A SOFTWARE SYSTEM FOR INTERACTIVELY
CREATING THREE DIMENSIONAL.
FREE FORM SURFACES

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A Thesis

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McMaster University September 1980 THREE DIMENSIONAL FREE

FORM SURFACE DESIGNS

To Professor J. N. Siddall and to my wife Nadia

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Creating Three-Dimensional Free

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ABSTRACT

A system for the interactive design of free form surfaces is presented. The system is best suited for creative design based on aesthetics, experience, or a number of empirical rules. The system provides the designer with a carefully integrated set of tools which permit a rapid and convenient creation of a curved 3-D surface of any type.

In addition to the uniqueness of the overall concept, there are several innovative features. These include definition of a patch by 16 surface points only; surface modification by dragging nodes to any desired location with the light pen; and a powerful technique for defining patches using plane curves on sections.

A new method for determining NC cutter path location is suggested using the developed system.

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CHAPTER 1

INTRODUCTION

1.1 The Role of Computer Graphics in Computer Aided Design

Design is regarded as the set of activities leading from the establishment of a product requirement to the generation of the information necessary for making the product. The design process itself differs widely from industry to industry. What fundamentally distinguishes CAD as a discipline from the ad hoc use of computers in the design process is that it involves the building of systems rather than disorderly collections of programs. This leads us to the more rigorous definition of CAD as: CAD is the integration of appropriate computer hardware and software modules to create design systems for particular requirements.

The role of computer graphics in this definition of CAD is a controversial topic. On the one hand there are those who consider the terms computer graphics and CAD almost synonymous, and on the other hand, those who see no use for computer graphics in CAD. To resolve some of this controversy we should concentrate on two issues:

- What use is computer graphics in design?
- . How are designer's needs for computer graphics best provided?

Almost all engineering industries rely very heavily on drawings both for communication and as a means for information storage. The cost of preparing, codifying and storing drawings is high and is increasing rapidly.

Computer graphics has helped some branches of industry to cut their costs significantly. Plotters and computer-output on microfilm (COM) devices are today heavily used in place of manual drafting. This application is sometimes called passive or non-interactive graphics. Input of the data from which drawings are plotted may, however, involve the use of interactive graphics.

Applications of interactive graphics fall into two categories:

1. Visual Scanning of Data

By presenting data graphically, CAD systems have made good use of designer's skills in pattern recognition, and in detecting special features in a design, such as poor fairing or a lack of clearance. This skill is particularly useful in detecting errors in large input data files, such as files representing input data for a three-dimensional finite element program.

2. Input of Design Data

Computer graphic techniques can help in defining and editing complex geometrical or topological relationships in the input data for CAD programs. Examples include the specifications of car body and aircraft component shapes for

₿.

control programs, and the definition of electronic circuits for simulation purposes. Input of design data is generally facilitated by the use of graphic input devices. These are of three main types: light pens; tablets, which are a stylus working on a flat surface; and other devices, such as joysticks, tracker-ball and mouse, which do not attempt to simulate a pen or a pencil [1].

This thesis is heavily concerned with the second application of interactive graphics, which is the input of design data, although the developed system could be used also in visual scanning of data. The light pen is the main input device used, since it was proved to be in use longest and has the strongest following [2]. It has been proven in this thesis that the light pen is very easy to use especially when we use a menu of light button commands.

particular attention in this thesis - the way designers communicate with computer graphics system, or man-machine interaction. We consider this to be an important topic because, without good communication, the designer will have difficulty using the system, and the system will be less effective. A truly effective communication cannot be established without considering flow of data in both directions (man-machine and machine-man), and it seems likely that improving the man's understanding of the picture will aid him in communication his ideas back to the computer. The light pen

was chosen as an input device as it can be used to alter a picture dynamically. The operation of the programs throughout this thesis is based on the use of light buttons, words or symbols displayed on the screen which when selected with the light pen cause some appropriate program to be executed, or option to be chosen. No effort was made to design a special text command language to aid in man-machine communication through the use of the keyboard. This was due to the following reasons:

- 1. At the lowest level, the data rate from a graphical device is often much higher than from a keyboard. A light pen tracking cross can be sampled automatically on each refresh cycle.
- 2. The directness of graphics allows selection of either control items or names to be made from a set restricted to the valid possibilities. This makes graphic commands less error-prone.
- open per conscious action can be higher, implying that more information can be supplied to the program. This in turn means that commands can be concise without losing intelligibility.

From the previous general discussion, one can emphasize the value of computer graphics in engineering design.

1.2 Literature Survey on Three-Dimensional Interactive Surface Design

In three-dimensional interactive graphics, three particular issues have been the primary fields for research,

those are, 3-D mathematics, 3-D graphic systems and 3-D hardware.

1. Three-Dimensional Mathematics:

By 3-D mathematics we mean the mathematical form selected to represent the shapes required by the design, either exactly or to a sufficiently close approximation.

The first useful description of a technique to represent free-form (3-D) surfaces [5] was based on the work of Professor Coons [6]. Coons divided the surface into smaller segments, called patches. In designing the patch three different entities have to be considered - points, slopes and twists, and the user typically has to supply numbers of three different orders. (*). Apart from the initial inconvenience, the effects of modifying slope and twist vectors is confusing (the surface bulges in an unexpected way). this period (1967) Gordon [7] and Forrest [8] extended and refined the general approach to 3-D curves and surfaces using one and two-dimensional parametric cubics [9] which received considerable attention after cubic curves evolved as the most While many advantages were evident, the popular form. parametric methods were not without problems. A network of four-sided surface patches had to be formed into design surfaces. Position definitions of each corner was simple enough to achieve,

^(*) See Appendix (I)

but the definition and manipulation of parametric derivatives were another problem. It seemed that potential users of the algorithms simply had to be in command of too much mathematics to be effective.

The early 1970's saw a new approach to surface patch models by Bezier [10], which offered more intuitive and more localized control of shape. Bezier's method in more general form [11] is the B-spline curve, which allows parametric derivatives or shape control through the use of so-called 'design points'. To control shape the designer manipulates a network of 3-D design points. Unfortunately, the design points are not on the surface and, therefore, the network only remotely resembles the intended design surface (Figure (1.1)). Moreover, in order to specify a Bezier patch, sixteen spatial points must be specified together with eight tangent vectors, which appears to be rather cumbersome.

Since 1970 up til now, the Coons patch proved to be the dominant technique for creating a 3-D surface. Almost all the publications on 3-D modeling [12, 13, 14, 15, ...etc.] were more or less based on the Coons patch. With virtually all of the established algorithms for free-form curve and surface design, the user (designer) is forced to contend in some manner with parametric tangents as boundary conditions. Unfortunately no special attempt was made to tackle this problem

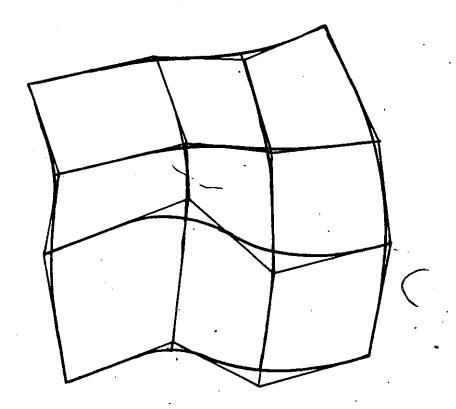


Figure (1.1) Bezier's patch defined by 16 points and 8 tangent vectors.

2. Three-Dimensional Interactive Graphic Systems:

Interactive computer graphics is usually thought to have been born with Ivan Sutherland's sketchpad in 1966 [3]. At that time, systems used very elaborate hardware, and the software was very hardware dependent [16, 17]. The system due to MacCalum [18] is noteworthy. The surfaces used were bicubic Coons patches. Design starts by causing the machine to read a tape of very approximate patch data. Thereafter design proceeds mainly by pointing with the light pen and 'dragging' part of the surface. Much work of the system is in the routines to decide what patch parameter is to be changed during dragging - e.g., a corner moved, a slope changed, etc. - following a pen hit on part of the displayed patch. The system lacks visualization aids and suffers from the readiness with which work is destroyed when errors are made.

Another interesting system is due to Armit [19, 20, 21]. By using typed commands the user creates a patch which appears on the screen with its corners labelled A, B, C, D and with its name shown. Patch modification commands are given in terms of the corner labels. There are over three hundred possible commands to allow patch modification, view modification and input/output. The significance of a language processor in 3-D design work was also demonstrated by Cordes and Brewer [22], who developed an interactive, user-oriented language called ICES/GETAM for 3-D data generation and manipulation using storage tube displays [1].

A different and more fundamental approach to language designs which can be used in 3-D work was reported by Kestner [23]. His language allows the user to deal directly with the mathematical constructs of curves and surfaces in analytical form. A user with a good and appropriate mathematical background would certainly find this system of utility.

Using a keyboard command interpreter as the primary input technique, Braid [24, 25] merged primitive objects (parallelepipeds, wedges, cylinder, etc.) to form more complicated objects. Volumes are added and/or subtracted until the desired shape is achieved. The method is not very effective in designing sculptured surfaces.

In 1975, Armit and Lemake [26] developed the ICON system for the interactive creation of data for the NASTRAN [27] suite of structured analysis programs. NASTRAN programs are used within Lloyd's Register of shipping to perform many kinds of static, elastic and dynamic analysis. The ICON system was designed to help in checking and modifying descriptions of finite element idealizations of ship hulls for input to the NASTRAN suite of structural analysis programs.

In 1976-Lacoste and Rothenberg [28] developed what they called 'Dialogue Programming', which uses interactive graphics and keyboard as input. Their objects were made from standard shapes or form elements, which were restricted to planes and cylinders. The system was coupled to EXAPT 2 for NC maching.

Pikier and Simon [29] designed a somewhat similar

System to Lacoste's system, called 'Interactive Geometrical System Using GD'71'. They used interactive drawing on the screen to input geometrical definitions (points, slopes and twists). Their program was restricted to plane geometry only. The system is currently used in interactive die design and interactive lathe programming has also been implemented.

Seifert's [30] system PROREN, said to be in use in eight mechanical engineering firms/sites in West Germany, uses input from a keyboard, arbitrary views and arbitrary intersections can be plotted, and part description is done by adding/subtracting primitive solids which could intersect and overlap.

The DUCT system designed by Welbourne, Mathews, Gossling, et al. [31, 32] is now in use in small firms for the design of patterns, moulds and dies. The program is semi-interactive and uses Bezier [11] polynomial interpolation to define surfaces and curves of intersections. A good feature of the system is that cutter paths (for ball-end cutters) are generated directly from the design.

A system to aid interactive modeling (in 3-D) of a physical object was designed by England [33]. The system allows a user to fit a bi-cubic parametric spline surface to an object by superimposing stereoscopic views of the computer surface with stereoscopic television views of the object.

Voelcker, Requicha, et al.[34] designed the PADL system. PADL (part and assembly description language) is a

language for defining solid objects via constructive solid geometry, in which complex solids may be defined as combinations of primitive solid building blocks. Unfortunately the system cannot handle sculptured surfaces.

3. Three-Dimensional Hardware:

There is now in existence an extensive family of devices to facilitate input of information to a computer.

Attempts have been made to overcome the limiting features of 2-D displays and input devices by designing new 3-D hardware. These include the spark pen, marketed by Science Accessories Corporation, 3-wire wand, and the Twinklebox.

The spark pen consists of a hand held stylus which produces small electrical sparks. These generate acoustic wave fronts which are detected by three orthogonally mounted strip microphones. Strip microphones are mounted on long tubes and are sensitive to sound along their entire length. At the time each spark is generated, a counter is started. The counter is read as the wave front is detected at each microphone, thereby determining the time taken for the wave front to reach each microphone. Knowing the speed of sound, the position of the spark can be determined.

The 3-wire wand [35] employs three shaft encoders mounted at the vertices of a triangle on the ceiling. Each shaft encoder is fitted with a spring loaded pulley, around which a wire is wrapped. The three wires are joined together on a hand grip. By maintaining a measure of the lengths of the three wires, the computer can determine the position of the

hand grip.

The spark pen and 3-wire wand both have severe limitations. Care must be taken to avoid obstructing the signals, either the sound waves or the wires. With these thoughts in mind the Twinklebox was developed by Robert Burton [36].

The Twinklebox is a device for sensing the positions of one or more small light sources. This is done using four scanners, one at each corner of the ceiling. Each scanner consists of a rotating disc around the edge of which radial slots have been cut. The axis of each disc points towards the center of the room. Consider positioning your eye behind a disc. As a slot passes your eye you see a planar slice through the room. If you were looking for a small light source, you would only see it when your eye, the slot, and the light source were all in the same plane. In the Twinklebox your eye is replaced by a photomultiplier which outputs a pulse when it sees a light. Two photomultipliers are used with each disc, subtending a right angle at the center of the disc, thereby giving two planes on which the light must lie. The four scanners therefore give eight planes to which a best fit point is found. Since only three planes are needed to define the position of a point, the system is highly redundant, thereby allowing up to five scanners to be obscured without The/Twinklebox is not entirely satisfactory due ill effect. to the significant mechanical content of the device, which has caused accuracy problems. Also, four 17" discs with

a fairly efficient siren.

Using a 3-wire wand Clark [37] developed a system for real-time 3-D surface design based on B-splines. Clark reported that the 3-D head-mounted display was "somewhat cumbersome". Position sensing mechanisms for the wand head mounted display were reported to have unacceptable accuracy and resolution problems. From a system software point of view, Clark expressed his dissatisfaction with the B-spline algorithm since control of surfaces is managed by design points off the design surface.

1.3 Motivations and Aim of the Present Thesis

From the discussion of the last two sections, it is apparent that there are three major problems facing the existing 3-D free form (sculptured) surface design systems, these are:

- 1. The definition of a free form surface is not easy for <u>designers</u>, it usually involves the definition of slopes and twists.
- 2. Input of 3-D data is a rather cumbersome task and error prone.
- 3. The dynamic interaction between the designer and the systems is lost.

A software system for the interactive design of free form surfaces, that tackles these problems, has been developed

in this thesis. The first problem was solved by the definition of a patch (parametric bi-cubic) by 16 surface points only; slopes and twists were hidden completely from the user. The second problem was solved using a powerful technique for defining patches using plane curves on sections via the light The third problem was solved by a dynamic surface modification which uses dragging of nodes (any of the 16 points defining the patch) to any desired location with the light pen. These solutions were integrated into a carefully designed software system, engineered to be conveniently and creatively interactive. In addition to the uniqueness of the overall concept, a fully developed, ship hull design program was designed as a powerful application of the developed A new technique for the calculation of the cutter path for NC end milling (ball-end) machine, for manufacturing free form surfaces, using an optimization technique, is proposed. A proposal is also made for achieving the complete integration of CAD-CAM using intersection curves; these are used as an output of the CAD system and as input to the CAM system.

CHAPTER 2

A NEW APPROACH TO 3-D SURFACE DESIGN

In Chapter I it was demonstrated that in almost all of the mathematical representation algorithms for 3-D curves and surfaces, slopes and twists (second derivatives) were involved. In this chapter, however, we will mainly be concerned with the mathematical techniques employed in the parametric bi-cubic surface patch [38, 39, 40, 13], and its definition by 16 surface points only, rather than its definition by points, slopes and twists [6], or by points not lying on the surface [11].

2.1 Three Dimensional Parametric Cubic Curves

In the parametric representation of 3-D curves the x, y and z coordinates of any point lying on the curve can be expressed as

$$x = f(u), y = g(u), z = h(u)$$

where f, g and h are different functions in the parameter u.

If we consider these functions to be polynomials of the third degree in the parameter u, we can express x, y and z as

$$x(u) = A_1 u^3 + A_2 u^2 + A_3 u + A_4$$

$$y(u) = B_1 u^3 + B_2 u^2 + B_3 u + B_4$$

$$z(u) = C_1 u^3 + C_2 u^2 + C_3 u + C_4$$
(1)

In matrix notation, Equation (1) could be rewritten

as:

$$\begin{bmatrix} x(u) \\ y(u) \\ z(u) \end{bmatrix} = (u^3 u^2 u 1) \begin{bmatrix} A_1 & A_2 & A_3 & A_4 \\ B_1 & B_2 & B_3 & B_4 \\ C_1 & C_2 & C_3 & C_4 \end{bmatrix}$$
(2)

The parameter u could take any value, but for convenience we will consider $0.0 \le u \le 1.0$, where the first point on the curve will have u = 0.0 and the last point on the curve will have u = 1, Figure (2.1). Hence, setting u = 0 and u = 1 in Equation (2) will yield

$$[x(0) \ y(0) \ z(0)]_{\bullet} = [A_4 \ B_4 \ C_4] = \overline{w}(0)$$
 (3)

and

$$\begin{bmatrix} x(1) \\ y(1) \\ z(1) \end{bmatrix} = \begin{bmatrix} 4 \\ \Sigma \\ A_{i=1} \\ 4 \\ \Sigma \\ i=1 \end{bmatrix} = \overline{w}(1)^{T}$$

$$(4)$$

Taking the parametric derivatives of Equation (2) at u = 0 and u = 1, we get

$$\begin{bmatrix} x'(0) \\ y'(0) \\ z'(0) \end{bmatrix} = \begin{bmatrix} A_3 \\ B_3 \\ C_3 \end{bmatrix} = \overline{w}'(0)^{T}$$
(5)

=

and

$$\begin{bmatrix} x'(1) \\ y'(1) \\ z'(1) \end{bmatrix} = \begin{bmatrix} 3A_1 + 2A_2 + A_3 \\ 3B_1 + 2B_2 + B_3 \\ 3C_1 + 2C_2 + C_3 \end{bmatrix} = \overline{w}'(1)^T$$

Equations (3), (4), (5) and (6) could be combined as:

$$\begin{bmatrix}
\overline{w}(0) \\
\overline{w}(1) \\
\overline{w}(0) \\
\overline{w}(1)
\end{bmatrix} = \begin{bmatrix}
0 & 0 & 0 & 1 \\
1 & 1 & 1 & 1 \\
0 & 0 & 1 & 0 \\
3 & 2 & 1 & 0
\end{bmatrix} \begin{bmatrix}
A_1 & B_1 & C_1 \\
A_2 & B_2 & C_2 \\
A_3 & B_3 & C_3 \\
A_4 & B_4 & C_4
\end{bmatrix} (7)$$

If we only consider the x coordinate we get:

$$\begin{bmatrix} x(0) \\ x(1) \\ x'(0) \\ x'(1) \end{bmatrix} = \begin{bmatrix} 0 & 0 & 0 & 1 \\ 1 & 1 & 1 & 1 \\ 0 & 0 & 1 & 0 \\ 3 & 2 & 1 & 0 \end{bmatrix} \begin{bmatrix} A_1 \\ A_2 \\ A_3 \\ A_4 \end{bmatrix} = \overline{w}_{x}(u)$$
(8)

Solving the linear system (8) for the coefficients A_1 , A_2 and A_4 produces

$$(A_1 \ A_2 \ A_3 \ A_4)^T = \overline{M} [x(0) \ x(1) \ x'(0) \ x'(1)]^T$$
 (9)

where M is the inverse of the matrix presented in Equation (8).

$$\mathbf{M} = \begin{bmatrix}
2 & -2 & 1 & 1 \\
-3 & 3 & -2 & -1 \\
0 & 0 & 1 & 0 \\
1 & 0 & 0 & 0
\end{bmatrix}$$
(10)

Substituting Equation (9) into Equation (2) for the coefficients A_1 , A_2 , A_3 , A_4 and B_1 , B_2 , B_3 , B_4 and C_1 , C_2 , C_3 , C_4 , we can totally represent the curve by

$$[x(u) \ y(u) \ z(u)] = (u^{3} \ u^{2} \ u \ 1)\overline{M} \begin{bmatrix} x(0) \ y(0) \ z(0) \\ x(1) \ y(1) \ z(1) \\ x'(0) \ y'(0) \ z'(0) \\ x'(1) \ y'(1) \ z'(1) \end{bmatrix}$$
(11)

Thus the curve can be defined if we know the coordinates of the end points x(0), y(0), z(0) and x(1), y(1), z(1), and the parametric slopes at the same end points x'(0), y'(0), z'(0) and x'(1), y'(1), z'(1).

2.2 The Parametric Bi-Cubic Surface Patch

Differential geometry [13] rests on Gauss' concept of a surface as a continuous function of two parameters u, v such that

$$x = F_1(u,v), y = F_2(u,v), z = F_3(u,v)$$

Following the same argument of representing the parametric cubic curve, one can define the parametric bi-cubic surface patch as:

[3]

$$\overline{w}(u,v) = (u^3 \ u^2 \ u \ 1)\overline{M} \ \overline{B} \ \overline{M} \ T \qquad \begin{bmatrix} v^3 \\ v^2 \\ v \\ 1 \end{bmatrix}$$
 (12)

where \overline{M} was defined in Equation (10) and \overline{B} will be called

the geometry matrix, since its elements control the shape of the surface patch, as will be shown later. The notation for elements of the E matrix is as follows:

$$B = \begin{bmatrix}
W_{00} & W_{01} & W_{00v} & W_{01v} \\
W_{10} & W_{11} & W_{10v} & W_{11v} \\
W_{00u} & W_{01u} & W_{00uv} & W_{01uv} \\
W_{10u} & W_{11u} & W_{10uv} & W_{11uv}
\end{bmatrix}$$

Figure (2.2) depicts the relation between 3-D space and the u, v parametric plane. The elements of the geometry matrix \overline{B} are explained as:

$$W_{00} = [W(u,v)]_{at u=0} = point data,$$

$$W_{00u} = [\frac{\delta w(u,v)}{\delta u}] \text{ at } u=0 = slope data,$$

$$W_{00uv} = [\frac{\delta w(u,v)}{\delta u \delta v}] \text{ at } u=0 = twist data$$

Thus the patch is now fully described by three different entities - points, slopes and twists, all related to the corner points of the patch.

