}_{ia}}{\cos \theta_i} \right\| + \alpha_r + \frac{1}{c_{23}} \alpha_2 \end{aligned}$$

with $\alpha_q = c_{12} \|\xi_{i2}(0)\|$, $\alpha_r = c_{13} \|\xi_{i3}(0)\|$, and applying L_∞ norm, together with (57) and (58), yield

$$\begin{aligned} \|\bar{q}_i(t)\|_{L_\infty} &\leq \frac{1}{c_{22}} \|\bar{\theta}_{ia}(t)\|_{L_\infty} + \alpha_q + \frac{1}{c_{22}} \alpha_2 \\ &\leq \frac{1}{c_{22}} \left(\frac{1}{k_{i\theta}} \|\bar{q}_i(t)\|_{L_\infty} + \alpha_\theta \right) + \alpha_q + \frac{1}{c_{22}} \alpha_2 \\ &= \frac{1}{c_{22}k_{i\theta}} \|\bar{q}_i(t)\|_{L_\infty} + \alpha_q + \frac{1}{c_{22}} (\alpha_\theta + \alpha_2) \end{aligned} \quad (63)$$

with $\alpha_\theta = \|\bar{\theta}_{ia}(0)\|$, and

$$\begin{aligned} \|\bar{\theta}_{ia}(t)\|_{L_\infty} &\leq \frac{1}{k_{i\theta}} \|\bar{q}_i(t)\|_{L_\infty} + \alpha_\theta \\ &\leq \frac{1}{c_{22}k_{i\theta}} \|\bar{\theta}_{ia}(t)\|_{L_\infty} + \alpha_\theta + \frac{1}{k_{i\theta}} \left(\alpha_q + \frac{1}{c_{22}} \alpha_2 \right). \end{aligned} \quad (64)$$

Considering the condition $1/c_{22}k_{i\theta} < 1$, we further get

$$\|\bar{q}_i(t)\|_{L_\infty} \leq \mu_2 \left(\alpha_q + \frac{1}{c_{22}} (\alpha_\theta + \alpha_2) \right) \quad (65)$$

$$\|\bar{\theta}_{ia}(t)\|_{L_\infty} \leq \mu_2 \left(\alpha_\theta + \frac{1}{k_{i\theta}} \left(\alpha_q + \frac{1}{c_{22}} \alpha_2 \right) \right) \quad (66)$$

with

$$\mu_2 = \frac{1}{1 - 1/c_{22}k_{i\theta}}.$$

Employing a similar argument, the bounds on $\|\bar{r}_i(t)\|$ and $\|\bar{\psi}_{ia}(t)\|$ can be estimated as follows:

$$\|\bar{r}_i(t)\|_{L_\infty} \leq \mu_3 \left(\alpha_r + \frac{1}{c_{23}} (\alpha'_\theta \alpha_\psi + \alpha_2) \right) \quad (67)$$

$$\|\bar{\psi}_{ia}(t)\|_{L_\infty} \leq \mu_3 \left(\alpha_\psi + \frac{\alpha'_\theta}{k_{i\psi}} \left(\alpha_r + \frac{1}{c_{23}} \alpha_2 \right) \right) \quad (68)$$

with $\alpha_\psi = \|\bar{\psi}_{ia}(0)\|$ and

$$\mu_3 = \frac{1}{1 - \alpha_\theta^2/c_{23}k_{i\psi}}.$$

As per the results obtained in steps 1 and 2, together with the derived boundedness properties of signals $\bar{\theta}_{ia}$, $\bar{\psi}_{ia}$, \bar{q}_i , and \bar{r}_i , it is readily concluded that all the states of the subsystems (30), (31), (49), and (59) are input-to-state stable. This completes the proof. ■

Remark 9: It is worthwhile noting that in Theorem 1 the boundedness conditions on \hat{u}_{ia}^{cmd} , \hat{q}_i^{cmd} , and \hat{r}_i^{cmd} are used, which is in effect easy to verify, due to the fact that the functions u_{ia}^{cmd} , r_i^{cmd} , and q_i^{cmd} defined in (11), (28), and (29), respectively, are locally Lipschitz over a domain of interest, and such a local property can be maintained by the input-to-state stability results.