2,3 Defining a Parametric Bi-Cubic Patch

Many ways have been used to define and create a parametric bi-cubic patch; and each approach must provide sufficient data to determine the 48 coefficients implies by Equation (12), i.e., 16 coefficients for each, x, y and z. Position definition of each patch corner is simple enough for

the designer to achieve, but the definition and manipulation of the parametric derivatives, contained in Equation (12), is much more difficult. From the interactive computer-aided graphics point of view, it would seem that users of such algorithms would have to be in command of too much mathematics for them to be effective. A more serious problem arises when the designer tries to modify his surface. Altering one or more parametric slope or twist will give unpredictable effects on the shape of the surface, i.e., the direct intuitive relationship between the designer and the surface design program is lost. One possible method of tackling these problems Peters 1131 is to define the surface by only spatial points. used a grid of 16 points (planar or twisted) with the corresponding "u,v" values specified in advance to define a bi-cubic surface patch. The definition of these "u,v" values in advance might cause problems for the designer because it is difficult for him to do the transformation from the 3-D space to the "u,v" plane. It is preferable if the designer specifies only the spatial points and the program calculates the corresponding "u,v" values. In this thesis a method is suggested for defining the patch by only 16 points, and estimating the "u,v" values of these points.

Using a more compact form for Equation (12), $\overline{w}(u,v)$ can be expressed as:

$$\overline{w}(u,v) = (u^3 u^2 u 1) \overline{S} \begin{bmatrix} v^3 \\ v^2 \\ v \end{bmatrix}$$
(14)

where $\overline{S} = \overline{m} \overline{B} \overline{M} T$.

Expanding Equation (14) will yield

$$(u^{3}v^{3})S_{11} + (u^{3}v^{2})S_{12} + (u^{3}v)S_{13} + (u^{3})S_{14} + (u^{2}v^{3})S_{21} + (u^{2}v^{2})S_{22} + (u^{2}v)S_{23} + (u^{2})S_{24} + (uv^{3})S_{31} + (uv^{2})S_{32} + (uv)S_{33} + (u)S_{34} + (v^{3})S_{41} + (v^{2})S_{42} + (v)S_{43} + S_{44} = W(u,v)$$
 or

 $\overline{R} \ \overline{T} = \overline{G} \tag{15}$

where \overline{R} is a 16 x 16 matrix of uv products (which are still unknown), \overline{T} is a 16 x 1 vector of the unknown \overline{S} elements, and \overline{G} is a 16 x 1 vector of the given 16 data points. Hence the result is a linear simultaneous system of equations whose unknowns are the elements of the \overline{T} vector and the elements of the \overline{R} matrix.

In some systems the u, v values are assumed [33], say u = 0, 0.25, 0.5, .75 and v = 0, 0.25 0.5, .75, and the u,v products are precomputed. This method could be useful only if the sata points are not scattered and the patch itself is very small, or if the data points are equally spaced. If the previous specifications are not satisfied, which is a more general case, a more rigorous approach should be followed.

Consider for example, Figure (2.3) in which the patch is presented with 16 spatial points lying on the surface of the patch. The points are first attached by straight lines

(dotted); then by calculating the length of these line segments, an estimate for the corresponding "u,v" values can be obtained, as shown in Table (2.1).

back and calculate the elements of 16 x 16 (R) matrix of Equation (15). Note that the solution to Equation (15) provides three S matrices, one for each of the coordinates x, y and z. The Gauss-Jordan elimination technique with maximum pivot strategy [41] handles this problem. After solving the linear system for the 48 unknown values of S, we can then go back and get the coordinates (x, y and z) for any arbitrary point lying on the patch, using the expanded form of Equation (15). This can be done by fixing the "u" value and incrementing "v" and thus obtaining lines of constant "u". Similarly we fix "v" and increment "u" to get lines of constant "v". These lines are used to draw the patch on a graphics terminal CRT.

Throughout this thesis the prescribed method for creating a bi-cubic surface patch using a grid of 16 points as the only data, has been used. It has been proven that this algorithm can handle very complicated shapes. Some typical output photographs are shown on Figures (5.25) and (5.31), showing the 16 data points lying on the patch surface. It has been also proven (as will be demonstrated in the next chapters) that the model can fit large regions of a surface, so a smaller number of patches could be used to define the whole surface.

That makes the algorithm a powerful, realistic, efficient and accurate device for QAD interactive graphics.

	`						
Point Number	u value	v value					
1	0.0	0.0					
2 .	L ₁ /(L ₁ +L ₂ +L ₃)	0.0					
3 .	$(L_1+L_2)/(L_1+L_2+L_3)$	0.0					
4	1.0	0.0					
5	0.0	L ₄ /(L ₄ +L ₁₁ +L ₁₈)					
6	L ₈ /(L ₈ +L ₉ +L ₁₀)	L ₅ /(L ₅ +L ₁₂ +L ₁₉)					
7	$(L_8+L_9)/(L_8+L_9+L_{10})$	L ₆ /(L ₆ +L ₁₃ +L ₂₀)					
, 8	1.0	L ₇ /(L ₇ +L ₁₄ +L ₂₁).					
9	0.0	$(L_4+L_{11})/(L_4+L_{11}+L_{18})$					
10	$L_{15}/(L_{15}+L_{16}+L_{17})$	(L ₁₂ +L ₅)/(L ₅ +L ₁₂ +L ₁₉).					
11	$(L_{15}^{+L}_{16})/(L_{15}^{+L}_{16}^{+L}_{17})$	$(L_{13}+L_{6})/(L_{6}+L_{13}+L_{20})$					
12	1.0	(L ₁₄ +L ₇ +L ₁₄ +L ₂₁)					
13	0.0	1.0					
14	L ₂₂ /(L ₂₂ +L ₂₃ +L ₂₄)	1.0					
15	$(L_{22}+L_{23})/(L_{22}+L_{23}+L_{24})$	1.0					
16	1.0	1.0					
L	· · · · · · · · · · · · · · · · · · ·						

Table (2.1) Estimate of the "u,v" falues

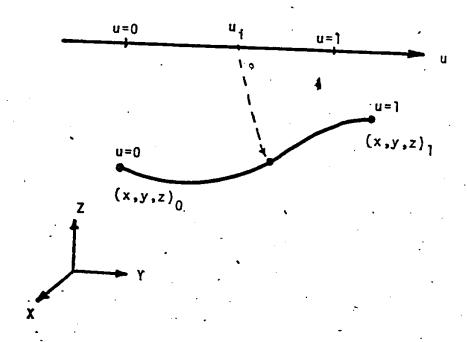
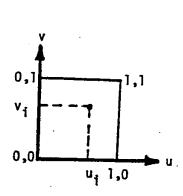


Figure (2.1) The correspondence between real x, y, z space and parametric space for parametric cubic space curve.



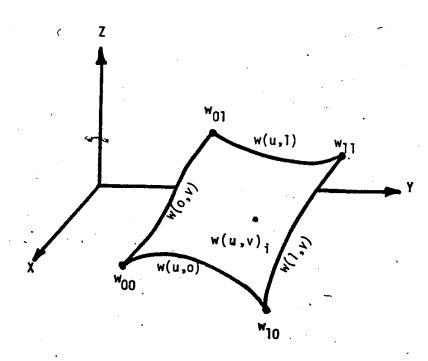


Figure (2.2) The correspondence between real x, y, z space and parametric space for parametric bi-cubic surface patch.

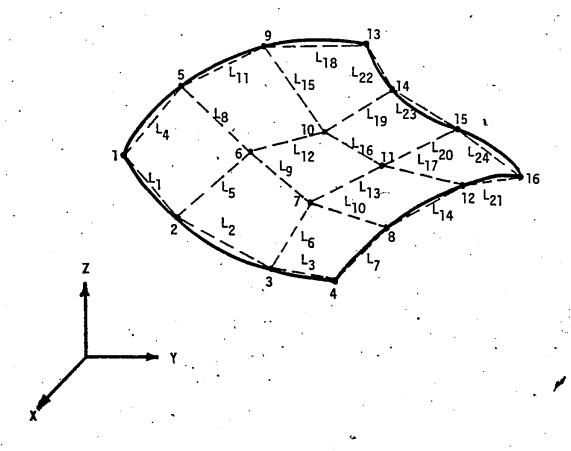


Figure (2.3) Parametric bi-cubic surface patch presented with 16 control points lying on the patch's surface.

CHAPTER 3

A SOFTWARE SYSTEM FOR INTERACTIVELY CREATING 3-D CURVED SURFACE DESIGNS

3.1 The Ideal System

The design of curved surfaces has always created difficulties for the engineer. Some types of curved surfaces, such as spheres, cones, cylinders, etc., can be represented very easily. However, this is not possible for free form shapes that are often used by designers.

In the last chapter we have investigated the mathematical modeling of such surfaces in a computer-amenable form. In this chapter we will confine our attention to the design and manipulation of these free form surfaces from the CAD graphics point of yiew.

Experience in CAD has indicated that the following software system specifications will ensure an ideal system for interactively creating 3-D curved surface designs:

- (1) Initial input need only be a very rough approximation of the desired surface, and can be defined via numerical coordinate data, digitized data from a sketch, or data generated directly by an input device, using an assembly of patches.
- (2) To define the surface, the designer need only define spatial points lying on the surface (nodes) and no slopes or twists of patches need be prescribed, i.e., no mathematical

C

knowledge of surface geometry is required.

- (3) The initial input is displayed immediately.
- (4) The assembly of patches can be displayed in any scale and orientation with or without hidden lines, so that it can be readily visualized.
- (5) Any surface can be created, including fully or partially closed surfaces, and fully smoothed or with discontinuities.
- (6) 'Adjoining patches can have their junction smoothed by the computer, using a simple command.
- (7) Transition or fairing patches between any adjacent but non-touching patches can be defined by a simple command.
- (8) One or more nodes on any patch can be relocated using an input device (e.g., light pen). The surface is immediately redefined and displayed with the new points fully refaired.
- (9) Any section through the surface can be defined and a true view obtained.
- (10) The surface is defined numerically by the computer in a manner suitable for physical duplication, or interfacing with metalworking processors, finite element processors, modelling processors, and the like.

3.2 Objectives of Software Development

The main objective of this thesis was to design a system for interactively creating 3-B curved (sculptured) surface designs, which will fulfill the previously mentioned ten specifications.

A method suitable for the interactive design of free form surfaces is presented. The method is best suited for creative design based on aesthetics, experience, or a number of empirical rules. The method provides the designer with a carefully integrated set of tools which permit a rapid and convenient creation of a curved 3-D surface of any type.

3.3 General Algorithm for Surface Design

The general objective is to provide a facility on a typical minicomputer graphics system (Figure 3.1) which enables the user to do the subjective design of 3-D curved surfaces and make the definition and modification of such surfaces as direct and simple as possible. The basic tool is a surface patch defined by space coordinates. A "patch" is simply a relatively small segment (Figure 3.2) of a curved 3-D surface. Any surface is to be built up from a collection of such patches.

The basic tool of this software system is the definition of a patch by 16 space points lying on its surface as indicated in Figure 3.2. The patch was developed in Chapter 2.

The second primary tool is the facility to select any node and relocate it interactively. The system will then resmooth the surface to fit this new point. The patch to be distorted is first identified, and it may be a sub-patch or a multiple patch. Multiple nodes may be relocated before reconstructing the surface.

The third primary tool is the facility to create a patch

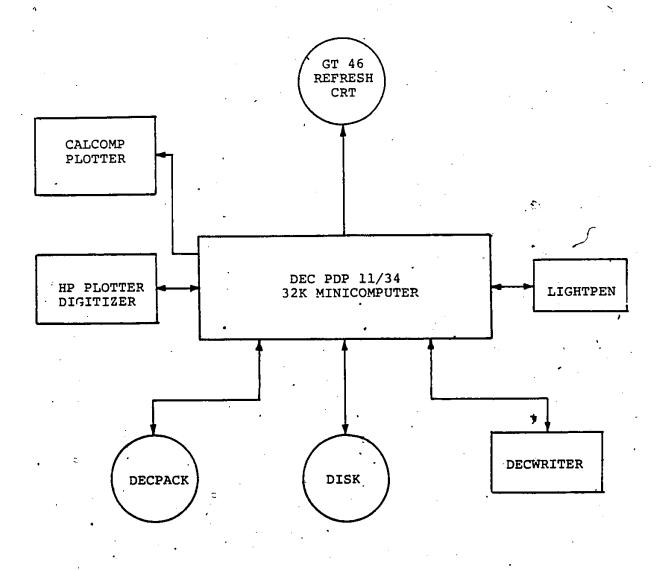


Figure 3.1 Interactive Graphic Hardware Configuration

between parallel or orthogonal sections. The user is asked to define four planes (parallel to x-y, y-z or z-x planes) and in each plane he is asked to define four points, i.e., define 16 points; and so define a patch. The program will then generate the surface patch which pass through these points. This facility is well, illustrated in section 3.3.2.

This leads to the fourth primary tool, which is the facility to create a sub-patch as shown in Figure 3.3. A sub-patch is a patch created by splitting an already defined patch.

A final primary tool in building up the patch work surface is the facility to smooth two patches along the common borders. Each patch is independently smoothed as shown in Figure 3.4. At the junction point "A" there can be an undesirable second order discontinuity. A facility is provided for designating such a point a "smooth junction", if this is desired, and one of the two patches will be so modified.

Other important facilities are also provided by the system, like rotation, scaling and obtaining surfaces of revolutions.

3.3.1 Free Form Surface Design Program

A necessary element in any CAD or CAM system is to design an efficient and economical method for describing a 3-D part geometry in a form understandable by a computer.

There are several ways to provide this description.

For example, a part can be described as an enormous number of coordinate points. This type of description, however, is

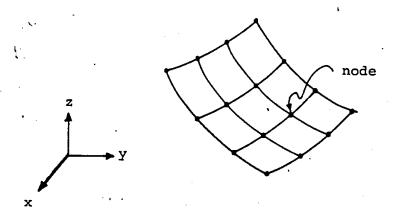


Figure 3.2 A Surface Patch Defined by 16 Points

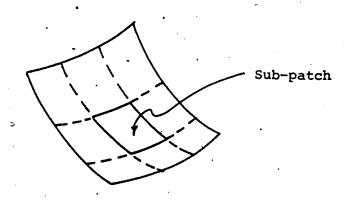
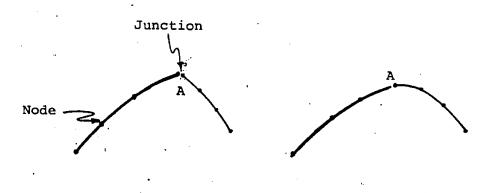


Figure 3.3 A Sub-patch



Section Through Two Adjacent Patches Figure 3.4

inefficient.

Another technique is to use the so-called "useroriented" NC programming language such as APT [42]. Such
languages can describe geometric entities, including planes,
cones, conic section and other complex surfaces, through
simple English-like statements. We will not elaborate on such
NC programming languages, but it should be noted that modeling
of sculptured surfaces is difficult using such languages. In
fact, the difficulty of modeling is one of the reasons that
relatively few dies (for metal forming) are now cut by NC.
Most of them are still made by tracer mills, where a stylus or
tracer mechanism moves along a plaster or wooden model to guide
a cutting tool that duplicates the shape in metal. The models,
in turn, are made by hand in a laborious procedure that is
anything but automated [43].

In this thesis, a completely different technique is used. Describing or actually designing 3-D geometry is done by defining selected coordinates (16 points per patch) and then having the computer blend them mathematically to generate the rest of the part configuration. In the following sections we will describe the various capabilities of the developed software system applying the previously mentioned technique.

3.3.2 Input of Initial Surface Data

Of all the input data needed for computer-aided design, the most difficult and time-consuming to produce is a description

of part geometry. For this reason, the system provides the designer with three input options.

The design process begins as soon as the designer arrives at the computer with approximate patch coordinates and a layout of patches on a piece of paper. The design process can also begin with no prior surface design at all, i.e., it can be completely generated at the computer. As soon as the designer runs the program, a menu having the three input options will be displayed on the CRT as shown in Figure 3.5.

If we consider only one patch, the designer is required to define his surface patch through the use of any of the three options of Figure 3.5. As mentioned before, a patch is defined only by 16 x, y, z coordinate points.

If the designer hits the light button command "INPUT VIA KEYBOARD" with the light pen, the program asks him to enter 16 x, y, z coordinate points defining the patch via the keyboard. The patch is then displayed immediately.

The procedure is the same with the "INPUT VIA DATA FILES" command; the program asks for the data file name containing coordinate data describing the patch, the designer enters the name, the program displays the patch.

Finally if the user hits the "INPUT VIA LIGHT PEN" command with the light pen, the program will use the following steps:

(1) The X, Y, Z axis will be displayed on the CRT together with two light buttons on the menu area, "DEFINE WORKING PLANE" and "DONE" as shown in Figure 3.6.

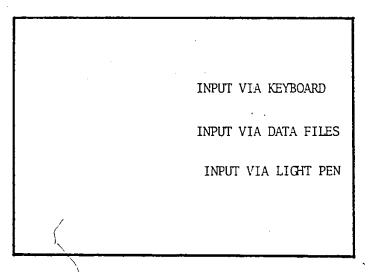


Figure (3.5) The three input options

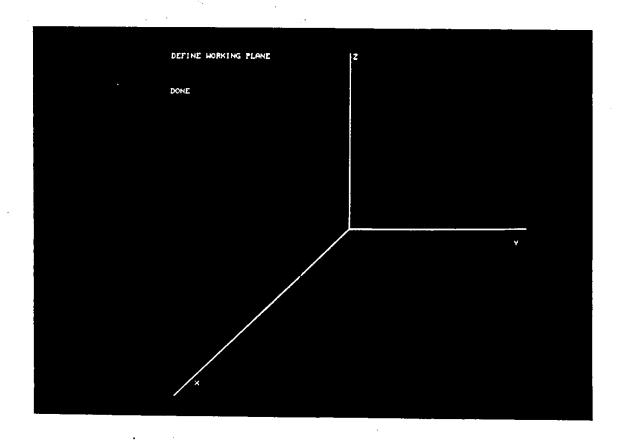


Figure (3.6) "DEFINE WORKING PLANS" and "DONE" options

- (2) To define a working plane, the designer hits the "DEFINE WORKING PLANE" option with the light pen; and the program types the following message: "POSITION THE TRACKING CROSS AT ANY POINT OF ANY ONE OF THE DISPLAYED X, OR Y, OR Z AXIS TO DEFINE THE LOCATION OF PLANE NUMBER 1".
- (3) The user has positioned the tracking cross at point "A" of the X axis as shown in Figure 3.7.
- (4) As soon as the first plane is displayed from point "A", another two menu light buttons will be displayed "POSITION" and "DONE" options. (Note that as the hit was on the X axis the working plane is parallel to the Y-Z plane).
- (5) To define nodes of the patch, the user hits "POSITION" option with the light pen, the program types the following message: "POSITION TRACKING CROSS IN PLANE NUMBER 1, TO DEFINE PATCH OR PATCHES DATA POINTS, TO DO SO:

TRACK THE TRACKING CROSS WITH THE LP AND POSITION IT AT ANY POINT ON THE PLANE. HIT "RETURN" KEY ON THE DECWRITER. REPEAT THE PREVIOUS PROCEDURE.

NOTE: TO DEFINE A PATCH YOU NEED TO POSITION TRACKING CROSS FOUR TIMES IN FOUR PLANES.

- (6) Figure 3.8 shows the four defined points (nodes) in plane number 1.
- (7) After defining the first four points of the patch the user can hit the "DONE" option. The program will erase "POSITION", "DONE" options and return to "DEFINE WORKING PLANE", "DONE" options. Again, the user could follow steps 2 to 7 to

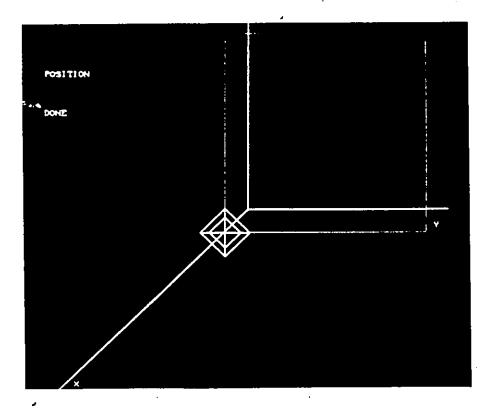


Figure (3.7) Tracking cross on the x-axis

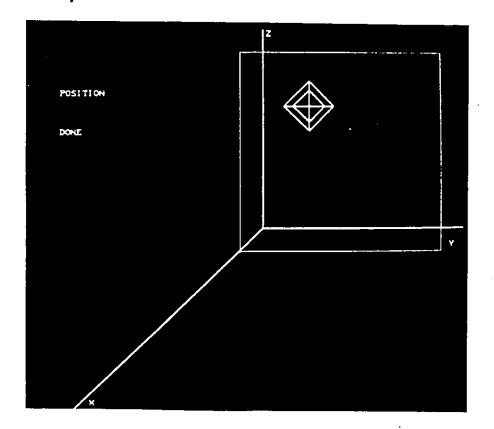


Figure (3.8) Four defined points in working plane number one

define the second four points of the patch in a second plane parallel to the plane through point A, and so on until the user is finished with entering the 16 points defining the patch.

Figure 3.9 shows the 16 points defining the patch (four points per plane). Note that the intensity level increases from the first to last so that the user can easily decide on the working plane. Note also that the working planes do not have to be parallel, it could also be orthogonal.

- (8) The user is ready now to hit the "DONE" option; the options "DEFINE", "DONE" will appear. Once more he could hit the "DONE" option, after which the patch surface will be dynamically drawn on the CRT. Figure 3.10 shows the typical patch surface.
- (9) A surface manipulation menu will then automatically appear on the CRT, e.g., "SMOOTH", "BLEND", "ROTATE", "ERASE"...etc.

3.3.3 Patch Modification

If the user wants to modify the shape of the patch, all he needs to do is to point at the "MOD" option (in the menu are shown in Figure 3.10) and then track the tracking cross with the light pen and position it at the new desired position (Figure 3.11); the program then asks the user to point with the light pen at the point to be repositioned (point B). The program then dynamically erases the original patch and draws the new modified patch, (Figure 3.12). The patch to be modified can be identified by only pointing with the light pen at any part of it, and also multiple design points may be relocated before reconstructing the patch surface.