Remark 10: It can be seen from the estimated bounds on error states that the robustness against the grouped disturbances depends mainly on the parameters $k_{i\theta}$, $k_{i\psi}$, and c_{2j} ($j = 1, 2, 3$); in particular, as mentioned $-c_{2j}$ are greater than the real part of the maximum eigenvalues of the T_j , which depends not only on k_u , k_q , and k_r , but also on a_{ij} , b_{ij} , and b'_{ij} . Thus, it is possible to avoid taking large values for k_u , k_q , and k_r , while achieving good robustness.

Remark 11: Note also that the shunting compensator introduced is essentially a dynamic model acting as a low-pass filter. Thus, except for the disturbance attenuation, it behaves well in terms of noise rejection and control smoothing. Besides, it is found that the outputs of the shunting compensators can be bounded upper by b_{ij} and lower by b'_{ij} , and hence the actuator saturation issue can be resolved. All of these properties show that the developed controller outperforms the conventional backstepping-based methods.

The overall stability of proposed UUVs formation system can be readily established using the following corollary.

Corollary 1: Under the assumptions of Lemma 2 and Theorem 1, the proposed formation tracking system of UUVs fleet is input-to-state stable with respect to the disturbances d_i , $i = 1, 2, \dots, N$ as input.

Proof: In Theorem 1, it is demonstrated that given the neurodynamics-based BCs, as developed in (48)–(53), the error variables, that is, $\tilde{\theta}_{ia}$, $\tilde{\psi}_{ia}$, \tilde{u}_{ia} , \tilde{q}_i , and \tilde{r}_i ($i = 1, 2, \dots, N$), can be made uniformly bounded with respect to the disturbances so long as the control parameters associated are chosen properly. Then, it is immediate by invoking Lemma 2 that the overall consensus formation tracking error e of the UUVs fleet is uniformly ultimately bounded. Also, due to (14) we conclude that \dot{e} is bounded as well, and thus together with the Assumption 2 as well as the kinematic (1) of UUVs, it is clear that all the states of the UUVs are uniformly ultimately bounded with respect to the disturbances. This completes the proof. ■

V. SIMULATION RESULTS

This section presents several numerical simulations to illustrate the effectiveness of the proposed constrained formation protocol as well as the neurodynamics-based robust BC. In the simulations, four underactuated AUVs are employed to construct a formation system. The aim of the system is that by leveraging the equipped onboard formation controller each UUV in the group can be steered to track a desired common 3-D straight line and, meanwhile, a predefined quadrilateral formation pattern can also be formed and maintained.

Since the developed formation protocol is implemented in a distributed fashion, which means only locally neighboring information can be accessed by each UUV, the communication topology associated is depicted in Fig. 2. The weights on the topological graph are selected as $a_{21} = a_{32} = a_{43} = 1$, $a_{12} = a_{23} = a_{34} = 0.8$, and $b_1 = b_2 = b_3 = b_4 = 1$. The equations of motion of UUVs are described by (1) and (2) with following parameters (every parameter follows an international standard unit): $m_i = 10$, $I_{y,i} = 3$, $I_{z,i} = 2$, $\beta_{u,i} = 6$, $\beta_{v,i} = 1.1$, $\beta_{w,i} = 1.15$, $\beta_{\dot{q},i} = 0.5$, $\beta_{\dot{r},i} = 0.45$, $\beta_{u,i} = 1$, $\beta_{v,i} = 1.1$, $\beta_{w,i} = 1.15$, $\beta_{q,i} = 0.2$, $\beta_{r,i} = 0.25$, and $\beta_{b,i} = 0.1$, ($i \in \{1, 2, 3, 4\}$). The desired 3-D path is defined as $\eta_1^d(t) = [0.7t + 5, 0.1t + 1, 5]^T$, its derivative as $\dot{\eta}_1^d(t) = [0.7, 0.1, 0]^T$. To form a prescribed formation shape, the relative positions between UUVs are given by $\Delta_{12} = [0, 10, 0]^T$, $\Delta_{21} = [0, -10, 0]^T$, $\Delta_{23} = [-10, 0, 0]^T$, $\Delta_{32} = [10, 0, 0]^T$, $\Delta_{34} = [0, -10, 0]^T$, and $\Delta_{43} = [0, 10, 0]^T$. The initial conditions of four UUVs are

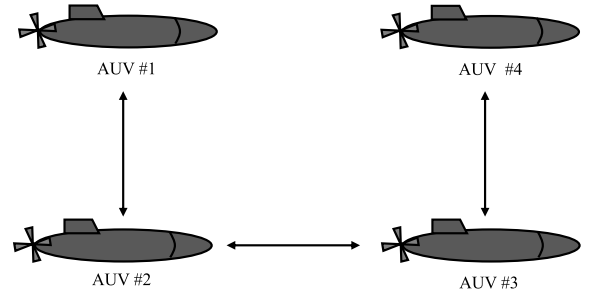


Fig. 2. Communication topology graph for the consensus formation tracking of four UUVs.

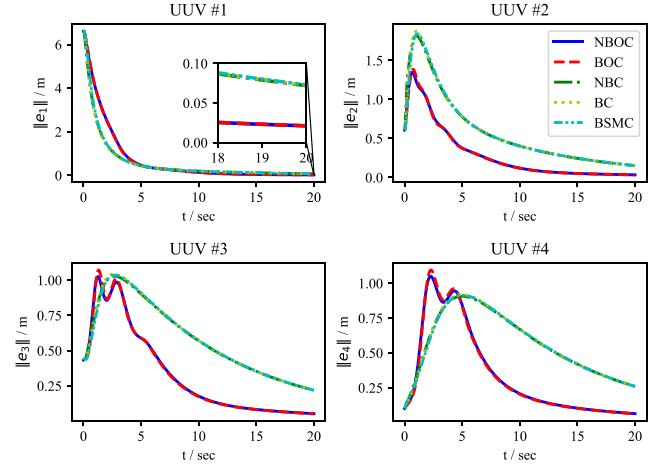


Fig. 3. Consensus formation tracking errors of four UUVs.

given as $\eta_1(0) = [0, 0, 0, 0, 0]^T$, $\eta_2(0) = [-1, -10, 0, 0, 0]^T$, $\eta_3(0) = [8.5, -10.1, 0, 0, 0]^T$, $\eta_4(0) = [8.4, -0.1, 0, 0, 0]^T$, and $v_i(0) = [0.1, 0, 0, 0, 0]^T$, ($i \in \{1, 2, 3, 4\}$). In order to make the simulation result more convincing, five controllers' performances are compared, that is, neurodynamics-based backstepping optimal controller (NBOC, i.e., the proposed approach), backstepping optimal controller (BOC, i.e., without neurodynamics), neurodynamics-based BC (NBC, i.e., without online optimization), BC, and backstepping sliding mode controller (BSMC). The control parameters of five controllers are listed in Table I.

In the first case, there are no external disturbances added to the vehicles. It can be seen from Figs. 3 and 4 that the formation tracking objective is perfectly achieved by all of the formation controllers. In particular, the controllers fitted with online optimization exhibit a faster rate of convergence as seen clearly from the behaviors of UUVs 2–4, while the control efforts needed are as nearly twice small as the NBC and BC approaches at the starting time, as shown in Fig. 5 (note that for conciseness only the UUV 1's control activities are presented, and actually the rest of vehicles behave much similar). In addition to that, another significant advantages of the online optimization are that it avoids an evident speed jump and, meanwhile, the velocity commands generated are confined within a given interval as observed in Fig. 6; in contrast, NBC, BC, and BSMC methods all yield a relatively large