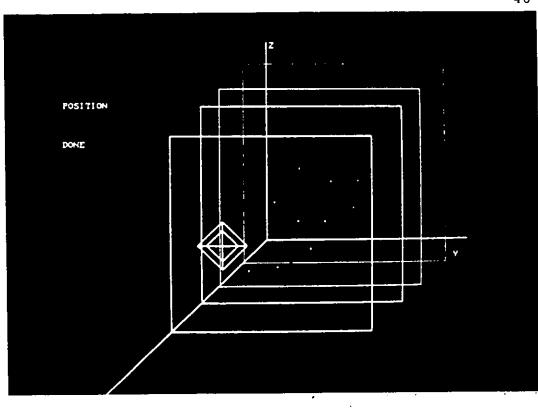


Figure (3.9) The 16 points defining the patch

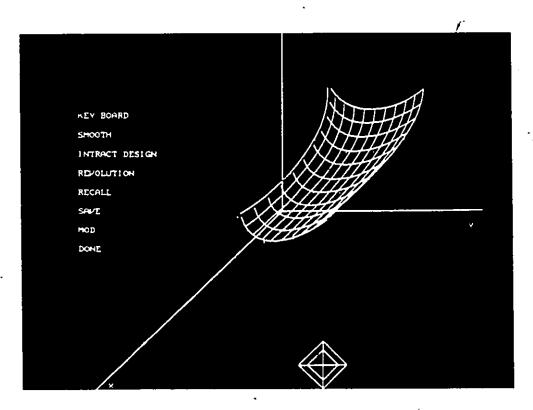


Figure (3.10) The generated patch

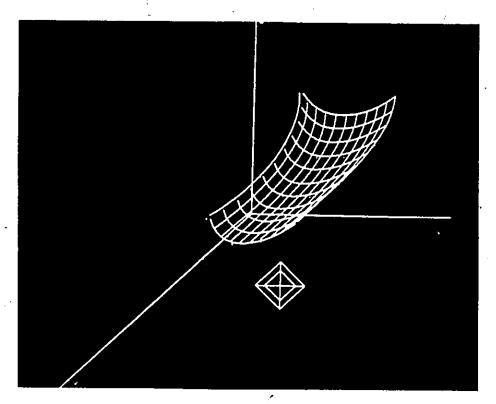


Figure (3.11) Tracking cross positioned at new designed location

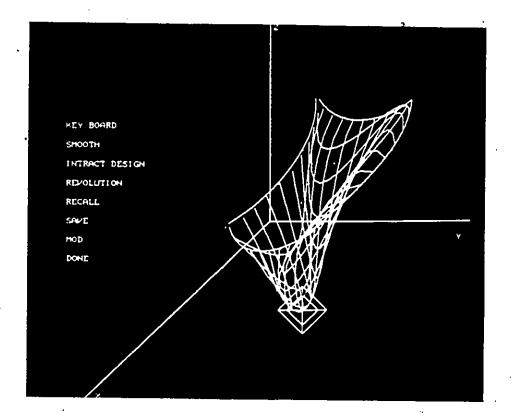


Figure (3.12) Patch after modification

3.3.4 Smoothing Two Patches at the Common Borders

Each patch is independently smoothed as shown in Figure 3.13. At the junction there can be an undesirable second order discontinuity. A facility is provided for designating such a border a "smooth junction" if this is desired, and any one of the two patches could be so modified to ensure "first and second order continuity" along the common borders. The user first hits the "IDENTIFY" option, then hits the first patch (to be fixed), then hits the second patch (to be smoothed with the first patch). Execution is initiated by hitting the SMOOTH" option, then the program smooths the two patches. Figure 3.14, shows the two patches after smoothing along the common borders.

3.3.5 Creating Transition Patches

The program can generate an inspetween transition patch which ensures first and second order continuity across the adjoining edges of the surrounding patches**. The designer defines his first and second non touching patches, hits the "BLEND" menu option, and the program generates the transition patch immediately. Figure 3.15 shows the two non touching patches before blending and Figure 3.16 shows the two patches after blending with a transition patch.

^{*} Appendix II ** Appendix III

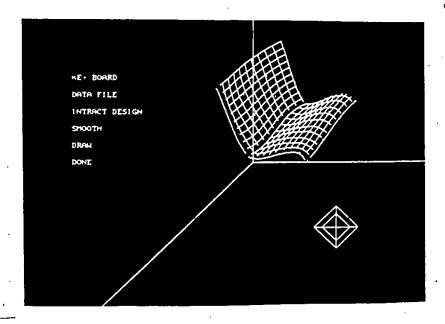


Figure (3.13) Two patches with discontinuity across the common border

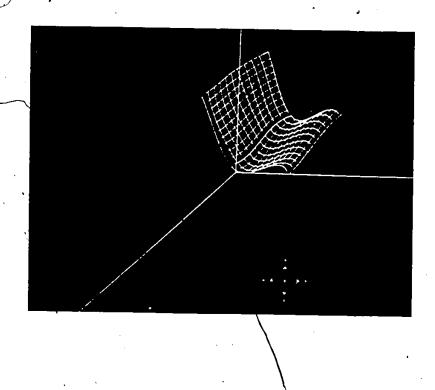


Figure (3.14) The two patches after "smoothing"

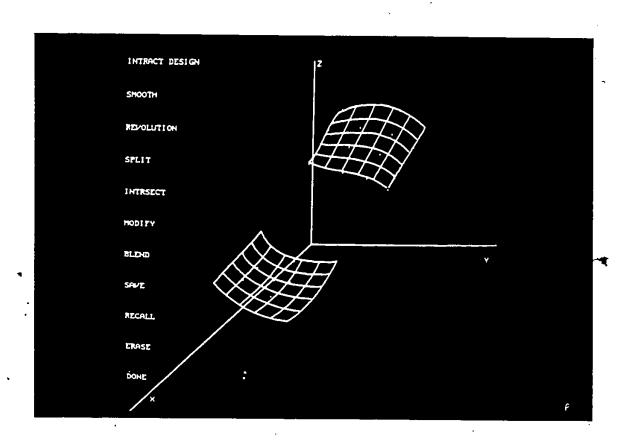


Figure (3.15) Two non-touching patches before "blending"

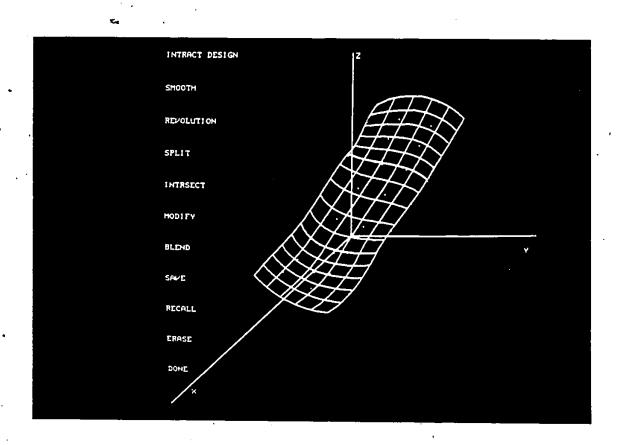


Figure (3.16) The two patches after blending

3.3.6 Creating Surfaces of Revolutions

Surfaces of revolutions can be created in a very simple way. As shown in Figure 3.17, the designer defines 6 points, the first two define the axis of revolution, the third, fourth, fifth and sixth define the surface as shown in Figure 3.18. The program uses points 3, 4, 5 and 6 to define the remaining 12 points defining the patch.

3.3.7 Creating Curves of Intersections with Planes

Curves of intersections between a patch and a plane can be obtained graphically. The designer, after defining his patch on CRT, hits the "INTRSC" menu option with the light pen. The program displays a representation of the pater on the U-V parametric plane (shown as a square on the upper left corner of the CRT in Figure 3.19, also the tracking cross will be displayed on the center of the CRT. The user is then asked to define the plane he wants to intersect with the patch; and, since we are working on the U-V parametric plane, the plane will be represented by a line. To define that line the user is asked to position the tracking object at two points on the borders of the U-V parametric plane, Figure 3.19. The program, then, dynamically draws the resulting continuous curve on the patch's surface, Figure 3.19. This way the user is only asked to define his plane graphically, rather than mathematically. Appendix IV illustrates the internal mathematics involved in generating these curves of intersection.

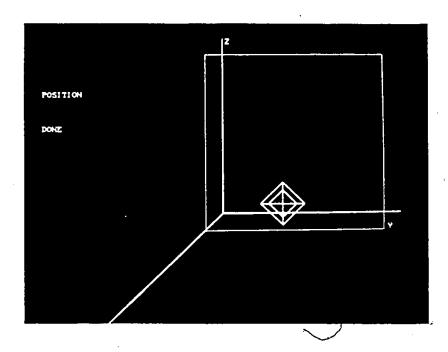


Figure (3.17) Six input points defining a surface of revolution

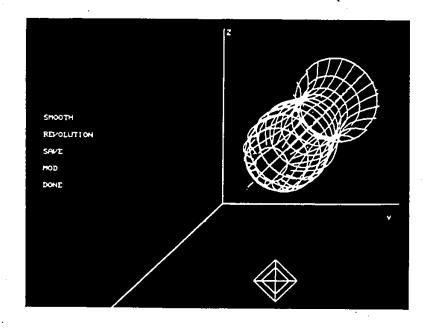


Figure (3.18) The generated surface of revolution

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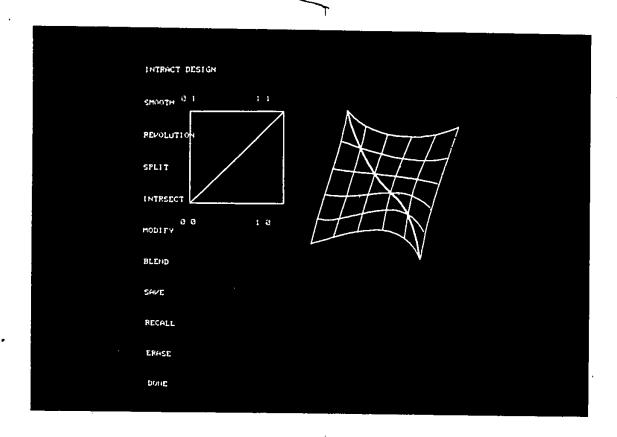


Figure (3.19) Intersection of a patch with a plane defined by a line in the U-W parametric plane

3.3.8 Patch Splitting

In engineering design of curved surfaces local refinements are necessary. A facility is provided to allow the designer to accomplish this task. The designer can split (subdivide) his patch to a smaller patch at the location he wants to do local refinements, then he can do whatever modifications he wants on the smaller patch. Using this technique the rest of the original patch surface would not be affected. Considering Figure 3.20, if the designer wants to create patch P2, by subdividing patch P1, all that he is asked to do is to define values for the u, parameters corresponding to the four corner points of patch P_2 , i.e., u_1 , v_1 , u_2 , v_2 . The user first hits the "SPLIT" option on the CRT menu area, then the program responds: DEFINE THE PATCH TO BE SPLIT; the user hits the patch to be split with the light pen; the program responds: "ENTER U1, U2, V1, V2 VALUES"; the user enters (e.g.) 0.1, 0.4, 0.1, 0.4; the program dynamically generates. the new patch (P2) which has geometric properties similar to the given patch. Figure 3.21 shows a typical splitting operation with .1, .5, .1, .5 for Ul, U2, V1, V2 respectively. The mathematics involved in creating the boundary (B) matrix of the patch generated by subdividing a given patch is illustrated in Appendix IV

3.3.9 Hidden Line Removal

The algorithms available for elimination of hidden lines in 3-D are among the most interesting in graphics work.

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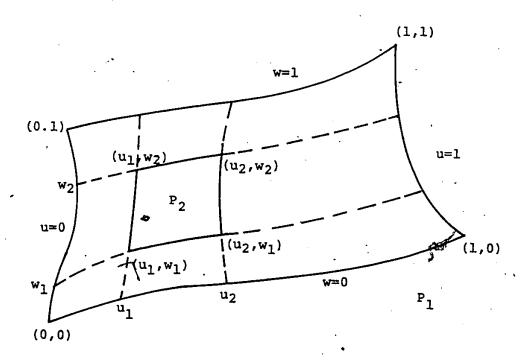
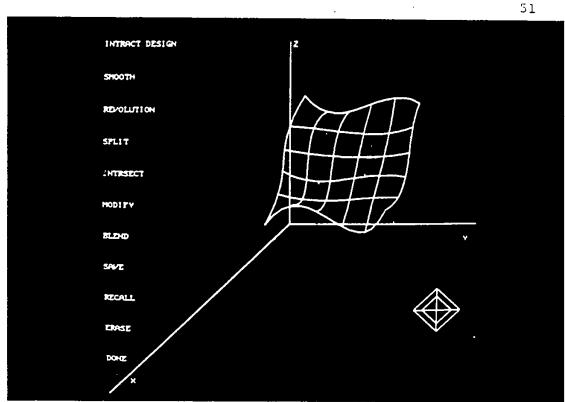


Figure 3.20 Geometry for Creating Patch P by Splitting Patch P $_1$



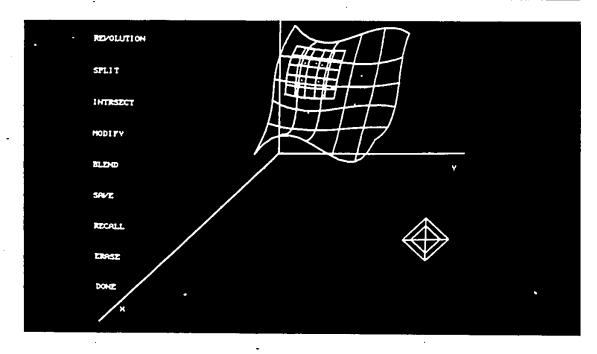
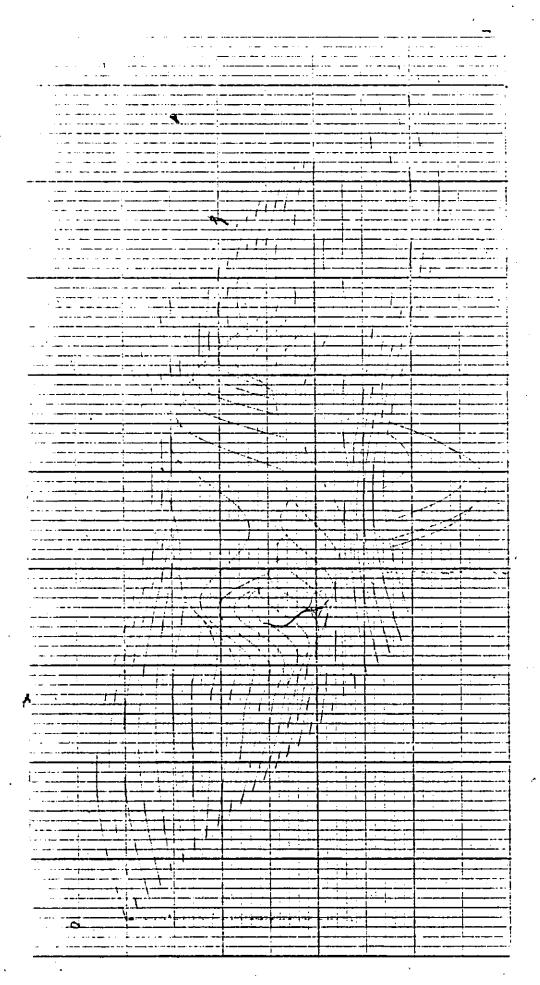


Figure (3.21) \hbar patch before and after "splitting"

The resulting displays are very attractive views of the scene which aid the designer visualizing his design on a CRT graphics terminal. The single disadvantage of hidden-line elimination is that considerable computation is required to decide which lines are hidden. A great deal of effort has been applied to solving the hidden line problem and the result is a sizeable collection of algorithms [44, 45, 46].

In this thesis no special attempt has been made to design a special hidden line removal program, instead a program called "HIDE" designed by H. Williamson and modified by M. Vannier and M. Oliff [47] has been adapted to our 3-D surface design program. "HIDE" is a FORTRAN-callable subroutine used to generate a 2-dimensional representation of a 3-D figure or surface. Subroutine HIDE is called once for each line to be plotted. The first line is plotted in its entirety. Only that portion of subsequent lines that is visually above (optionally below) all previous lines will be plotted. The result is an orthographic projection with hidden lines eliminated. A typical design for a designed surface after hidden line removal is shown in Figure 3.22.



gure (3.22) A designed surface with hidden
lines removed; (hard copy from
a CALCOMP drum plotter)

CHAPTER 4

SOFTWARE TECHNICAL INFORMATION

4.1 Introduction

Programs are designs, and software engineers began to realize that design techniques are very important in software design. Advanced programming techniques [75], have been used to develop the software of this thesis. Top-down and structured [76] programming techniques were used. The software system is composed of an executive or mainline program which acts as the control-reading in data, making major logical decisions, and calling subprograms which do various jobs in the free form surface design package. The software is also "modularized" to ensure the independency of modules. The structure of the program modules is described in hierarchical input-processing-output [75] charts, as shown in Appendix A. The function of each module is briefly described as well as the input and output of each.

Three main programs were designed to perform the design and manipulation of curved surfaces. Each program is capable of performing the design and manipulation of curved surfaces; the only distinction is the kind of input devices used. Program JOLIA was designed to accept input from the LP; program KEY to accept input from the keyboard; and program DATAFL to accept input from data points stored on data files.

A two-dimensional curve design program CURDES is also included in Appendix A. The program can be used for the interactive design of two-dimensional curves of any type.

4.2 Program Specifications

The specification is the document which fully defines the requirements of the design of the software. The program design was based on the following specifications:

- A CAD package is required for the interactive graphics design of a free form surface.
 - 2. Technical Level of Users Related to Programming and Modeling Skills

User should not require any familiarity with programming skills or mathematical surface modeling techniques.

3. Input and Output Software Configuration Needed for Use of the Package

The only input data permitted is x, y, z coordinates of spatial points lying on the surface. The output should include data files defining the designed surface.

4. Input and Output Hardware to be Used

A PDP-11/34 mini-computer, with a refresh type (GT46) graphics terminal CRT. Input media is floppy discs or, DEC packs (RK05).

Input - DEC WRITER II, Light pen.

Output - CRT

Core Memory Available for the Package 5: 32 K words.

5.1 Ship Hull Design

The preliminary design of a ship, is common with that of most other engineering objects, involves careful. compromise between a number of factors in order to produce the most economical result satisfying the functional requirements of the design. In the design of large commercial ships the basic functional requirements are usually stated as the ability to take a given load a certain distance. Thus, in the initial specification of the required ship, usually the parameters available to the ship builder are speed, length, beam, draught, deadweight, together with the class of ship, whether it be a tanker or cargo ship. From such sparse data, the hull surface definition has to be built up. The only geometrical information that is given for the surface specification is the length, beam and draught. Making use of the previous data, the traditional method of hull design was to define the form approximately by a series of points through which the surface would pass [48]. Initially these might define a set of smooth curves representing vertical plane sections across the hull. If these curves were intersected by horizontal plane sections one could plot sets of points representing the shape of the hull at various waterplanes.

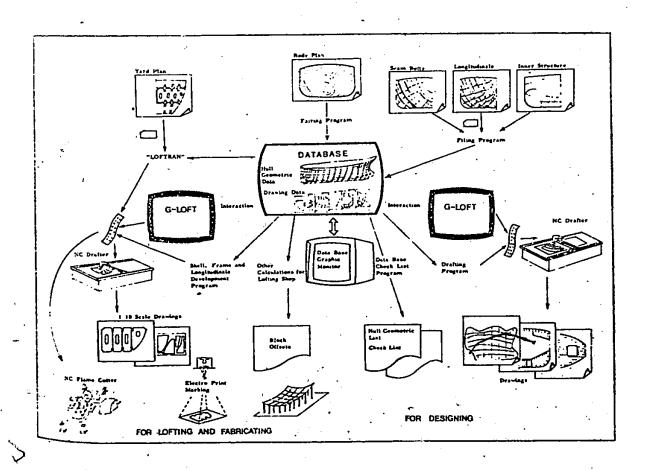


Figure (5.1) The NK-ASD CAD system for overall shipbuilding operations, from design to building

In general, curves through these points would not be smooth. The process of fairing then begins and consists of reconciling the two sets of curves until both are smooth and both represent intersections with the same surface. Usually this was a tedious and time consuming task. Most of this recent effort has been devoted to the 'hull fairing' problem in which a given lines plan is to be approximated mathematically, often being improved in fairness at the same time. D. Taylor [49] mathematically generated the hull forms in his standard series, defining sectional area curve and design waterline by fifth order polynomials in accordance with form parameters* he prescribed. Weinblum [50, 51, 52] extended the principles of parametric lines creation in connection with the systematic variation and hydrodynamic optimization of hull forms.

More recently two distinct goals have been pursued by numerous investigators: Lines creation by 'distortion of a parent hull form' versus lines creation from 'given hull form parameters'. In the former category significant contributions were made by Lackenby [53], Schneekluth [54], and Puchstein [55]. With the latter problem, good progress was also made over the last two decades owing to research efforts by Thieme [56], Miller and Kuo [57], Williams [58], Kwik [59, 60], Kuiper [61], and Reed and Nowacki [62] and others.

Following these traditional efforts for ship hull

^{*} See Appendix B

design, MacCallum [63, 18] developed an excellent system for the preliminary design of ship hulls using interactive graphics. However, MacCallum's system had some disadvantages

- 1. The design consisted of an arrangement of surface patches joined along common boundaries and made continuous only to the first degree. In some cases a second degree: continuity is necessary.
- 2. Curves of intersections (body plan) could not be obtained dynamically.
- 3. Design could not start from scratch; however it could be started using a library of topologies, i.e., standard types of bow forms, types of sterns, etc. The program can be left to construct an initial form by piecing together items from its library of standard forms. If the need arises to define a new topology the program is handicaped.
- 4. Shape modification was achieved through 'dragging' one of the corner points of a patch using a light pen, but any other control point on the patch's surface could not be so dragged. If this was required, the original patch was split at the required control point and then the regenerated patch was changed.