TABLE I
CONTROL PARAMETERS

Parameters	NBOC	BOC	NBC	BC	BSMC
$K_{1,i}$	diag(0.3,0.3,0.3)	diag(0.3,0.3,0.3)	diag(0.6,0.6,0.6)	diag(0.6,0.6,0.6)	diag(0.6,0.6,0.6)
$K_{2,i}$	diag(10,10,10)	diag(10,10,10)	diag(10,10,10)	diag(10,10,10)	diag(20,15,15)
Q	diag(10,10,10)	diag(10,10,10)	N/A	N/A	N/A
R_1	diag(1,1,1)	diag(1,1,1)	N/A	N/A	N/A
R_2	diag(1,1,1)	diag(1,1,1)	N/A	N/A	N/A
p_1,p_2,p_3	0.1,0.1,0.1	0.1,0.1,0.1	N/A	N/A	N/A
a_i	10	N/A	10	N/A	N/A
b_i	30	N/A	30	N/A	N/A
d_i	30	N/A	30	N/A	N/A

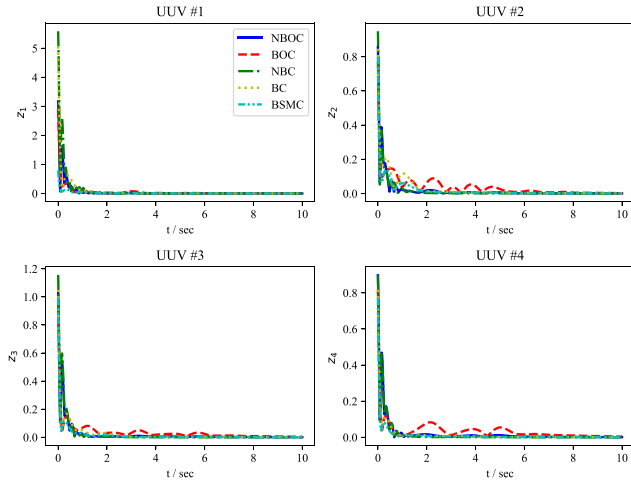


Fig. 4. Velocity tracking errors of four UUVs ($z_i = \sqrt{\hat{u}_{ia}^2 + \hat{q}_i^2 + \hat{r}_i^2}$).

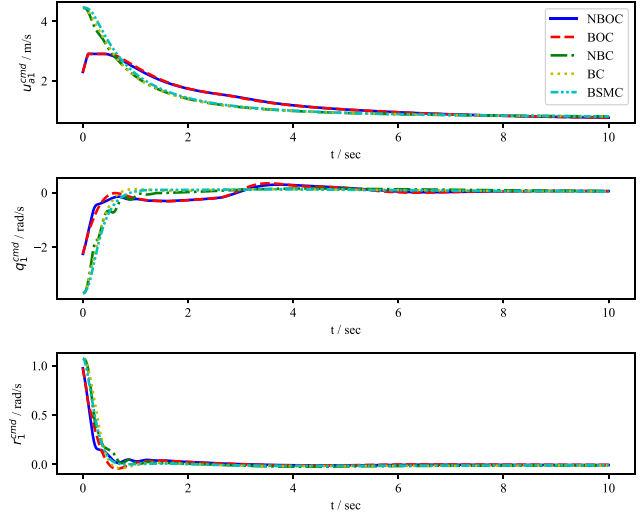


Fig. 6. Velocity commands of UUV 1.

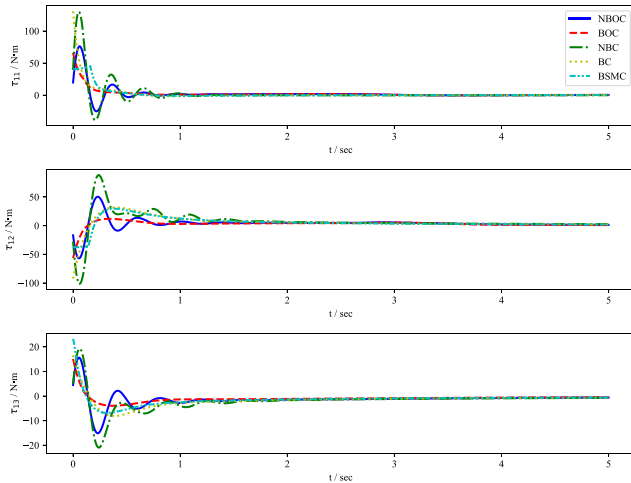


Fig. 5. Control inputs of UUV 1.

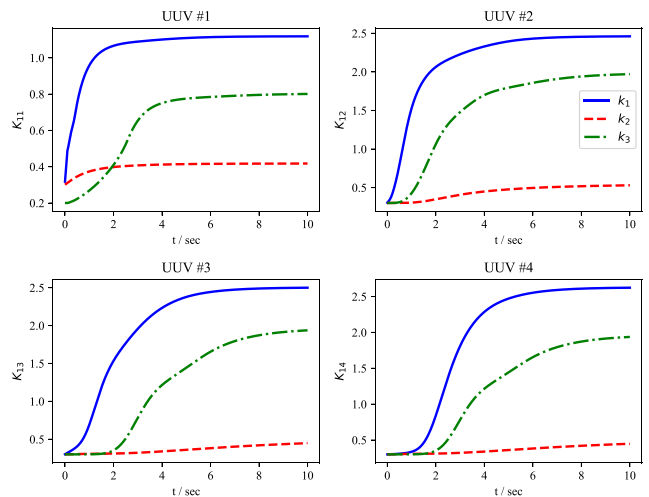


Fig. 7. Evolution of virtual control gains of four UUVs in NBOC.

velocity necessity in the beginning, due to the initial consensus errors. The properties obtained by the motion optimization are important for the controller design, since all of the real UUVs have their physical limitations on maneuvering capability. The optimization processes of the NBOC method for each UUV are presented in Fig. 7, from which an automatic adjustment for the virtual control gains can be observed.

In order to verify the robustness of the proposed formation control protocol, in the next case we inject the period exogenous disturbances into the four UUVs to simulate the influence of the ocean waves and currents. The disturbances applied are described by $d_i = [3.1 \sin(t), 3.1 \cos(t), 2.1 \sin(t),$

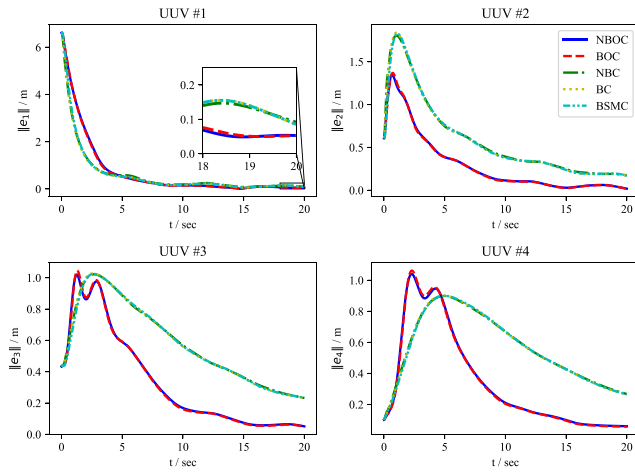


Fig. 8. Consensus formation tracking errors of four UUVs applied with disturbances.