Another interesting system was designed by Yuille [64]. The user has to deal with parametric derivatives and cross derivatives, which makes the system difficult to use. The program is not suitable for making large changes to the shape of a patch in one step because it is not practical to

move points on a patch very far in a given coordinate direction by altering values of the cross-derivatives [10]. One can, however, obtain good results by making a series of small changes in different directions.

Walker [65] argued that the best way to define a ship's hull is through the use of a close mesh of points rather than analytic surfaces.

Groot [66] presented a design method together with practical results of designed hull surfaces composed of simple analytical functions. Still; the method is not capable of designing any desired curved surface.

It is worthwhile here to discuss the CAD system for shipbuilding at the Japanese 'Chita' shippard, the most modern shippard in the world [68]. In 1978, the shipbuilding division of Nippon Kokan KK, one of the major shipbuilders in Japan, completed a comprehensive CAD system for shipbuilding operations from designing to building [67]. The NK-ASD system features are a database system, improved usage graphic display equipment and direct control of NC equipment. In the NK-ASD system the common data, such as hull configuration and structure, are filed in a database and can be extracted at will. Figure 5.1 shows a functional outline of the system. The major programs are as follows:

- 1. Fairing program for calculating the outer hull configuration.
- 2. A filing program that files the detailed structure of the vessel into the database.

- 3. LOFTRAN for producing the control tape for NC parts generation.
 - 4. G-LOFT for parts generation by interactive graphics.
 - 5. The database graphic monitor program.
 - 6. The shell development program.
- 7. A program providing a checklist. The graphic display unit allows designers to extract information from the files. The G-LOFT program helps preparing drawings of design and parts generation. The graphic monitor program helps the user to study the contents of the database in numerical or graphic form. One of the great strengths of the NK-ASD system is that the information concerning the design is directly connected to the manufacturing processes.

A new method is introduced in this thesis for ship hull design. The method makes use of the free form surface design program explained in Chapter 3. When using the program one is concerned with direct manipulation of a surface from the start; and not with curves. The naval architect interacts with the computer, which stores the information and then calculates and draws curves in the current designed surface. The procedure, when using this program to design a new hull, is first to sketch the vessel and the outlines of the patches that will form a preliminary representation of its hull. A new surface is computed and stored with its patch corners at positions corresponding to those in the sketch. Actually, the designer can start designing his hull from scratch or by modifying a previously stored surface. The designer can

easily obtain true views of plane cross sections across and along the new surface representing the hull. He can also commence an iterative procedure during which he repeatedly modifies the surface by dragging control points; and he then examines the shape of the plane cross sections; repeating this until he obtains the shape he requires.

In addition to the fairness of the hull form the program could be very easily extended to accommodate some analysis routines which will aid the naval architect in the design process, e.g.,

- 1. Subroutines to calculate the resistance of the underwater body to motion through the water.
- 2. Subroutines to calculate the flow (pattern) of water around the vessel and into the propellers.
- 3. Subroutines to calculate the displacement and center of buoyancy of the underwater form.
- 4. Subroutines to measure the static and dynamic stability of the ship.
- In the next section we will describe the general algorithm of the ship hull design program from a user's point of view.

.1.1 Ship Hull Design Algorithm

. The general approach of the program will be as

follows:

. User inputs data from which major dimensions could be

calculated.

Service Speed (Knots)

Deadweight (tons)

Deadweight coefficient (deadweight/light weight

- + deadweight)
- 2. Program Calculates L, B, H, D as shown in Figure 5.2.

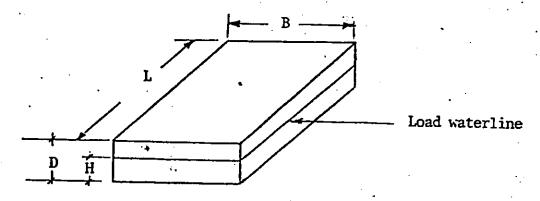


Figure 5.2 Box shaped vessel

(Note: User would be able to change any of these parameters).

- 3. Program displays the box shaped vessel showing the load waterline.
- 4. User is asked to divide the box into four major divisions, using the light pen and tracking object, as shown in Figure 5.3.

F - Forward

M - Middle

A - Aft

S - Stern

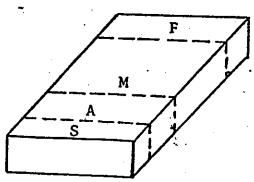


Figure 5.3 Divided vessel

The program returns the length of each division. If the user wants to change any length he could go back and repeat the previous step (Figure 5.3).

including the midship section curve) resulting from 'slicing' the ship with vertical planes through the two edges and the dotted lines of Figure 5.3.

5.1.2 Designing the Middle Body

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1. As the shape of the midsection is characteristic of size and form as well as function, the design process should begin by designing the middle body (M).

The user is asked to define the shape of the midship section (using the light pen and tracking object the designer enters points defining the midship section curve, using the same algorithm for the free form surface design program described in Chpater 3, section 3.3.2). The boundary of the curve is the rectangle shown in Figure 5.4. Due to the symmetry of the ship along the centerline, only half of the curve need be defined, see Figure 5.5.

2. After the user defines the shape of the midship section the program returns the value of the "midship section coefficient" ($C_{\rm m}$)

$$\cdot C_{m} = \frac{A_{m}}{A}$$

If the value of $C_{\overline{m}}$ is different than what the user



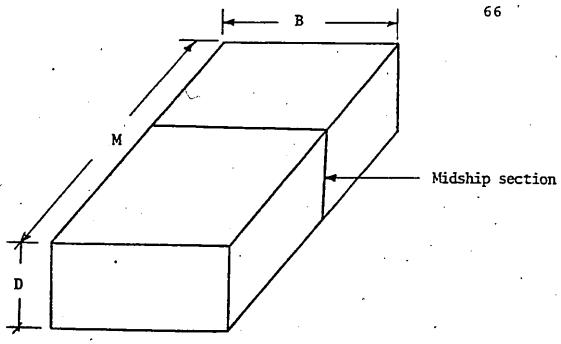


Figure 5.4 Middle body of a ship

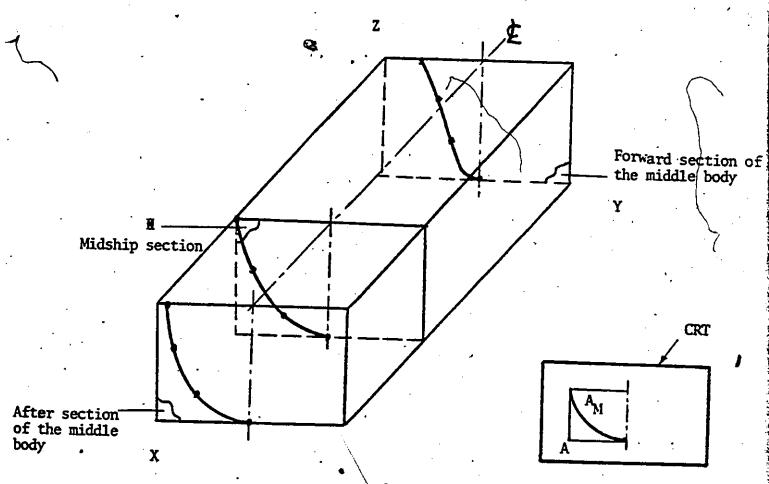


Figure 5.5 Middle body boundary curves

has in mind, he could go back and redesign the curve.

3. The user is then asked to define the curve of the forward section of the middle body using the same procedure as the previous step, see Figure 5.6.

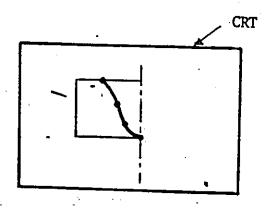


Figure 5.6 Curve of the forward section of the middle body

4. The two previously created curves are then plotted on the 3-D space on the Y-Z planes (Figure 5.7).

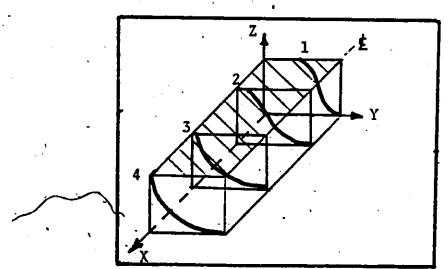


Figure 5.7 Forward part of middle body sectional curves

- 5. The user is asked to define (using the LP and tracking object) two additional curves in two parallel planes so that curves 1, 2, 3 and 4 (four defined points/plane) will now define the patch.
- 6. The resulting patch is then dynamically plotted on the CRT (see Figure 5.8).

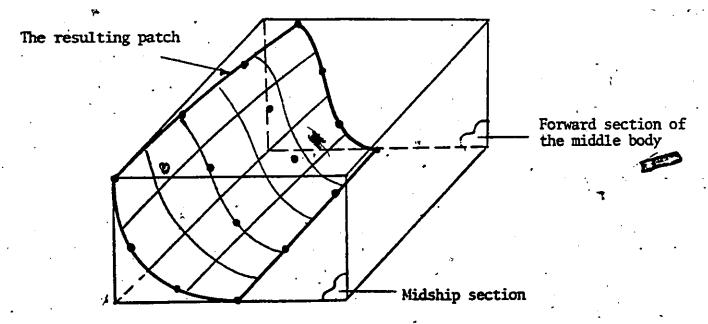


Figure 5.8 Resulting patch of the forward part of middle body

- 7. If the user is not satisfied with the shape of the patch, he could go back and change the position of any of the points defining the patch until he is convinced with the shape of the patch.
- 8. The same algorithm could be used to develop the surface of the afterpart of the middle body, making sure that continuity exists along the common border between the forward and afterparts (patches) of the middle body. The

resulting middle body will look like Figure 5.9.

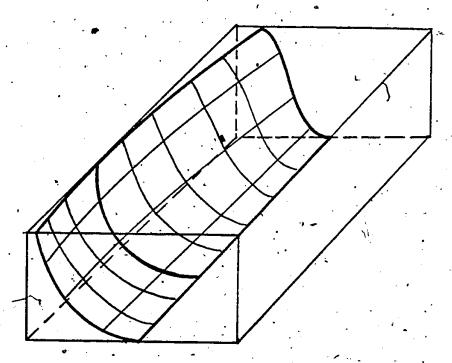


Figure 5.9 Middle body surface

5.1.3 Designing the Forward Part [F]

The user is asked now to define only three sections of the forward part (the after section of the forward body has been defined in designing the middle body), see Figure 5.9.

Notcie that the forward section of the forward body has a bulbous bow. Actually the user would be able to define any shape for the forward part.

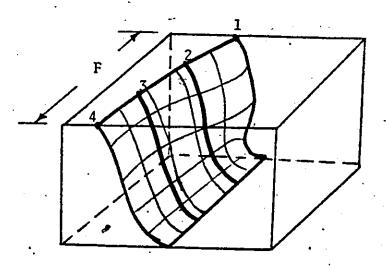
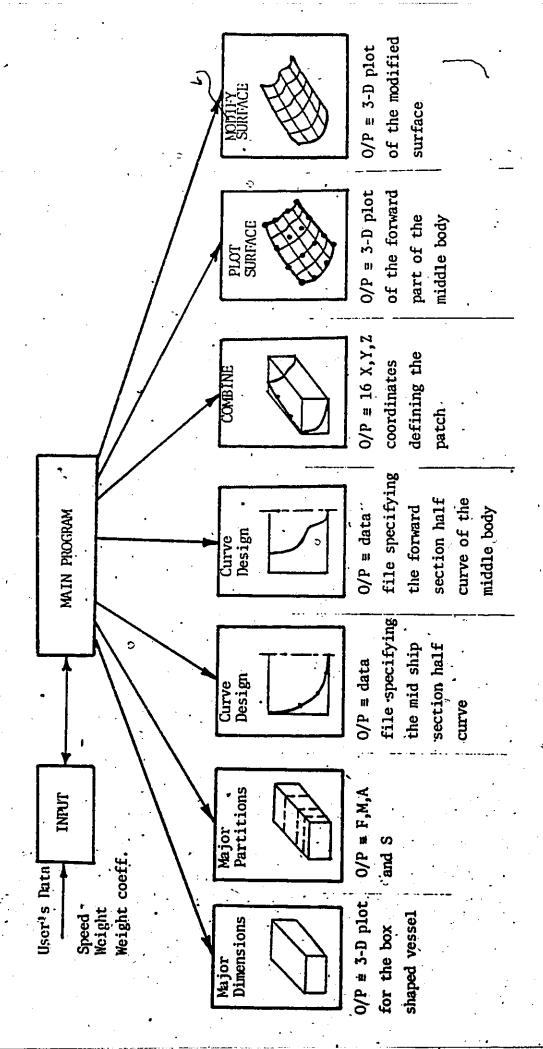


Figure 5.10 The Forward surface

5.1.4 Designing the Aft Part and Stern [A], [S]

The same procedure developed in the previous sections could be used to define the 'Aft', and 'Stern' parts of a ship.

Figure 5.11 shows a complete 'top-down' program design for ship hull surface design. A complete ship hull design using the prescribed algorithm is shown in Figure 5.29. True views of the curves of intersections, i.e., body plans are also shown in Figures 5.23 to 5.27.



HIPO chart for the hull design program Figure (5.11)

5.2 <u>Chair Design</u>

The main steps taken to design a chair are as follows.

- 1. The industrial designer has some new idea for designing a chair. He sketches the new shape on a piece of paper as shown in Figure 5.12.
- 2. The designer begins by subdividing his surface into patches whenever he feels the need to define a certain part of the surface by a patch.
- 3. The divided chair would look like Figure 5.13, (note that the designer is still working on a piece of paper).
- 4. The designer is now ready to define to the computer each patch using the light pen input command.
- 5. To define patch number one, the user would define
 16 points (four per working plane) as shown in Figure 5.14.
 The program will then display the surface fitting these points.
- 6. Using the same four working planes (parallel to X-Z plane), the user would continue defining patch number two.

 Note that point number one of patch 2 should be at the same position as point 4 of patch number 1, and similarly for points 5 and 8, 9 and 12, and 13 and 16 respectively.
- 7. After the user is finished with defining his five patches; using the prescribed four working planes; he can now alter the position of any point (node) and see the effect on the chair's shape until he is satisfied. He can then 'SAVE' the display and create a data file containing the X, Y, Z

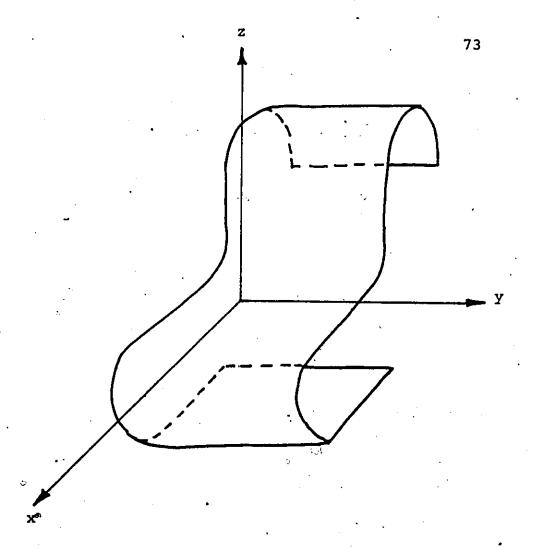


Figure (5.12) 3-D view of a chair as sketched by a designer

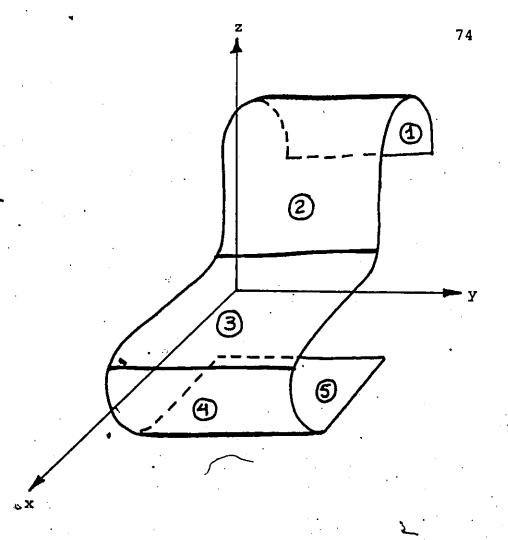
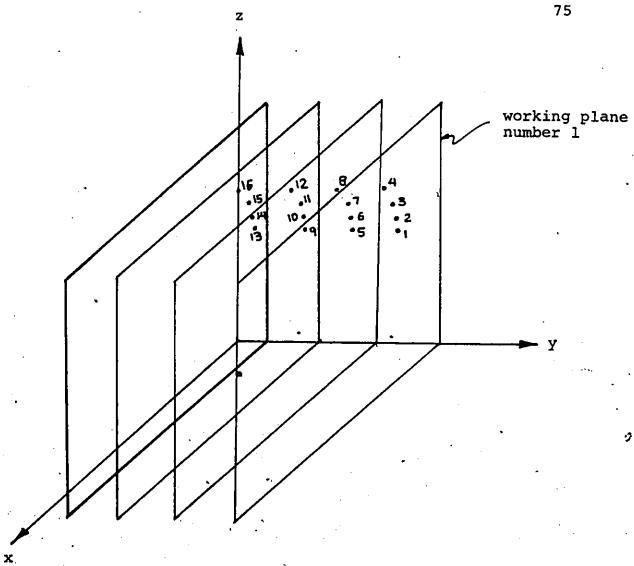


Figure (5.13) Chair divided into 5 patches, as sketched by the designer



16 points defining patch number one, 4 points per working plane Figure (5.14)

coordinates of the points defining the chair's surface.

A typical chair design as designed on the CRT graphics terminal using the prescribed algorithm is shown in Figure 5.15.

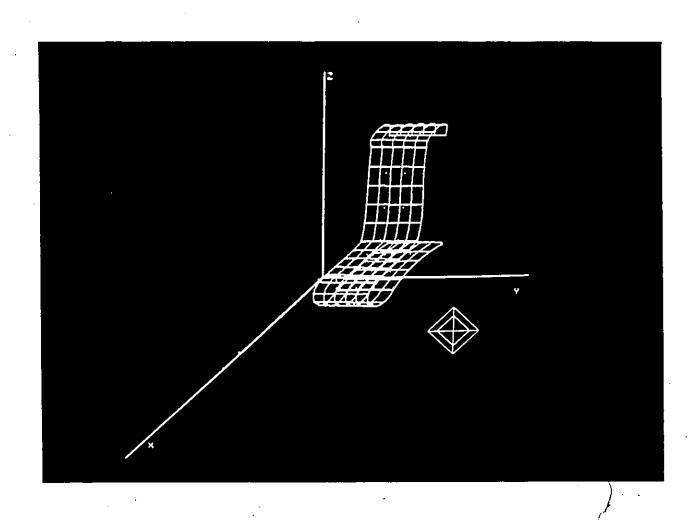


Figure 5.15 Three-dimensional view of chair design

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5.3 <u>Duct Design</u>

A typical procedure for a duct design using digitized X, Y, Z coordinates as input data is explained in detail in the following section:

- \odot 1. The designer draws, elevation and plan for the duct as shown in Figures 5.16 and 5.17.
- 2. From both figures the designer can easily obtain X, Y, Z coordinates of 16 control points of each patch as shown in Figure 5.16.
- 3. Using input via keyboard command, the user can now enter the 16 X, Y, Z coordinates of each of the four patches.
- 4. The program will dynamically generate the surface of the duct as shown in Figure 5.18.
- 5. The user can 'SAVE' the display and create a corresponding data file. The user can also obtain curves of intersections and get the X, Y, Z coordinates of the points defining the intersection curves.

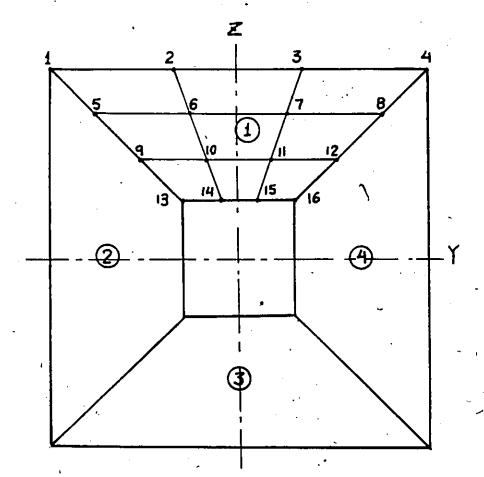


Figure 5.16 Elevation of the duct

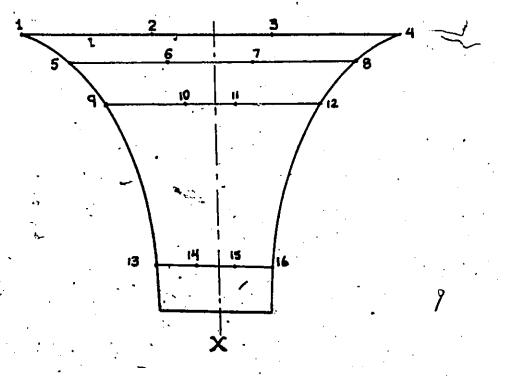


Figure 5.17 · Plan of the duct

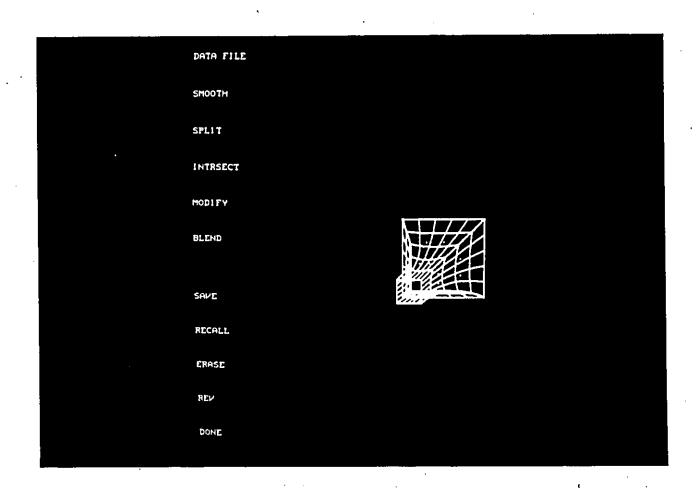


Figure (518) Duct surface as shown on CRT

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5.4 Glass Container Designs

Simple round containers, such as bottles and wine glasses, can be completely specified with a single section line defining the profile of the container. The surface design program enables the designer to 'draw' the profile with a light pen (in reality, he is defining 4 points which construct a parametric cubic curve). This profile can be modified until the designer is satisfied and he can observe a pictorial view of the complete container, whilst he is working. The real power of this developed technique is in designing non-round containers from a predesigned round container (using control points dragging).