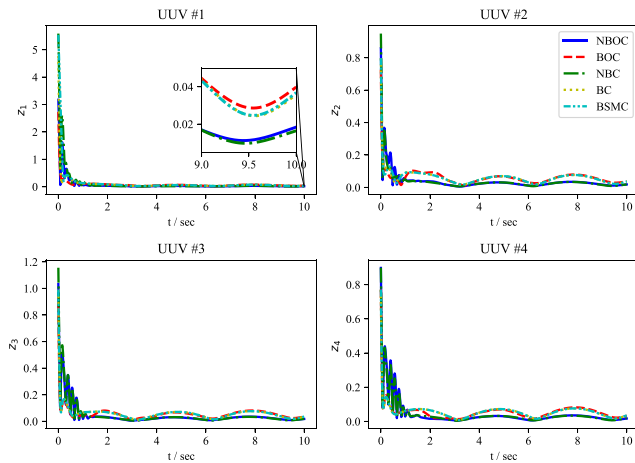


Fig. 9. Velocity tracking errors of four UUVs applied with disturbances.

$1.1 \sin(t)$, $1.1 \sin(t)$, ($i \in \{1, 2, 3, 4\}$). The formation performances of five control methods under disturbances are plotted in Figs. 8 and 9. Similar to the unperturbed situation, NBOC and BOC methods (i.e., assisted with online optimization) show a faster convergence property as seen in Fig. 8 and, meanwhile, have smaller consensus tracking errors compared to the other approaches. It implies that the optimal virtual control commands developed exhibit a better robustness property when faced with unknown disturbances. At the dynamic level, as illustrated by Fig. 9, the controllers equipped with the neuro-dynamics model render apparently smaller velocity tracking errors, thus suggesting that such methods possess good robustness in disturbance attenuation. The optimization processes of the NBOC method for each UUV are depicted in Fig. 10, all of which show a smooth convergence behavior even in the presence of disturbances. Based on the above observations, the proposed NBC nested with an online optimization procedure achieves the best formation performances over the other four methods in terms of convergence speed, steady state accuracy, disturbance attenuation, and constraint fulfillment.

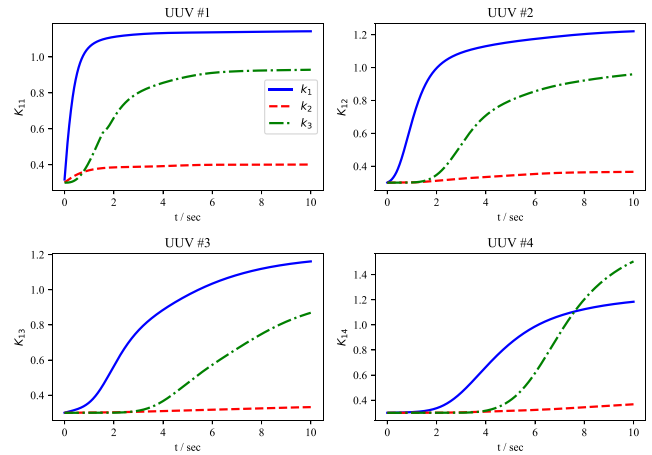


Fig. 10. Evolution of virtual control gains of four UUVs in NBOC applied with disturbances.

VI. CONCLUSION

This article addresses the robust constrained consensus formation tracking problem for a fleet of underactuated AUVs in 3-D space. A spherical coordinate transformation is introduced, based on which a novel distributed optimal formation control protocol is synthesized by iteratively solving a designed constrained optimization problem. As such, an optimal performance index can be achieved while the constraints on UUVs velocities can be fulfilled. Then, the feasibility and stability of the optimization problem are discussed. In order to realize the optimal control commands efficiently, a neuro-dynamics-based robust BC is designed. The issue of “explosion of terms” incurred in conventional BCs is addressed, and the control performance as well as robustness properties against unknown disturbances are improved. Furthermore, a rigorous stability proof of the proposed formation control method is performed to guarantee the desired performance at the theoretical level. Finally, extensive numerical simulations are carried out to further demonstrate the effectiveness and superiority of the developed UUVs formation protocol.

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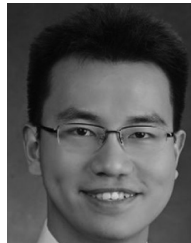
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