Designing round containers is very easy using the developed free form surface design program as will be proved in the following practical desing of a bottle:

- 1. The designer hits the 'REVOLUTION' option on the meanu area. The program responds by typing the following message: 'PLEASE ENTER NUMBER OF PATCHES'. Let us say that the user decided to use two patches to define his surface (neck and main body of the bottle). The user enters: 2.
- 2. The program responds: 'DEFINE WORKING PLANE, YOU SHOULD DEFINE 10 POINTS ON THAT PLANE. THE FIRST TWO DEFINE AXIS OF REVOLULTION, THE REST ARE USED TO DEFINE SURFACE'.
- on that plane (as described in detail in Chapter 3, section 3.3.6).

4. The program dynamically generates the two patches defining the bottle's surface as shown in Figure 5.19.

Figure 5.20 shows a design for a non-round container, designed from a predesigned round container by dragging some of the control points defining the original surface.

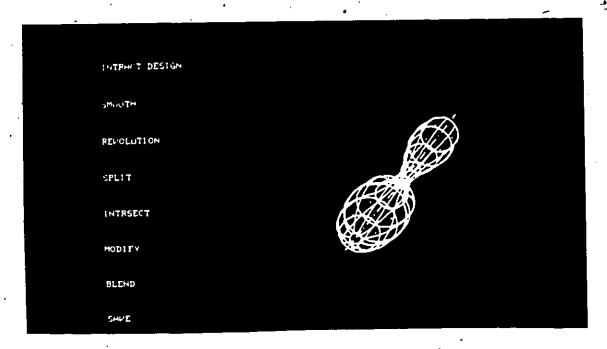


Figure (5.19) Design of a round container

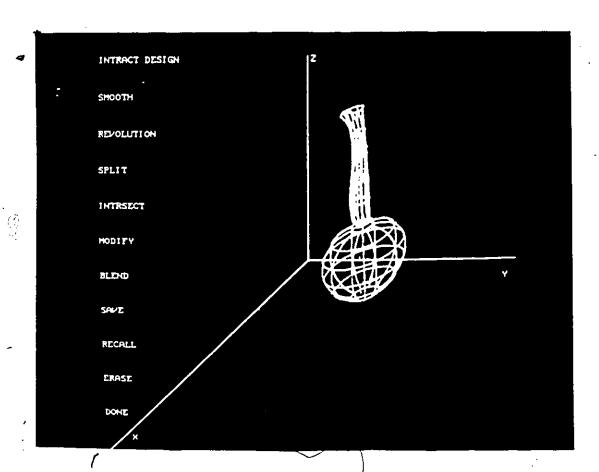


Figure (5.20) Design of a non-round container

5.5 Other Designs

Figure 5.21 shows a photo taken of the CRT graphics terminal, for a "nozzel" designed using only one patch.

Figure 5.22 shows a "smoking pipe", that was de-

Figures 5.23-5.29 show the different stages of a ship hull design.

The most interesting feature of the prescribed software free form surface design system is the short elapsed design time. The nozzel shown in Figure 5.21 was designed in less than one minute. The smoking pipe of Figure 5.22 was designed in less than 4 minutes. The ship hull of Figure 5.29 was designed in less than 10 minutes.

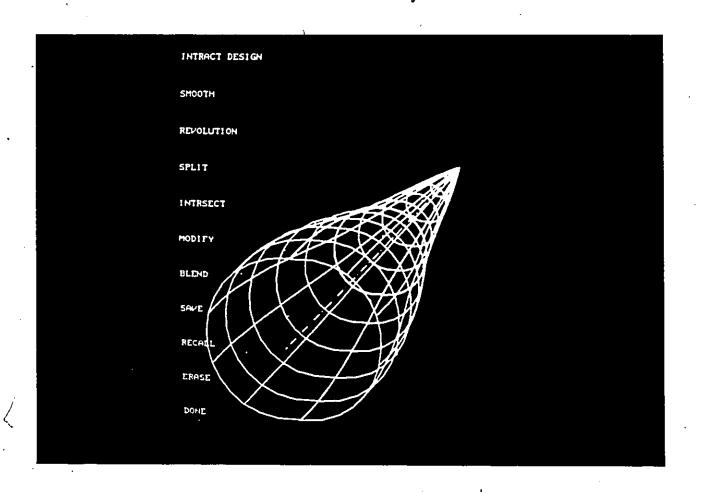


Figure (5.21) A nozzel design using only one patch

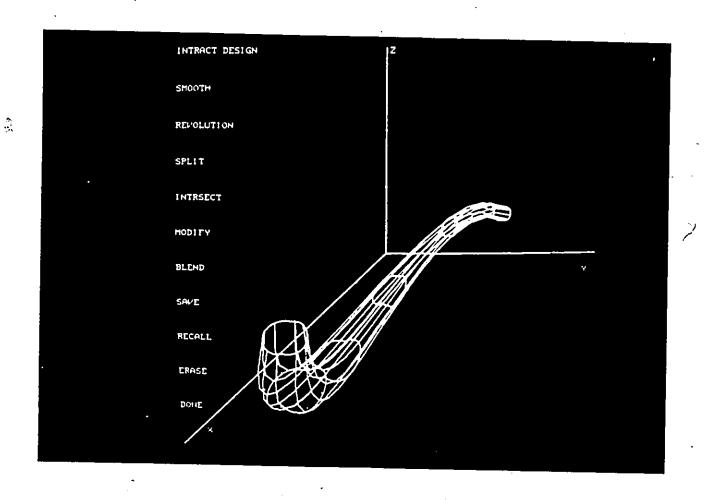


Figure (5.22) A smoking pipe design using only two patches

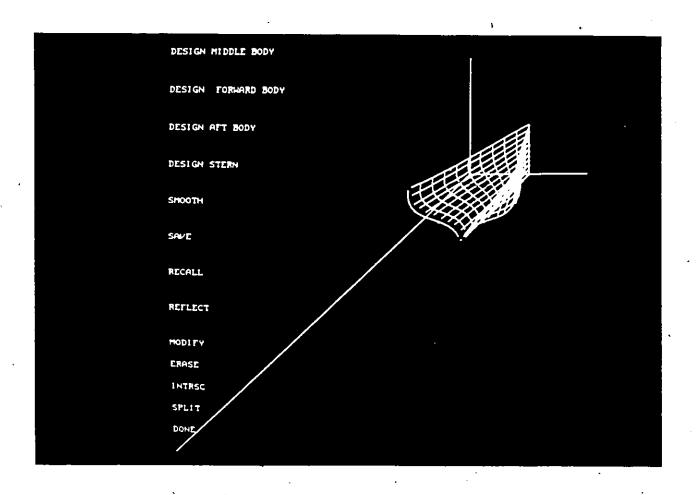


Figure (5.23) Design of the forward body of a ship

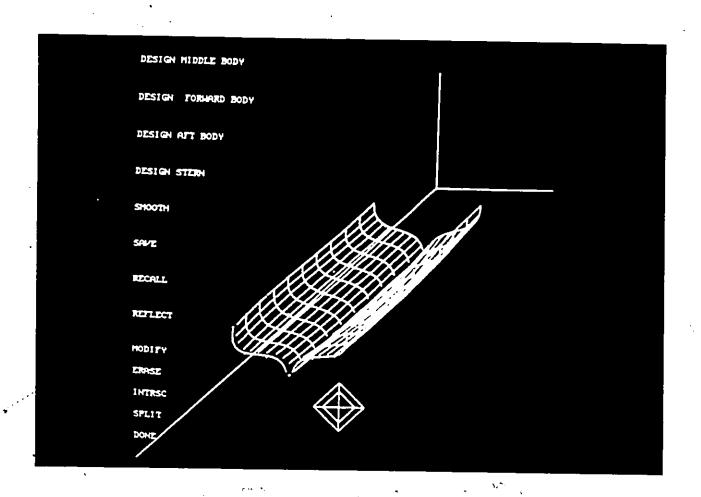


Figure (5.24) Design of the middle body of a ship

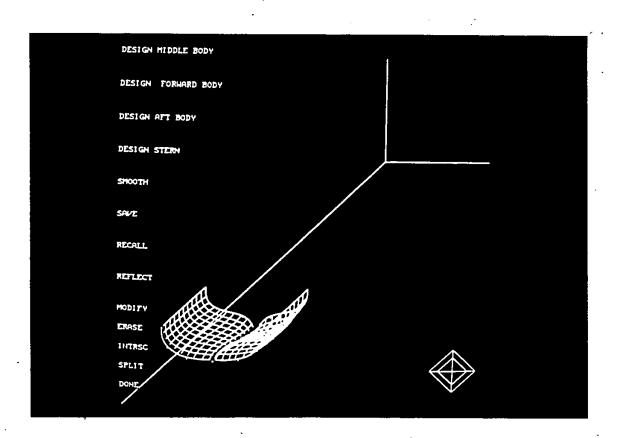
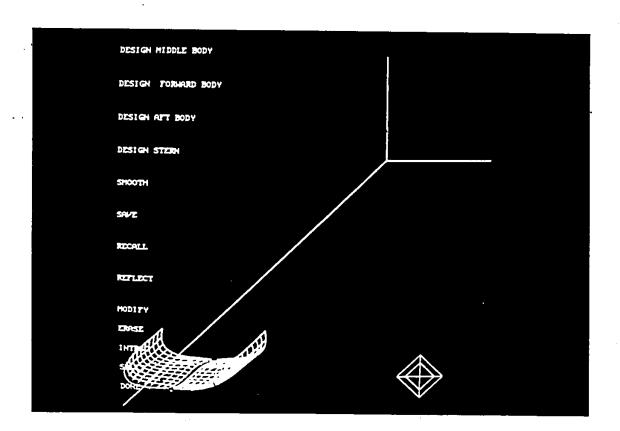


Figure (5.25) Design of the aft body of a ship



A

Figure (5.26) Design of the stern of a ship

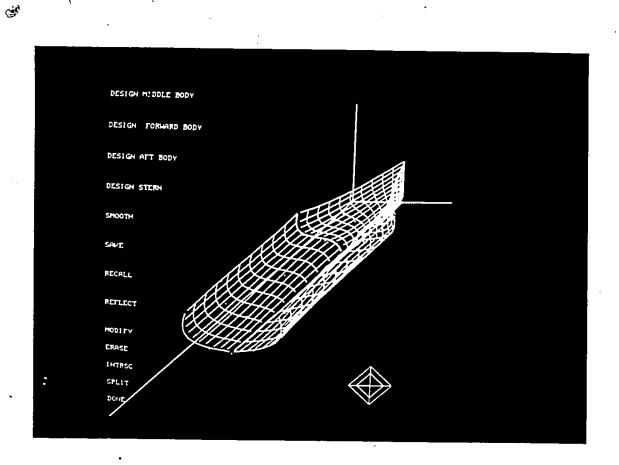
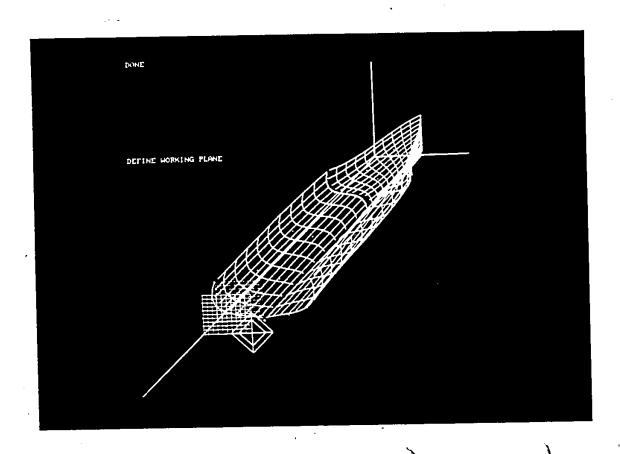


Figure (5.27) The forward and middle bodies combined together



Designing the aft body as combined to middle and forward bodies of the ship Figure (5.28)

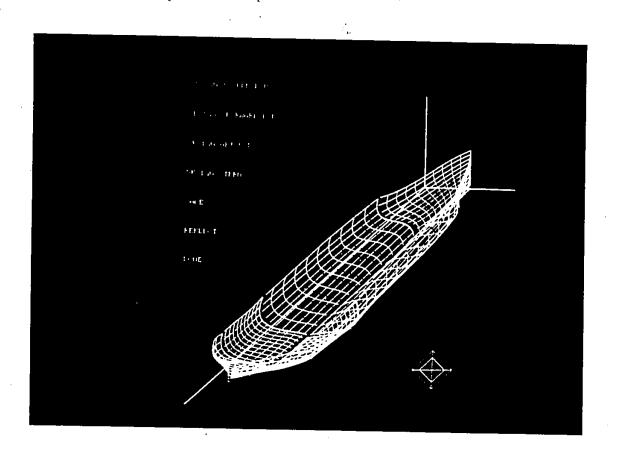


Figure (5.29) A complete ship hull design

CHAPTER 6

MANUFACTURING SCULPTURED SURFACES USING NC SYSTEMS

6.1 What is NC?

Before we discuss the possible ways of manufacturing sculptured surfaces designed using the CAD graphics system described in the previous chapters, we will have a quick look at NC [42]. Numerical control is not a wind of machine tool but a technique for controlling a wide variety of machines. It is a system that can interpret a set of prerecorded instructions in some symbolic format; it can cause the controlled machine to execute the instructions, and then can monitor the results so that the required precision and function are maintained The numerical control , system forms a communication link as shown in Figure (6-1). Symbolic instructions are input to an electronic control unit which decodes them, performs any logical operations required, and outputs precise instructions that control the operation of the machine. The feedback enables the controller to verify that the machine operation conforms to the symbolic input instructions.

In 1957, the first successful NC installations were being used in production; however, many users were experiencing difficulty in generating part programs for input to the

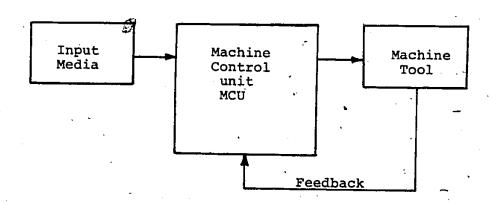


Figure (6-1) A simplified schematic of an NC system

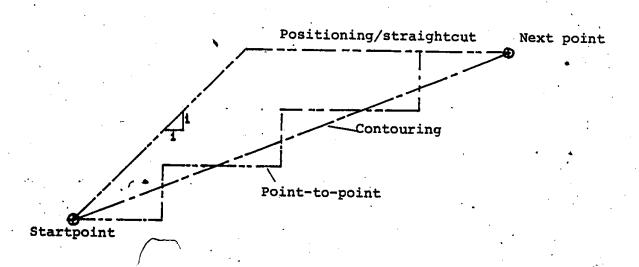


Figure (6-2) Comparison of control system paths

machine controller. To remedy this situation, M.I.T. began the development of a computer based part programming language called APT - automatically programmed tools. The objective was to devise a symbolic language which would enable the part programmer to specify mathematical relationships in a straightforward manner. APT provides the programmer with three tools.

- A geometry description capability that enables him to describe necessary calculations without having to execute them.
- 2. A method of describing tool motion.
- 3. A means for specifying inactive tool information such as feeds, speeds, and miscellaneous functions.

Although modern NC systems perform many functions, the most important controlled operation is dynamic positioning of the cutting tool with the use of system coordinates that are general enough to define any geometric motion. Points along the part profile* are defined by x, y, z coordinates and fed in sequence to the NC controller which generates the appropriate positioning commands. Positioning can be accomplished using two distinct methods, absolute positioning or incremental movement. The incremental system uses the change in x, y, z dimensions to specify position, whereas the absolute system uses coordinate values.

^{*} In many cases offset points (cutter path) are required.

The path which the cutting tool follows as it traverses from point to point depends on the type of control system used. Figure (6-2) describes the different control system paths. The contouring controller, the one that we are interested in, generates a path between points by interpolating intermediate coordinates. All contouring systems have a linear interpolation capability (i.e., the ability to generate a straight line between two points).

A numerically controlled machine vi/11 function only if the proper instructions are developed and passed on the machine control. The process by which the symbolic NC instructions are transferred to the control unit is termed the "communication cycle". The cycle begins with the development of a set of NC instructions, called a "part program", that specifies positioning data and related machining functions in a machine readable format. The next step in the communication cycle(is the physical transfer of the part program to the machine controller. The "communication medium" (usually a tape) transports a symbolically coded part program to the control upit. Even a relatively short set of NC instructions may contain hundreds, and possibly thousands, of alphanumeric characters and special symbols. For this reason a communication medium must represent a symbolic code in a compact form which can be easily deciphered by the machine control. Usually this communication medium (NC tape) is created using a special computer program [69] (post processor) called by the processor and used to convert

cutter location (CL) into that medium that is understandable by an MCU (machine control unit). A typical post processor contains five elements: input, motion analysis, auxiliary functions, output and control and diagnostics.

The "input" element reads the cutter location data and miscellaneous information that is output from the processor. It verifies the format of the data and transfers appropriate values to other elements of the post processor.

The "motion analysis" element contains the dynamics and geometry sections. The geometry section performs co-ordinate transformations to convert the general CL data into specific machine tool coordinates. The geometry section insures that the machine's physical limits are not exceeded and that the tool does not cut into any part of the machine. Finally, it is the job of the geometry section to select proper linear and rotary motions and to insure that the resultant path is within tolerance.

The "dynamics" section of the motion analysis element calculates the appropriate tool velocity based on servo type, and the acceleration/deceleration characteristics of the machine tool.

The "auxiliary function" element provides for the output of miscellaneous and preparatory command codes, i.e., translates the machine control commands, e.g., COOLANT/ON, SPINDLE/ON, COOLANT/OFF, etc.

The "output" element of the post processor generates two types of output: (1) the actual numerical control blocks

in a media form that can be either directly input to the MCU or easily converted into a form for direct input to the MCU and (2) computer printout of each NC block in a readable format.

The control and diagnostic element of the post processor is necessary to insure that a proper flow of information occurs in the program and that analysis errors are diagnosed and brought to the attention of the NC programmer.

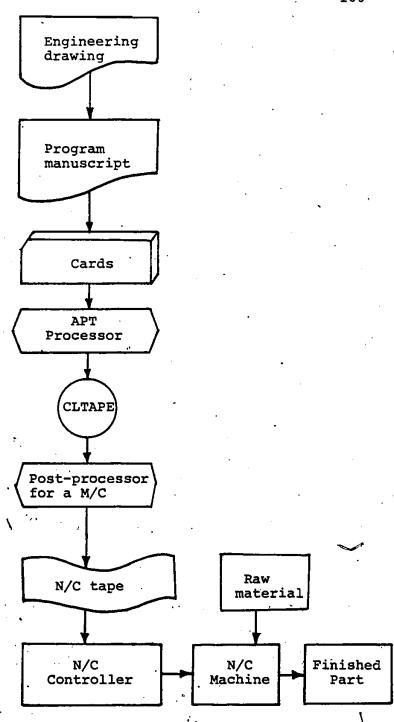
The total picture from engineering drawing to finished product is shown in Figure (6-3) using an APT processor.

6.2 A Proposal for an Integrated CAD/CAM System

Our aim is to build a semiautomatic programming system that can handle both the design and manufacturing of sculptured surfaces.

In Section 6.3 of this thesis, we introduced a new technique for determining the CL. This method could be very easily incorporated in our interactive free form surface design program in the following fashion:

- 1. After the designer has designed his surface and he is satisfied with his design he can interactively create curves of intersections with the surface using the 'INTERSECT' light button command.
 - Depending upon the complexity of the designed surface,



The total picture from engineering drawing to finished part Figure (6.3)

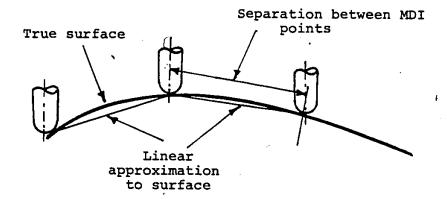
4

the desired accuracy and the designer's experience, the designer can decide upon the number of points to be interpolated on a curve of intersection (one complete cutter pass or what is referred to as master dimension information MDI) and the number of these MDI's (Figure 6-4). Automatic selection of the number of points on an MDI and separation between successive MDI's requires a knowledge of the radius of curvature at various positions in the surface. A method for the evaluation of minimum curvature for a parametric cruve has been developed by Helpert [70] and involves an optimum search technique for determining the minimum value of a function. The consequent knowledge of minimum radius of curvature enables the maximum cutter size (diameter) to be used [71].

3. Using these data points, the CL's could be calculated as described in section 6.3 using an NC processor, e.g., NELAPT, APT [72] (an NC processor developed by the National Engineering Laboratory (NEL)).

As the engineering drawing is the major interface between design and manufacture it was logical to consider curves of intersections (MDI's or contours) as a possible digital interface between CAD and CAM [73]. The whole idea of using MDI's as an interface between CAD and CAM is shown in Figure (6-5).

Calculation of cutting conditions and determining technological data (optimized machining sequence, tool radius, feed, rotational spindle speeds, etc.) as a part of an



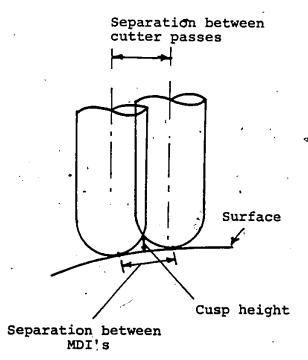


Figure (6.4) Separation between points and between cutter passes

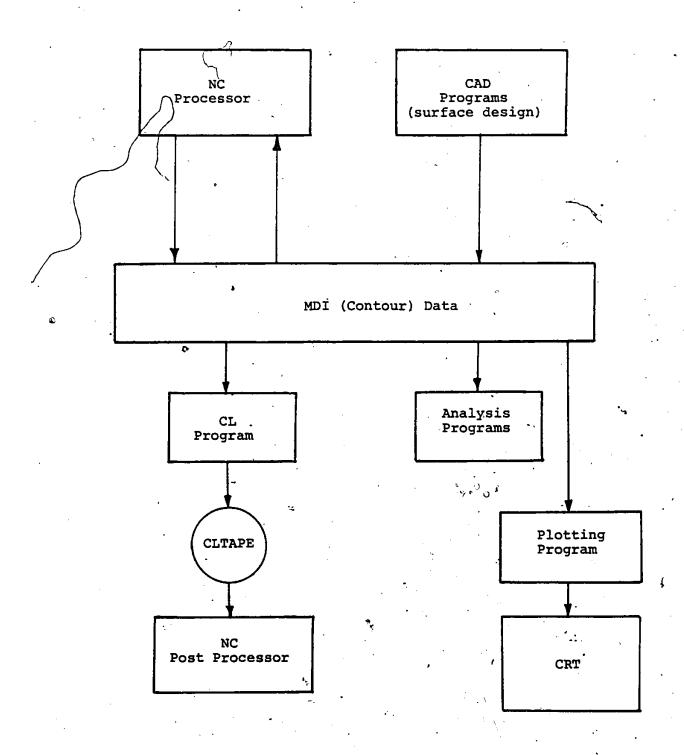


Figure (6.5) The MDI as an interface between CAD and CAM

integrated information processing system could be achieved $^{\prime}$ using an especially designed preprocessor [74]. The practical application of this technique using a minicomputer could be performed as follows. The preprocessor control program calls a predesigned milling optimization program which in turn will scan the geometrical data files. During the execution of the optimization program, the cutting conditions are computed via interaction with the machinability data, operation requirements, tooling available, etc., stored on the technological data files. The optimized cutting parameters should not be imposed on the user. The part programmer should make use of his experience and commonsense to select the cutting parameters and tool specifications from a displayed table containing the optimum values. A very good example of the preprocessor technique is given in Reference [74] in Chapter 4.

6.2.1 Advantages of the Proposed System

In contrast with processing in a batch environment (Figure 6-3), the proposed system will provide the user with the capability of generating the control tape interactively. In an interactive graphic environment, the on-line computer graphics will facilitate the part programmer's job and reduce the number of trial runs before producing a good part. Part-geometry input verification, via an interactive graphics terminal, can shorten and refine the procedure

used to produce a part. Errors are detected before the actual processing begins. In the event of an error prediction the user can easily and dynamically modify and correct the errors via the refreshed CRT graphics terminal. To this can be added the ability to generate a graphical simulation of the cutting operation (trajectories of center of ball-end cutter) on a graphics terminal CRT.

6.3 A Proposed Method for Obtaining an NC Cutter Path for Milling a Three-Dimensional Curved Surface

We have developed an interactive graphics program which can generate any three-dimensional surface. One product of this system is the availability of any section through the surface. It further provides the coordinates of any point on the section where it transects the surface.

The algorithm used for our graphics system has special features which make it uniquely adaptable to this problem. Two possible modes could be used.

6.3.1 Mode 1 - Tangent Point Follows the Section Line

In this approach the nominal position of the cutter is offset so as to hold the tangent point on the section line. It is proposed that the cutter path be defined so as to cut along tangent lines corresponding to any convenient set of sections - usually a closely spaced set parallel to one coordinate plane.

Figure (6-6) illustrated such a transection, and Figure (6-7) shows a cross-section at point P_t parallel to the y-z plane. In order to establish tangency at P_t the cutter must be offset from the nominal position along the section an amount Δy . A similar offset Δx must occur in the x-z plane.

The two offsets, and the vertical location of the cutter centre point C, can be treated as optimization variables, and a nonlinear programming technique used to minimize the difference between the vector CP_t, R_v, and the cutter radius, R. At the correct location this criterion quality should reduce to nearly zero.

6.3.2 Mode 2 - Cutter Path Follows the Section Line

In this approach the cutter is maintained in the section, and the tangent path is allowed to wander as necessary. Only the height of the cutter need be determined.

This can be done by using our patch splitting facility to generate a new micro patch with 16 points in the quadrant containing the tangent point. This quadrant can be determined from the tangent lines T_1 and T_2 , shown in Figure (6-8).

We designate the ith point in the patch as P_i , and the distance from C to P_i as L_i . An optimization strategy is now used with two stages. For a given ΔR we identify the P_i which gives minimum L_i -R. We then use a second

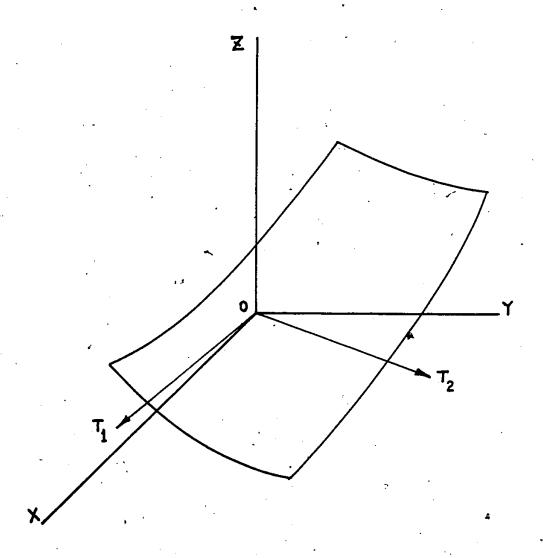


Figure (6.6) Three-dimensional surface

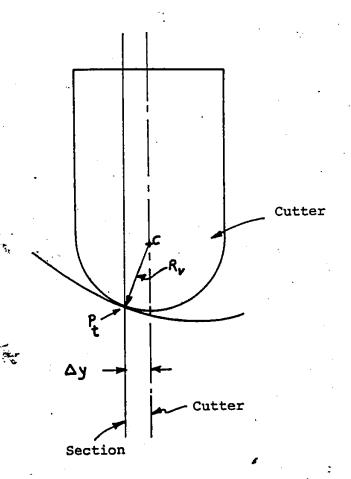


Figure (6.7) Section through surface parallel to y-3 plane

one-dimensional search startegy to adjust ΔR so that (L_i-R) is nearly zero. The mesh must be fine for adequate accuracy, and the tangent point may be off the patch. This will be observed if (L_i-R) does not reduce to zero, and then a second adjacent patch can be used.

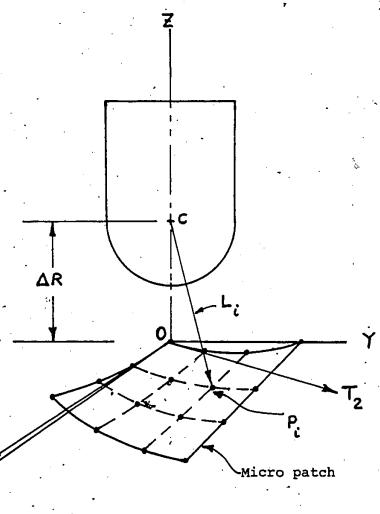


Figure (6.8) Micro patch used to determine AR

CHAPTER 7 CONCLUSION

Shape is one of the the most important variables in engineering design, and the computer aided design of 3-D shapes is an active field. In any computer-based shape handling system there must be two aspects: the definition of shape, and the interrogation of shape. These two aspects, although distinct, are completely interdependent and neither should be stressed to the exclusion of the other.

In Chapter 1 of this thesis we have investigated the already existing computer aided design graphics systems for interactively creating three dimensional curved surfaces. In almost all of these systems surface definition algorithms were developed from the Coon's patch * algorithm where, as was described earlier in this thesis, 48 coefficients are required to define a patch [77] . To define a patch, slope vectors and twist vectors are involved in the design process. The user has to define or give values to these vectors in order to generate the patch surface. Designers usually have great difficulty dealing with such slope and

^{*} Appendix D

twist entities. On the other hand, surface modification requires redefinition of these entities. The effect of changing one element of the slope or twist vectors is unpredictable and consequently the interaction between the designer and the computer model was troublesome.

The first aim of this thesis was tobtackle the previous problem. Surface patch definition was achieved using only a grid of 16 spatial points lying on the patch surface. Surface modification is done dynamically by relocating the position of any of the predefined mesh points (or what we called control points) using a refresh type graphics terminal CRT, [78].

One of the more difficult problems associated with the design of an interactive system is to make it easy to use - the more facilities the system has, the more difficult it is to make them available to the user. The free form surface design program developed in this thesis proved to be very easy and simple to use in creating three-dimensional surfaces. The mathematics involved in creating such surfaces is completely hidden from the user so that he can direct his attention only to the design process. The system provides the designer with a powerful and integrated set of design tools (smoothing, blending,...) which aids in the design and refinement of any curved surface. It is worth reviewing the traditional drawing board method of designing surfaces, in order to highlight the difficulties that have to be overcome by designers. As a special application,

let's examine how a ship's hull might be tackled on a drawing board. The first stage would be to lay down the principal profiles and sections in a three-view drawing.

. In many cases these would be laid down within specified constraints - length, breadth, depth, etc. - but the designer would be exercising considerable freedom of choice within these. The next stage is to add more sections to the drawing in order to specify the surface in more detail. This is done by using a graphical fairing technique that is essentially iterative. The designer constructs some diagonal planes which intersect the sections that he drew in the first stage and are so arranged that the section line is approximately normal to the diagonal plane. The fairing procedure consists of fitting a spline curve through the known points of this diagonal line which is then used to interpolate the shape of intermediate sections. is to construct these intermediate [797 final stage sections and then to construct the shapes of waterlines and buttocks. Very often these constructed sections show hollow and humps that should not be there. To remove them requires another cycle of the fairing process starting at Figure 7-2. There are obvious disadvantages associated with the traditional method of designing surfaces, especially ship hulls, based on the following reasons:

Designing surfaces by hand requires design draughting skills of a high order. Given these skills it is still a long job to specify the surface in sufficient

detail to manufacture the design.

This type of procedure does not design the surface, it merely designs lines on the surface. Very often further constructions are necessary in order to obtain manufacturing information. This of course is time-consuming and could lead to inaccuracies.

Complete freedom of choice is only being exercised at the first stage [77] . The design becomes more and more constrained as the design proceeds until the final stage is merely a graphical construc-

The system developed in this thesis overcomes all the above objections. a special ship hull design program was designed and proved to be an efficient and easy to use design program. Aesthetic, geometric designs (bottles, furniture, pipes, nozzles, etc.) also proved to be a rich field for the applications of the system.

tion, with no scope for applying design talent.

A new technique for generating the cutter path for NC manufacturing of free form surface was proposed in this thesis. A literature survey indicates that this technique could solve many problems in manufacturing sculptured surfaces. Chapter 6, which contains a description of this technique) could be a basis for further research. Chapter 6 of this thesis also suggests an interface between the CAD system developed in this thesis and any existing CAM system, in order to achieve an integrated CAD/CAM system, [80].

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APPENDICES

USER'S MANUAL

Program JOLIA

Function 🖼

This program aids in the design of any curved surface using input data only from the light pen.

User's Manual

We will consider here a practical design of a smoking pipe using two patches. The main steps required to design the pipe are as follows:

- 1. User starts up the PDP-11/34 minicomputer.*
- User runs the program, by typing.

.RJN RK2: JOLIA < CR

- 3. Program responds by typing a menu of design commands on the CRT graphic terminal, Figure (5.25).
- 4. User hits the 'INTACT DESIGN' command with the light pen.
- Program responds by typing the following message;
 'PLEASE ENTER THE NUMBER OF PATCHES'.
- 6. User enters*

2 ° ≪R >

7. Program responds by typing the following message:

'PLEASE ENTER NUMBER OF PATCHES PER WORKING PLANE.

^{*} H.A. ElMaraghy, "Operating Procedure of PDP 11/34 Minicomputer And GT-46 Graphic Terminal, McMaster Univ., 1977.

- User enters
 - 2 <CR> (i.e. the points defining the first and second patch lie on the same working planes).
- Program responds by showing the X-Y-Z axis and a menu having two commands, 'DEFINE WORKING PLANE' and 'DONE'.
- 10. User hits 'DEFINE WORKING PLANE'option with the light pen.
- 11. Program responds by typing:

 'DEFINE PATCH IN PLANE NUMBER 1'

 'POSITION TRACKING OBJECT ON ANY POINT OF THE

 THREE AXIS TO DEFINE WORKING PLANE NUMBER 1'.
- 12. User positions the tracking cross on the z axis and hits 'RETURN' key on the TT.
- Program respond by showing a working plane parallel to the x-y plane and passing with the point the user has positioned the tracking cross at, on the z axis, as shown in Figure (3.7).
- 14. User hits 'DONE' option with the LP.
- 15. Program erases the 'DEFINE WORKING PLANE' and 'DONE' options and shows another two options 'POSITION' and 'DONE'.
- 16. User tracks the tracking cross with the LP and position it (the tracking cross) at any point on the working plane, then hits the 'POSITION' option with LP.

- 17. Program responds by plotting a dot at the predefined position.
- 18. User repeats the 16 and 17 steps, seven more times (4 points per patch per plane); user should make sure that point number eight should lie exactly over point number one since the surface is to be closed as shown in Figure (5.25).
- 19. User hits the 'DONE' option.
- 20. Program respond by typing:

 'DEFINE PATCH IN PLANE NUMBER 2' and shows the

 'DEFINE WORKING PLANE' replacing the 'POSITION'

 option.
- User continues defining the points defining the two patches at different working planes until he finishes defining 32 points (16 per patch), as illustrated in steps 1 to 20, as shown in Figure (5.25).
- 22. User hits the 'DONE' option two times.
- 23. Program responds by plotting the surface passing with the predefined 32 points as shown in Figure (5.25).
- 24. If the user is satisfied with his design he can hit the 'SAVE' option, the program will ask for a file name, and the user can enter any name, e.g. PIPE.

- 25. User can now hit the 'EXIT' option since he is finished with the design.
- 26. If the user wants to have the data describing his surface, he should enter the following command on the TT:

.TYPE <CR>

computer will respond by typing

FILE NAME?

user enters PIPE < CR>

program types the x, y, z coordinates of the points defining the surface.

HIERARCHICAL CHART AND LISTING FOR PROGRAM
JOLIA

A hierarchical input-processing-output HIPO chart for the program JOLIA is shown in Figure (A.1). The function of each module is briefly described as well as the input and output of each one. Figure (A.2) shows the three overlay regions of the program. A listing for the program is also included. The program modules were "linked" to the DEC-graphics library "GLIB", the linking procedure is included in the file "JOLIA.COM". A listing for that command file is shown in Figure (A.3). In case of any extension or motification of any of the program modules, the programmer should edit the modified module, compile it and then type:

@JOLIA.COM <CR>

to perform the linking operation.

Subprogram Level 3

8

Accepts input data from light pen. Calls MENUH to choose a design option.

First option is interactive design of curved surfaces using input from LP. calls INTACT to perform the first option

surfaces or surfaces of

revolutions, X-Y co-

ordinate from LP.

define either curved

calls DRW-JOL to draw the

designed patch calls TEST 1 to create the boundary vectors of the designed patch.

Output - x, y, z, co-ordinates of the control

points defining the

surface patch.

Second option is to smooth two patches at a common border.

calls SMOOTH to perform the

either points or working

planes.

DRWJOL

Calls MENUH to define

second option. Calls DRW.JOL to draw the

smoothed patch.

Third option is to create surfaces of revolutions.

calls INTACT to define axis of rotation and the control points of the surface of revolution.

Calls ROTATE to generate the the surface from the control est of the points defining points defined in the

previous step A. Calls DRW, JOL to draw the surface of revolution. ပ

Fourth option is to split a predefined patch.

Input - number of patches, 6

ordinates of a patch

using input from LP

Creates x, y, z, co-

INTACT

number of patches per working plane, flag to

MENUH

(by LP) subpicture ased to return the tag of a hidden

> boundary vector, array containing the Z boundary the patch, flag to decide rray containing lines and control points Used to draw the patch. either to draw lines or vector, the tag of the ndary vector, array containing the Y of a patch, number of patches per working subpicture containing the x bo Input plane.

the control points of the x, y, z coordinates POINTS Calls POINTS to create x, y, z coordinates of

of N points lying on used to calculate a patch surface

Calls Test 1 to create the Calls DRW. JOL to draw the boundary vectors of the splitted patch. ပ æ

Fifth option is to create curves splitted patch.

A - Calls INT-JOL to generate of intersections of a patch with planes.

the curve of intersection. x, y, z coordinates of

Sixth option is to modify a patch.

patch number (tag). Calls TRAKXY to define x, y Calls LPEN to define the 4 M

coordinates of the new point Calls LPEN to define the

number of the point to be relocated.

Calls TEST 1 to create the boundary vectors of the modified patch.

Calls DRWJO1 to draw the new Seventh option is to blend two adjacent but non touching (modified) patch. ш

blending functions of the Calls GETB to get the patches. ď

the blending functions of Calls GETB to get the second patch. first patch.

boundary vectors of the third blending functions of the third (blended) patch. Calls GETBX to create the Calls BLEND to create the A

Output - image of the patch on CRT.

TEST 1

Used to calculate the x, y and z boundary defined by 16 x, y, vectors of a patch coordinates. nput - x, y, z coordinates defining the patch.

Calls LENGTH to get the length between input points.

Calls CMATRX to define the linear system of equation parametric products. Calls SOLVE to solve the elements of the matrix for boundary vectors. containing the U, W

boundary vectors defining Output - x, y and z the patch.

points.

SMOOTH

adjacent patches along used to smooth two the common border.

flag to determine whether containing x, y and z boundary vectors of the patch before smoothing, patches to be smoothed, to smooth along longit-Input - three arrays udinal or transverse number of defined borders.

Calls IDNTFY to define the tags of the two

obtained from

functions

patch from the blending

metric values of the points, boundary vector (x or y or z) W arrays defining the patch, defining the paranumber of points

or z) coordinates of containing (x or y Output - array the points. . E

LENGTH

ength 16) containthe length between points (16 points) coordinates of 16 used to calculate each successive Input - x, y, z Output - array ing the length between each 2 points.

tains U, W parametric products) using the parametric bi-cubic the C matrix (conused to calculate oatch expanded equation. CMATRX

٥

Input - two arrays containing U and W parametric values, C matrix number of data Output ooints.

SOLVE

used to solve

the blending functions obtained from the previous step C. E - Calls DRWJOL to draw the third (blended patch).

display and create a correspond-Bighth option is to save the x, y, z coordinates of the Calls OBSAVE to save the displayed patches. ing data file.

Nineth option is to recall the saved display Calls GETSHP to recall x, y, z coordinates of the saved display.

Tenth option is to erase a predefined patch.

- Calls LPEN to define the number (kag)

to be erased. Calls ERAS to erase the atch.

Calls FREE to stop and exit from Eleventh option is to exit. the program.

of the two patches to be Calls GETBX to calculate Calls GETB to calculate the boundary vectors of the smoothed patch. patches to be smoothed. the blending functions smoothed.

boundary vectors of the smoothed patch. Output - x, y and z

ROTATE

used to perform rotation of points about any arbitrary axis. Input - number of x, y,

triplets, x, y, z coordinates of the points to be rotated, direction cosiness of axis of rotation, rotation angle in degrees.

y, z coordinates matrix with transformation Call's MU to multiply x, matrix. Output - x, y, z coordinates of the rotated oints.

GENJOL

division of a given patch used to create the x, y, z coordinates of a patch generated by the sub-

Input - x, y, z boundary veçtors of the patch e splitted

Gauss elimination. system of linear equations using

nand side, the M by M matrix of the right coefficient matrix, Input - The M by N system, the number of right hand side vectors, relative equations in the the number of tolerance.

Output - solution of the system, flag if zero there is no error, if -1 there is no solution.

used to identify the IDNTFY

Input - number of patch's number. already defined Calls MENUH to patches.

identify the patch's Output - the hidden (with LP) patch's number. number.

GETB

used to calculate the boundary matrix of a Input - x, y and patch.

boundary vectors of the patch. Calls MU to perform matrix multiplications

used to multiply general matrices

 \mathbf{E}

number of rows in the second matrix, number of rows in the first of the second matrix name of the resultand columns of the ant matrix, number first matrix, name Output - elements resultant matrix. Input - name the of the resultant of rows in the second matrix, matrix. Output - x, y and z boundary matrices of used to calculate the boundary matrices of boundary vectors of Calls MU to perform matrix multiplic-Input - x, y and the patch. the patch. a patch. ation. GETBX Output - x, y, z coordinates of the patch used to draw the resultcreated by subdividing splitting) the given from the intersection ant continuous curve coordinates of the of a plane and a calculate the x, Calls POINTS to created patch. TOFINI patch.

Input - x, y, z boundary vectors defining the patch.

Calls POINTS to calculate the x, y, z coordinates of the curve of intersection.

Output - x, y and z boundary vectors of

patch.

the patch.

Output - the x, y, z coordinates of the curve of intersection, and an image of that curve on the patch at CRT.

BLEND

used to blend two non adjacent patches by creating an in-between patch, smoothed to the predefined two patches. Input - boundary matrices of the two patches to be blended.

Output - boundary matrix of the in-between

(generated) patch.

OBSAVE

used to save the display in a data and a display file.

Input - x, y, z coordinates of the patch (or patches) to be saved. Calls INFILE to define the file name.

Output - data file containing x, y, z coordinates of the patch to be saved.

INFILE used to accept a

file name.

Input - number of patches to be saved or recalled, name of

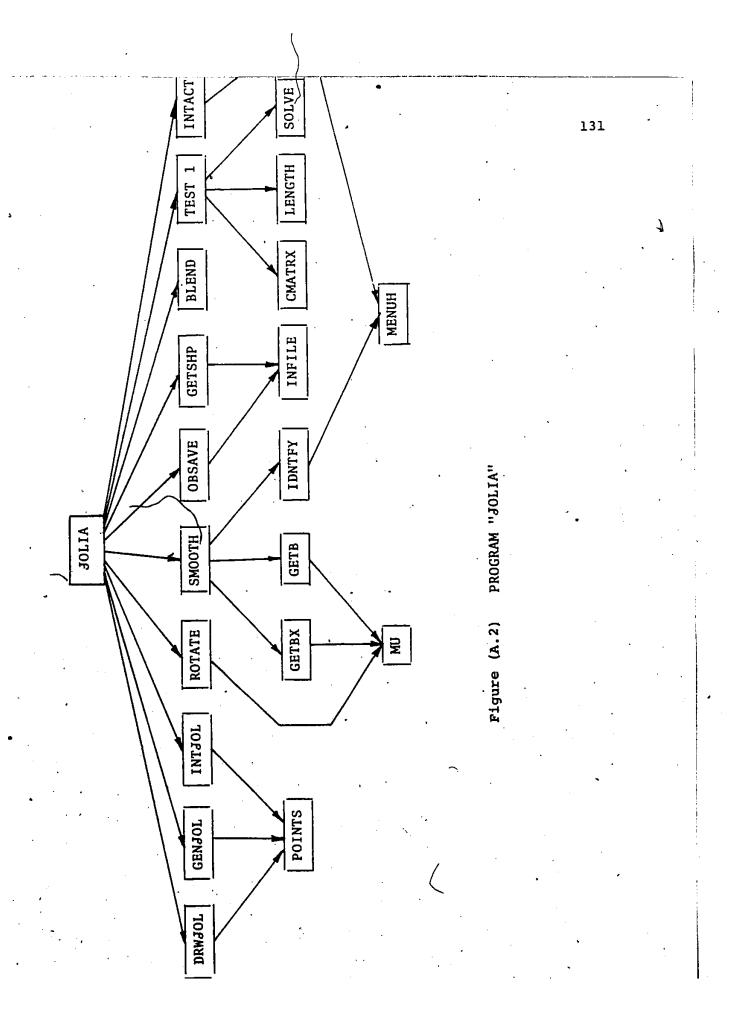
GETSHP

used to recall a previously saved display.

Calls INFILE to define the name of the file to be recalled.

Output - image of the saved display on CRT.

Figure (A.1) a Hierarchical Chart For Program 'JOLIA'



```
DIR TU
TYPE
Files? JOLIA.COM
R LINK
RK2:JOLIA, JOLIA=RK2:JOLIA, RKO:GLIB, FORLIB//
RK2:GENJOL/O:1
RK2:INTACT/0:1
RK2:ROTATE/0:1
RK2:SMOOTH/0:1
RK2:BLEND/0:1
RK2:0BSAVE,GETSHP/0:1
RK2:INTJOL/0:1
RK2:TEST1/0:1
RK2:DRWJOL/0:1
RK2:GETB/0:2
RK2:GETBX/0:2
RK2:IDNTFY,INFILE/0:2
RK2:CMATRX/0:2
RK2:LENGTH/0:2
RK2:SOLVE/0:2
RK2:POINTS/0:2
RK2:MENUH, MU/0:3
11
GT OFF
RU JOLIA
```

Figure (A.3) Listing of JOLIA.COM

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Program KEY

Function

This program aids in the preliminary design of any curved surface using input data only from the keyboard.

User's Manual

The program main flow from the user's point of view is as follows:

- 1 User starts up the mini-computer.
- 2 User runs the program, he types:

RUN RK2: KEY

- 3 Program responds by typing a menu of design commands on the CRT graphic terminal.
- 4 User hits the 'KEY BOARD' light button with LP.
- 5 Program responds by typing the following message:
 'PLEASE ENTER THE NUMBER OF PATCHES'
- 6 User enters the number of patches required to define the surface, e.g.

<u>l</u> < CR >

- 7 Program responds by typing:
 'ENTER 16 x,y,z COORDINATES OF PATCH NUMBER 1'
- 8 Enter the coordinates e.g.

10., 100., 50. <CR>
20., 30., 40. <CR>
501., 20., 60. <CR>
200., 300., 121.

- 9 Program responds by displaying the patch of CRT, passing by the predifined points, together with the menu of design commands.
- 10 User is ready now to make use of the design light button commands till he is satisfied with his design.

HIERARCHICAL CHART AND LISTING FOR PROGRAM KEY

A [HIPO] chart of program KEY is shown in Figure (A.4). Figure (A.5) shows the three overlay regions of the program. The modular concept has been preserved during the design stage of the programs presented in this thesis and that's why we made use of some of the modules already designed for program JOLIA, in program KEY.

Subprogram

Calls MENUH to choose a design

option.

board.

Accepts input data from key

Main Program (KEY)

First option is to input data points defining the patch via

Cálls KEYBRD to return to

board.

key

the main program the

used to return to the main defined via keyboard program the x, coordinates

of a patch or a group of patches, defined via key

Calls TEST1 to create the

board.

Output - number of coordinates of the patches, x, y and

TEST1

GETBX GETB

KEYBRD

and z

patch.

LENGTH CMATRX SOLVE

DRWJOL

Calls SMOOTH to perform the

Calls DRWJOL to draw the "

second option.

Third option is to split a

predefined patch.

smoothed patch.

Second option is to smooth two

د

patches at a common border.

Calls DRWJOL to draw the

t, y and z boundary rectors of the patch.

SMOOTHA

IDNTFY

POINTS

GEN401

alls TEST1 to create the boundary vectors of the splitted patch.

the control points of the

Calls GENJOL to create

(created) patch. Calls DRWJOL to draw the created patch. splitted

is to create carves of intersection of patch with planes Fourth option

夏

	•		•	
		•	BLEND	30 (1) (1) (1) (1) (1) (1) (1) (1) (1) (1)
Fifth option is to modify a patch. A - Calls LPEN to define the patch number (tag). B - Calls TRAKXY to define x, y coordinates of the new noint		ion is to but not to ith a thir s GETB to ding funct	lls GETB to ge indary matrice cond patch. lls BLEND to c anding functio	rd (ble 1s GETB ndary v rd patc 1s DRWJ

OBSAVE

Seventh option is to save the

display and create a

corresponding data file.
- Calls OBSAVE to save the x, y and z coordinates of the displayed patch (or

patches).

Eighth option is to recall the saved display. Calls GETSHP to recall the x, y and z coordinates of the saved display.

GETSHP

is to erase Nineth option

number (tag) of the patch to be erased. a predefined patch. A 1- Calls LPEN to define the Calls ERAS to erase the

Tenth option is to stop and exit from the program.

patch.

A Hierarchical Chart For Program 'KEY' Figure (A.4)

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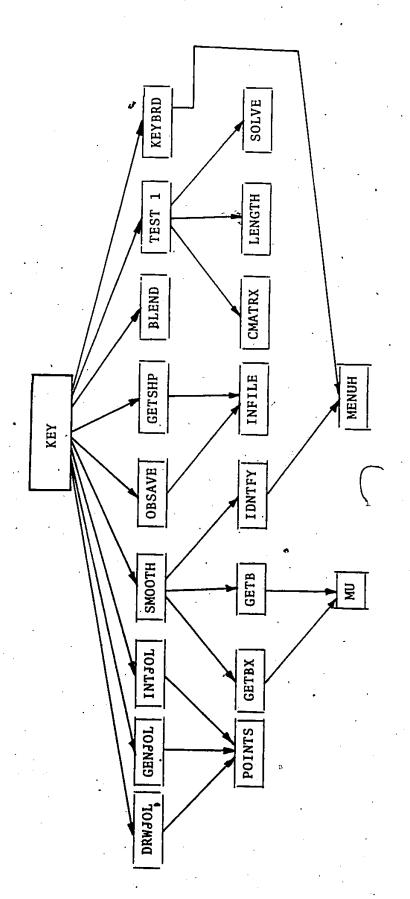


Figure (A.5) PROGRAM 'KEY'

Program DATAFL

Function

This program aids in the preliminary design of any curved surface using input data only from x,y,x coordinates stored on data files.

User's Manual

The program main flow from the user's point of view is as follows:

- 1 User starts the mini-computer.
- 2 User runs the program, type:
 - RUN DATAFL < CR >
- 3 Program responds by showing a menu of design commands.
- 4 User hit the 'DATA FILE' light button with the LP.
- 5 Program responds by typing the following message:

 'PLEASE ENTER THE NUMBER OF PATCHES (DATA FILES)'
- 6 User enter the number of data files e.g.

_1 < CR >

7 - Program responds by typing:

'ENTER THE NAME OF FILE NUMBER 1'

8 - User enters the file name (containing the data points defining the patch) e.g.

PATCH 1 <CR'>

- 9 Program responds by displaying the patch surface and the design commands menu.
- 10 User is now ready to modify the surface; using the light button commands; till he is satisfied with the design.

6-3-6 HIERARCHICAL CHART AND LISTING FOR PROGRAM DATAFL

A [HIPO] chart for program DATFL is shown in Figure (A.6). Figure (A-7) shows the three overlay regions of the program. Listing for program DATAFL, together with listing of the command file DATAFL.COM are also included.

Accepts input data from stored data files containing x,y,z

coordinates defining the sur-

face.

Calls MENUH to choose a design

option.

First option is to input data

points defining the surface

patches via data files.

LEVEL 2

SUBPROGRAM

SUBPROGRAM LEVEL 3

LEVEL 1

used to return to the x,y, and z co-ordinates of a the main program DATAFIL

data fîles (patches) Input- number of

ordinates stored on Output- x,y,z cothe data files.

of patches, stored patch or a group on data files.

> the one of program KEY, except design option.

MENUH

the main program the x,y,z coordinates of a patch or a group of patches, stored A. Calls DATFIL to return to on data Files.

The rest of this hierarchical chart is exactly the same as that here we have an extra

Tenth option is to rotate the designed surface about any arbitrary axis.

. Calls INTREV to return to the main program the x,y,z coordinates of the two points defining the axis of rotation.

INTREV
used to return to
the main program
the x,y,z coordinates of the two
points defining
the axis of rotation.

Output- x,y,z coordinates of the two points defining the axis of rotation.

B. Calls ROTATE to rotate the predicted points describing the surface about the axis of rotation.

C. Calls TEST 1 to create the new x,y and z boundary vectors of the rotated patch.

D. Calls DRWJOL to draw the rotated patch.

Eleventh option is to exit and stop the program.

Figure (A.6) A Hierarchical Chart for Program DATAFL

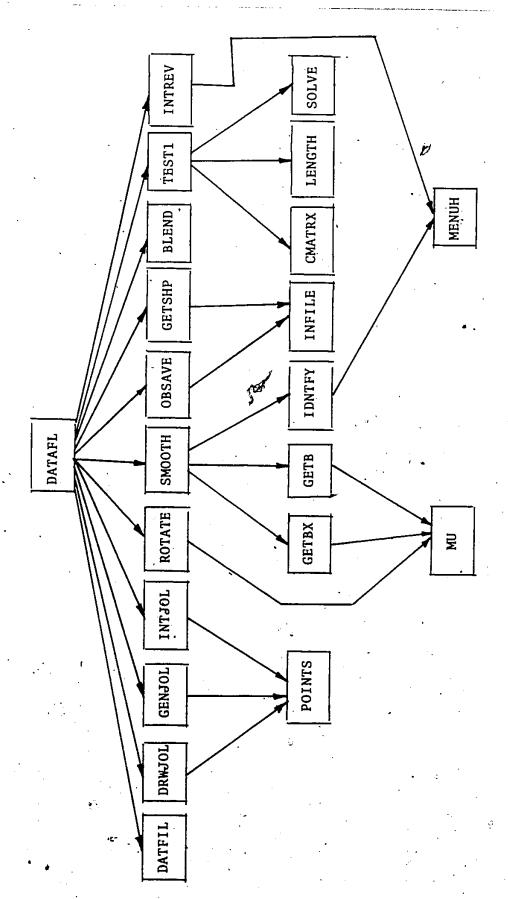


Figure (A.7) PROGRAM "DATAFL"

PROGRAM SEC1

FUNCTION

This program allows the user to perform shape modification of a certain patch. It also allows him to interactively translate his patch and to obtain the resultant continuous curve from the intersection of a plane with his patch.

USER'S MANUAL

- 1 create your own data file containing the X, Y, Z coordinates of 16 points defining the patch, to do so
 - a start up the computer
 - b R EDIT
 - * EWDXL: NAME. DAT \$\$
 - * I X1, Y1, Z1 X2, Y2, Z2

X16, Y16, Z16

<u>\$\$</u>

EX\$\$

Remarks

- NAME is the name of your data file
- The underlined commands are commands entered by the user through the DECWRITER,
- DX1 is a user floppy disk, if the user wants to load his data file on any other device he should replace DX1 by the device name (e.g. RK2 for a magnetic disk).
- 2 Run program SEC1, write

. RUN RK2: SEC1

The program will respond by writing
THIS PROGRAM ALLOWS PATCH MODIFICATIONS
ENTER THE DATA FILE NAME

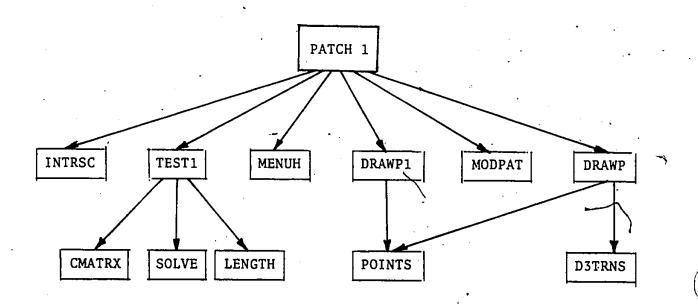
*DX1: NAME.DAT

The program will respond by drawing the sixteen points with straight lines joining each successive point. A menu area will be seen on the left hand side of the CRT containing four options, move point, intersect, translate and done (see Figure A.8).

MOV PT
INTRSC
TRNSLT
DONE

Figure (A.8) A typical plotting of a patch on the CRT showing the menu area

- 4 The user is free now to choose anyone of the options on the menu area by just pointing at it with the light pen.
- 5 To aid in understanding the option 'MOV PT'
 see SUBROUTINE MODPAT, 'INTRSC' see SUBROUTINE
 INTRSC, 'TRNSLT' see SUBROUTINE DRAWP and D3TRNS.
- At any time the user can exit by pointing at



PROGRAM GENRAT

FUNCTION

This program allows the user to generate a patch from a given patch.

USER'S MANUAL

- 1. Start up the computer.
- 2. Run program GENRAT type

.RUN RK2: GENRAT

The program will respond by asking the user to enter x, y, z coordinates of the original patch (i.e. define patch).

'ENTER X, Y, Z COORDINATES'

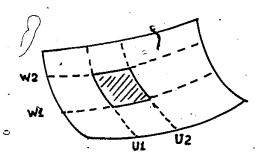
- 4. The user should respond by typing the x, y, z coordinates of a 16 points defining his patch.
- 5. The program will then display the patch on the CRT in the 3-D space.
- of the patch to be generated from the given patch.

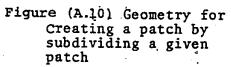
 'ENTER U9L), U(2), W(1), W(2)'
- 7. The user should respond by supplying these values e.g.

0.2, 0.6, 0.2, 0.6

< CR >

The program will generate the new patch on the original patch (see Figures A.9 and A.10).





The said and the said of the

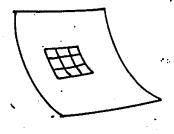


Figure (A.11) PC patch subdivision

and reason to be a first of the same of the same of

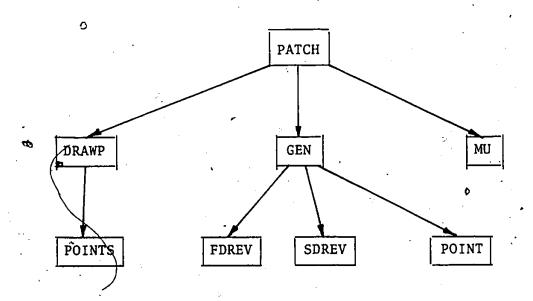


Figure (A.12) FLOW CHART OF PROGRAM 'GENRAT'

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i pagada kangga sempera dan kipada beli 😂 dan dan da

and the second of the second of the

FUNCTION

This program performs a hidden line removal operation. User enters the patch file name, program displays the patch with hidden lines removed.

USER'S MANUAL

- 1. Create your own data file containing the X, Y, Z coordinates of 16 points defining the patch surface.
- 2. Run program HIDDN, type

.RUN RK2: HIDDN <CR>

The program will respond by typing,

'THIS PROGRAM SHOWS THE PATCH WITH HIDDEN LINES REMOVED'.

'ENTER THE DATA FILE NAME'.

3. User enters the data file name, e.g.

FILNAM <CR>

4. Program immediately displays the patch with hidden lines removed.

APPENDIX (B)

CURVE DESIGN PROGRAM

INTRODUCTION

An interactive computer program was designed to aid in the use of computer graphics in the design of curves (two-dimension).

The main purpose is to simplify the design process so that the designer is not required to know things not particularly relevant to his role in the design process. The requirements of the formulation were that it must automatically maintain curve continuity and yet allow changes in shape information to be specified by moving control points (using LP) that affect the curve in an intuitive way.

CURVE DESIGN

To satisfy the real time constraint, the curves must be computable by a very fast algorithm. This requirement was satisfied by using the parametric basis - splines [B - spline], [78].

A B-spline curve "intuitively" mimics the shape of control polygon, which is an ordered sequence of points (1, 2, 3,..., N) as shown in Figure (B.3). The curve that follows the shape of this control polygon is composed of a sequence of spline segments. The designer

enters the control points on the CRT (using LP), the program generates the piecewise curve using a continuous first derivative B-spline curve as shown in Figure (B.3).

A menu area on the CRT is assigned to give the user the ability to interact with the program as shown in Figure (B.1).

D.

EXAMPLE ON HOW TO USE PROGRAM CURDES

- 1. START UP THE COMPUTER
- 2. Type

.RUN RK2: CURDES

3. A menu containing different options will then be drawn on the CRT (Fig. (B.1).

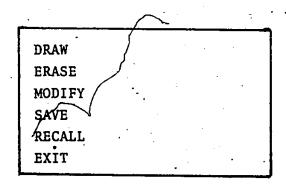


Figure (B.1) Menu Area

- 4. The user can now point at 'DRAW' with the light pen.
- 5. Another menu will be drawn on the CRT (Fig. B.2) and also a tracking object will be seen on the center of the CRT.

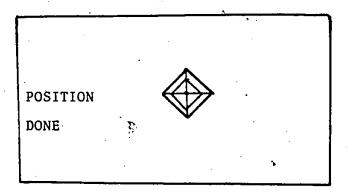


Figure (B.2) Menu area after the user has pointed at 'DRAW'

6. The user can position the tracking object at any point on the CRT and then press < CR > (RETURN KEY) to enter the control points that control the shape of 2-D curve, after he is done, he can point at 'DONE'.

The program then generates the curve automatically (Fig. (B.3)).

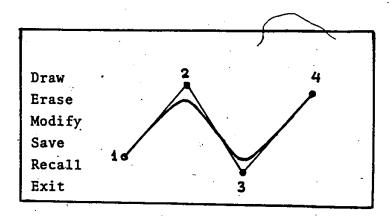
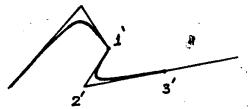


Figure (B.3) The control points (1, 2, 3, 4) and the resulting curve

. The user can then choose any other option (including 'DRAW'); he can erase, modify, save and recall his curve. It is worth mentioning here that the 'MODIFY' option will allow the user to erase any part of his curve. Suppose that the user has designed the following curve (Figure B.4).



and he wants to modify that curve to the following (Figure B.5).



He can do that by splitting the lines at points 1 and 2 and then erasing the curve in between, then using the draw option to enter the new control points 1', 2' and 3'.

APPLICATIONS '

The program can be used in an interactive way to design value curves [*] relating to consumer products. It can be also used to draw contour lines of any figure. See Figure (B.6).

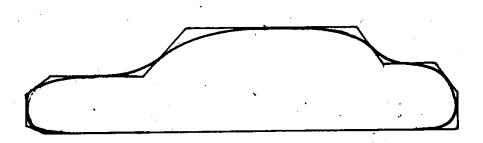


Figure (B.6) Demonstration of the possibilities of B-spline approximations

*J.N. Siddall, 'Value Theory and User Participation, Architectural Design, Vol. 42, No. 5 (May 1972), pp. 319-322.

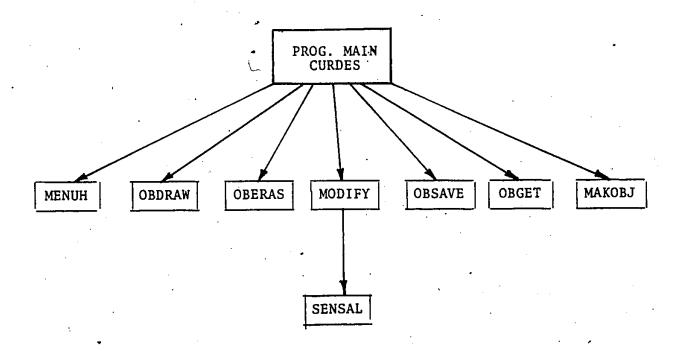


Figure (B.7) FLOW CHART FOR PROGRAM CURDES

PROGRAM CURDES .

FUNCTION

- This program can be used to design curves in 2-D.

The program sets a menu area on the screen showing the options of

DRAW ERASE MODIFY SAVE RECALL EXIT

If the user points (with the light pen) at DRAW, a tracking (diamond shape) object will appear on the screen, and a message will be written on the Decwriter:

POSITION THE TRACKING OBJECT, TYPE < CR> WHEN DONE.

so the user can enter his control points (see Figure 3).

USAGE

.RUN RK2: CURDES < CR>

FORTRAN

LANGUAGE

SUBROUTINE MENUH (IT, M1, M2)

FUNCTION

- This SUBROUTINE detects

a light pen hit on one of the options in the menu area.

USAGE

- CALL MENUH (IT, M1, M2)

PARAMETERS

IT - Tag of the hidden subpicture

Mì

- Tag of the first option in the

menu area.

M2 .

- Tag of the last option in the

menu area.

LANGUAĜE

- FORTRAN

SUBROUTINE QBDRAW

FUNCTION

- This subroutine draws a curve on the CRT, by positioning the control points.

USAGE

- CALL OBDRAW.

LANGUAGE

- FORTRAN

ALGORITHM

This subroutine uses the cubic spline algorith to draw a curve. For more information about cubic splines see reference [*]

*George J. Peters, 'Interactive computer graphics applications of the parametric BI-CUVIC SURFACE', McDonnell Douglas Automation Company, St. Louis, Missouri.

SUBROUTINE OBERAS

FUNCTION

- This subroutine erases an object out of the CRT.

USAGE

- CALL OBERAS

LANGUAGE

FORTRAN

SUBROUTINE MODIFY

FUNCTION

This subroutine modifies
an object on the CRT.
The user gets two options
if a call is made to MODIFY,
erasing any line and or,
splitting a line, when he is
done, he can point with the
light pen on the option
'DONE', then a RETURN is
made to the main program
CURDES.

USAGE

LANGUAGE -

CALL MODIFY

FORTRAN

SUBROUTINE OBSAVE

FUNCTION

- This subroutine saves the display in a file named by the user. When a call is made to OBSAVE, the program asks for a name to assign to the file to be saved.

USAGE

- CALL OBSAVE

LANGUAGE

. SUBROUTINE OBGET

FUNCTION

- This subroutine recalls
the file saved by OBSAVE,
i.e., restores the display.

USAGE

- CALL OBGET

LANGUAGE

SUBROUTINE SENSAL

FUNCTION

- This subroutine turns the light pen sensitivity on and off for all the objects.

USACE

- CALL SENSAL

LANGUAGE

APPENDIX (C)

SHIP HULL FORM COEFFICIENTS

FORM COEFFICIENTS

In order to refer to certain proportions of ships; to compare them in form with regard to their actual dimensions or difference in dimensions; to be able to describe their shapes more precisely than "fat" or "thin", "full" or "fine"; there are certain geometric qualities that can be related as ratios or dimensionless coefficients. These coefficients of form are exceptionally useful in comparing certain performance characteristics associated with hydrodynamic phenomena.

In the following relationships, the symbols used are defined as follows:

L_{pp} - length between perpendiculars or designed waterline length

T - draft to the waterline, or draft /

B - beam or breadth molded

displacement volume at draft T

area of midsection at draft T

- area of waterplane at draft T

The coefficients most commonly used by naval architects are as follows:

Midship section coefficient

$$C_{m} = \frac{Am}{BT}$$

Block coefficient

$$C_{b} = \frac{\Delta}{L_{pp}BT}$$

Prismatic coefficient

$$C_p = \frac{\Delta}{A_M L_{pp}} = \frac{\Delta}{C_M BTL_{pp}} = C_b/C_M$$

Waterline coefficient

$$C_{WP} = \frac{Aw}{BL_{pp}}$$

There are also certain commonly used ratios of dimensions, and these with their approximate range of values are:

Length - beam ratio

range, 3 to 12

Length - draft ratio

range, 7 to 30.

Beam - draft ratio

B/T

range, 1.8 to 4

Displacement-length ratio

Displacement - length coefficient

range, 50 to 500

APPENDIX D

COON'S PATCHES

APPENDIX (D) COON'S PATCHES

In Coon's notation points on the surface of a bicubic patch satisfy

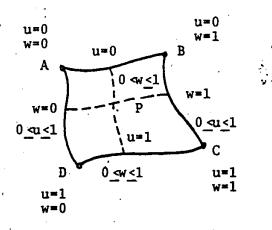
$$UW = U M B M^{T} W^{T}$$

$$= [u^{3} u^{2} u 1] \begin{bmatrix} 2 & -2 & 1 & 1 \\ -3 & 3 & 3 & -2 \\ 0 & 0 & 1 & 0 \\ 1 & 0 & 0 & 0 \end{bmatrix} \begin{bmatrix} 00 & 01 & 00W & 01W \\ 10 & 11 & 10W & 11W \\ 00U & 01U & 00UW & 01UW \\ 10U & 11U & 10UW & 11UW \end{bmatrix}$$

The so-called "boundary conditions matrix" is in fact a tensor - each entry above (e.g., 00 01W, etc.) is a vector. The corners (points A, B, C and D) of the patch (00 01 11 10) are vectors relative to the origin (and axes) of the design, the slope vectors (00W 01W 01U 11U), etc. are vectors relative to their respective patch corners. Further, the twist vectors are (best thought of) relative to the point on the parallelogram completed from respective slope vectors.

A qualitative description of Coon's patches may be useful. A patch has four boundary curve segments (edges) which

meet at four points (patch corners) in the fashion suggested by Figure D.1.



x = f(u,w)

Point P is at y = g(u,w)

z = h (u,w)

Figure D.1 Parameters on Coon's Patch

APPENDIX (E)

THE MATHEMATICAL FORMULATION OF SMOOTHING TWO PATCHES AT THE COMMON BORDERS

APPENDIX (E)

THE MATHEMATICAL FORMULATION OF SMOOTHING TWO PATCHES AT THE COMMON BORDERS

Consider Figure E.1 in which patches P_1 and P_2 have a common border and we wish to smooth the two patches such that first order $C^{(0)}$ and second order $C^{(1)}$ continuity exist across that common boundary. We know (from Chapter 2) that the PC bi-cubic surface patch is defined as

$$v(u,w) = (u^3 u^2 u 1) (M) (B) (M)^T$$

$$\begin{bmatrix} w^3 \\ w^2 \\ w \\ 1 \end{bmatrix}$$

where (M) is a constant matrix defined in the equation and (B) is the boundary matrix

$$(B) = \begin{bmatrix} v_{00} & v_{01} & v_{00W} & v_{01W} \\ v_{10} & v_{11} & v_{10W} & v_{11W} \\ ----- & ---- & ---- \\ v_{00U} & v_{01U} & v_{00UW} & v_{01UW} \\ v_{10U} & v_{11U} & v_{10UW} & v_{11UW} \end{bmatrix}$$

From Figure E.1, to ensure the $C^{(0)}$ continuity between P_1 and P_2 ;

$$\left[\frac{\partial v(u,w)}{\partial u}\right]_{u=1}$$
 of P_1 should be equal to

$$\left[\frac{\partial v(u,w)}{\partial u}\right]_{u=0}$$
 of P_2 .

That is

$$(u = 1, row 2)_{P_1} = (u = 0, row 1)_{P_2}$$
To ensure $C^{(1)}$ continuity between P_1 and P_2 ;
$$[\frac{\partial^2 v(u,w)}{\partial u\partial w}]_{u=1}$$
 of P_1 should be equal to
$$[\frac{\partial^2 v(u,w)}{\partial u\partial w}]_{u=1}$$
 of P_2

so that
$$(u = 1, row 4)_{P_1} = (u = 0, row 3)_{P_2}$$

Therefore, adjacent patches have position and slope continuity if common position rows (or columns) are identical and if common slope rows (or columns) are multiples of each other. Graphically, the elements of interest in the (B) matrices for both patches of Figure `E.1\are as shown in Figure E.2.

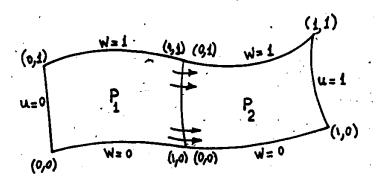


Figure E.1 Geometry for Smoothing Between Two Patches at a Common Border

				b ₂₁	b ₂₂	b ₂₃	b ₂₄
b ₂]	b ₂₂	b ₂₃	b ₂₄			•	
	, .	-		b ₄₁	b ₄₂	. b ₄₃	b ₄₄
b _{4]}	b ₄₂	b ₄₃	b ₄₄				·

Figure E.2 Graphical Representation of the Boundary (B) Matrices Elements of Interest of the Two Patches of Figure E.1

APPENDIX (F)

THE MATHEMATICAL FORMULATION OF BLENDING BETWEEN TWO NON-ADJACENT PATCHES

APPENDIX (F)

THE MATHEMATICAL FORMULATION OF BLENDING BETWEEN TWO NON-ADJACENT PATCHES

Figure F.1 shows patches P_1 and P_2 to be blended by patch P_3 such that it ensures $C^{(0)}$ and $C^{(1)}$ continuity at the respective common borders. The method applied here is the same as that used in Appendix E on smoothing between two patches at a common border, i.e., to ensure $C^{(0)}$ continuity the following relationship between the (B) matrices of P_1 , P_2 and P_3 should be preserved.

$$(u=1, row 2)_{p_1} = (u=0, row 1)_{p_3}$$

$$(u=0, row 1)_{P_2} = (u=1, row 2)_{P_3}$$

and to ensure C(1) continuity

$$(u=1, row 4)_{p_1} = (u=0, row 3)_{p_3}$$
 and

$$(u=0, row 3)_{p_2} = (u=1, row 4)_{p_3}$$

Graphically, the previous relationships are shown in Figure F-2.



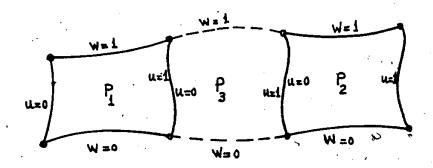


Figure F.1 Geometry for Blending Between Two Non-adjacent Patches

P ₁					P ₃			P ₂			
								^B 11	B ₁₂	^B 13	B ₁₄
b ₂₁	b ₂₂	b ₂₃	b ₂₄	B ₁₁	B ₁₂	,B ₁₃	B ₁₄				
				b ₄₁	ъ ₄₂	ъ ₄₃	b ₄₄	^B 31	B-32	B ₃₃	B ₃₄
b ₄₁	b ₄₂	b ₄₃	b ₄₄	B ₃₁	B ₃₂	B ₃₃	B ₃₄	•			

Figure F.2 Boundary (B) Matrices Elements of Interest for Blending Between Two Non-adjacent Patches

APPENDEX (G)

THE MATHEMATICAL FORMULATION OF INTERSECTING A PATCH WITH A PLANE

APPENDIX (G)

THE MATHEMATICAL FORMULATION OF INTERSECTING A PATCH WITH A PLANE

Consider Figure G.1 in which a patch is represented in the u-w parametric plane. In order to generate cut points to be fitted later by a parametric cubic curve, we must calculate the corresponding u,w values of the cut plane, which is represented by a line on Figure G.1. We have

$$\Delta w = w_2 - w_1$$
 , $w_2 > w_1$

where w_1 and w_2 are obtained by intersecting the patch's boundary with the plane. For any point on the curve of intersection, let

$$DELTAW = \frac{w - w_1}{w_2 - w_1}$$

hence, $w = w_1 + DELTAW \cdot \Delta W$ for $w = w^*$

$$w^* = w_1 + DELTAW . \Delta W$$

and u^* is produced by cutting $w = w^*$ by the plane. Hence, the point on the patch surface $v(u^*,w^*)$ is easily computed since we already know the patch's boundary matrix. The points produced in this fashion are then fitted with a parametric cubic curve to give the actual continuous curve of intersection.

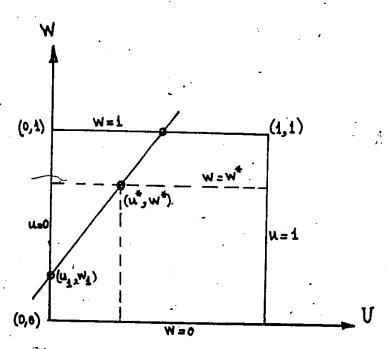


Figure G.1 Representation of a Patch and a Cut Plane in U-W Parametric Plane

APPENDIX (H)

CREATING BOUNDARY MATRIX (B) OF A PATCH GENERATED BY SPLITTING A GIVEN PATCH

APPENDIX (H)

CREATING BOUNDARY MATRIX (B) OF A PATCH GENERATED BY SPLITTING A GIVEN PATCH

Referring to Figure 3.22 of Chapter 3, section 3.3.8, our task is to obtain the boundary (B) matrix of patch P_2 , knowing the boundary (B) matrix of patch P_1 . Before we do that, let us first consider splitting a parametric cubic curve. Asume we wish to define a new curve from u_1 to u_2 on the original curve v(u) as shown on Figure H.1. Using the linear transformation

$$u = u_1 + t (u_2 - u_1)$$

where t is equivalent to u, but for the new curve segment,

$$\frac{du}{dt} = u_2 - u_1$$

then at t=0, $u=u_1$; and at t=1, $u=u_2$

$$\frac{dV}{dt} = \frac{dV}{du} \frac{du}{dt}$$

(H.1)

Using Equation (H.1) and denoting the new curve by C(t), the geometric coefficients of the split curve in terms of the original curve are:

C(0)
$$V(u_1)$$

C(1) $V(u_2)$
C'(0) $(u_2-u_1)V'(u_1)$
C'(1) $(u_2-u_1)V'(u_2)$

Note that C(0) and C(1) represent position data pertaining to end points of the split curve, and C'(0), C'(1) represent a parametric slope data at the corresponding end points.

Now, following the same argument with respec to the PC bi-cubic surface patch splitting, and referring to Figure 3.22 we have

$$[V(0,0)]_{P_2} = [V(u_1,w_1)]_{P_1}$$

and

$$[V(0,0)_{u}]_{P_{2}} = (u_{2}-u_{1})[V(u_{1},w_{1})_{u}]_{P_{1}}$$

The corner cross-derivatives (twists) for P_2 are obtained by evaluating the cross-derivatives of P_1 at the given u, w values, e.g.,

$$[V(1,1)_{uw}]_{P_{2}} = [\frac{\partial^{2}V(u,w)}{\partial u\partial w}]_{P_{1}}$$

$$u=u_{2}$$

$$w=w_{2}$$

$$[V(0,0)_{uw}]_{P_2} = [\frac{a^2V(u,w)}{auaw}]_{P_1}$$
 $u=u_1$
 $w=w_1$

$$[V(0,1)_{uw}]_{P_2} = [\frac{\partial^2 V(u,w)}{\partial u \partial w}]_{P_1}$$

and

$$[V(1,0)_{uw}]_{P_2} = [\frac{\partial^2 V(u,w)}{\partial u \partial w}]_{P_2} = u=u_2$$

 $w=w_1$

Therefore, the boundary matrix (B) of the split patch can be easily obtained.

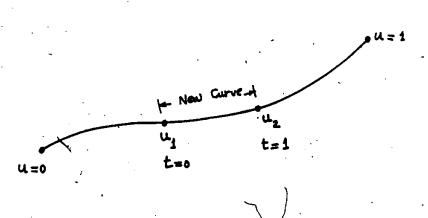


Figure E.1 Geometry for Splitting a Parametric Cubic Curve

APPENDIX (I)

LISTINGS FOR THE GENERAL SUBROUTINES

APPENDIX (I)

GENRAL SUBROUTINES

· 1	-		SUBROUTINE	TEST 1
2	-		SUBROUTINE	CMATRX
3	-		SUBROUTINE	SOLVE
4	-	K	SUBROUTINE	LENGTH
5	-	*	SUBROUTINE	POINTS
6	· -		SUBROUTINE	DRAWP
7	-	•	SUBROUTINE	D3TRNS
8	-	•	SUBROUTINE	MENUH
9	-		SUBROUTINE	MINV
10.	-		SUBROUTINE	MU
11	-		SUBROUTINE	HIDE
12	· <u>-</u>		SUBROUTINE	LOOKUP
13	-		SUBROUTINE	PDATAX
14	-		SUBROUTINE	GEN
15	-		SUBROUTINE	FDREV
16	-		SUBROUTINE	SDREV
17	-	•	SUBROUTINE	DRAWP1
18	-		SUBROUTINE	DRAWP2
19	-		SUBROUTINE	BLEND
20	. -		SUBROUTINE	MODPAT
21	-		SUBROUTINE	INTRSC
22	-,		SUBROUTINE	MODPAT
23	<u>:</u>		SUBROUTINE	BETA

24 - SUBROUTINE BETA1

25 - SUBROUTINE BXBYBZ

26 - SUBROUTINE TOTA

SUBROUTINE BLEND (BX, BBX, BY, BBY, BZ, BBZ, FBX, FBY, FBZ)

FUNCTION - This SUBROUTINE blends two non-adjacent patches.

USAGE - - CALL BLEND (BX, BBX, BY, BBY, BZ, BBZ, FBX, FBY, FBZ).

PARAMETERS BX - Input (B) matrix of patch P1.

for the x-coordinates.

BBX - Input (B) matrix of patch P2 for the x-coordinates.

BY - Input (B) matrix of patch Pl for the y-coordinates.

BBY - Input (B) matrix of patch P2 for the y-coordinates.

BZ - Input (B) matrix of patch P1 for the z-coordinates.

BBZ - Input (B) matrix of patch P2 for the . z-coordinates.

FBX - Output (B) matrix for patch P3 for the x-coordinates.

FBY - Output (B) matrix for patch P3 for the y-coordinates.

FBZ - Output (B) matrix for patch P3 for the z-coordinates.

LANGUAGE :

SUBROUTINE TEST1 (X, Y, Z, BX, BY, BZ)

FUNCTION

- This SUBROUTINE calculates the BX, BY and BZ vectors of a patch. defined by 16 x, y and z triplets.

USAGE

- CALL TEST1 (X, Y, Z, BX, BY, BZ)

PARAMETERS

X - Input vector of length 16 of the data points defining the patch (x-coordinates)

Y - same as x, but (Y-coordinates)

Z - same as X and Y (Z-coordinates)

BX - Output BX vector of length 16

BY - Output BY vector of length 16

BZ - Output BZ vector of length 16

LANGUAGE

SUBROUTINE CMATRX (U, W, C, N)

FUNCTION

- This SUBROUTINE calculates the C matrix of a patch.

USAGE

- CALL CMATRX (U, W, C, N)

PARAMETERS

U - Input vector of U values of the transformation U-W plane (see Fig.2.2) of length N.

W - Input vector of W values of the transformation U-W plane (see Fig.2.2) of length N.

C - Output C matrix dimensioned N by N.

LANGUAGE

SUBROUTINE SOLVE (R, A, M, N, EPS, IER)

FUNCTION

- This SUBROUTINE solves a general system of simultaneous linear equations.

USAGE

- CALL SOLVE (R, AM, M, N, EPS, IER)

PARAMETERS

- R The M by N matrix of right hand sides (destroyed), on return R contains the solution of the equations.
- A The M by M coefficient matrix (destroyed).
- M The number of equation in the system.
- N The number of right hand side vectors.
- EPS An input constant which is used as relative tolerance for test on loss of significance.

METHOD

- Solution is done by means of
Gauss-Elimination with complete
pivoting.

SUBROUTINE LENGTH (L, X, Y, Z)

FUNCTION

- This SUBROUTINE calculates the length between each successive points.

USAGE

- CALL LENGTH (L, X, Y, Z).

PARAMETERS

- L Output vector of the lengths
 between the points. If the
 number of points defining the
 patch is 16, L should be
 dimensioned by L(24) in the
 calling program.
- X Input vector of the X-coordinates of the points defining the patch.
- Y Input vector of the Y-coordinates of the points defining the patch.
- Z Input vector of the Z-coordinates of the points defining the patch.

LANGUAGE

EXAMPLE

If the points defining a patch were entered with the following ordering

13	. 14	15	. 16
		11	1.0
9	10	11	12
	•		
- 5	6	7	8
}	1		
1,	2	. 3	4

the output lengths will have the following ordering.

•	L(10)	L(11)	L(12)	
L(21)	L(22)	L(23)		L(24)
	L(7)	L(8)	L(9)	
L(17)	L(18)	L(19)	•	L(20)
	L(4)	·L(5)	L(6)	
L(13)	L(14)	L(15)		L(16)
1	L(1)	L(2)	L(3)	

SUBROUTINE POINTS (U, W, B, X, N)

FUNCTION

This SUBROUTINE calculates the coordinate (X or Y or Z) of N points lying on the surface of a patch; defined by the corresponding B vector (BX or BY or BZ); at a corresponding U and W values.

USAGE

- CALL POINTS (U, W, BY, Y, 16)

PARAMETERS

U - vector of V values of length N

W - vector of W values of length N

B - BX or BY or BZ vector defining the patch.

X - output vector containing the

X (or Y or Z depending on the call)

coordinate values of the points

lying on the surface.

LANGUAGE

FORTRAN

SUBROUTINE DRAWP (BX, BY, BZ, IFLAG)

FUNCTION

This SUBROUTINE draws the patch on the CRT.

USAGE

- CALL DRAWP (BX, BY, BZ, IFLAG).

PARAMETERS

BX - Input boundary vector of a patch defined by 16 points.

BY - Input boundary vector of a patch defined by 16 points.

BZ - Input boundary vector of a patch defined by 16 points.

IFLAG = 0 only DRAWP draws the patch.

IFLAG = 1 DRAWP calls subroutine D3TRNS
to perform translation of the
patch to another location on
the CRT.

LANGUAGE

FORTRAN

SUBROUTINE D3TRNS (N, X, Y, Z, L, M, NN)

FUNCTION

- This SUBROUTINE performs a 3-D translation.

USAGE

- CALL D3TRNS (N, X, Y, Z, L, M, NN)

PARAMETERS

N - Number of X, Y, Z triplets.

X - Array containing X-coordinates, dimensioned with N.

Y - Array containing Y-coordinates dimensioned with N.

Z - Array containing Z-coordinates dimensioned with N.

L - X-translation factor (real)

M - Y-translation factor (real)

NN - Z-translation factor (real)

LANGUAGE

SUBROUTINE MENUH (IT, M1, M2)

FUNCTION

This SUBROUTINE delays execution of the program till a light pen hit is detected on the menu area, and then returns the order of the hit subpicture.

PARAMETERS

if there are three options on
the menu area and the user
points at the second option,
MENUH will return IT = 2.

M1 - Tag of the first subpicture.

M2 - Tag of the last subpicture.

USAGE

- CALL MENUH (IT, M1, M2)

LANGUAGE

SUBROUTINE DRAWP1 (BX, BY, BZ, X, Y, Z)

FUNCTION

Same as subroutine DRAWP, the only difference is that DRAWP1 puts the points of the patch on a set of subpictures beginning with a tag of 1000 and ending with a tag of 1032, and the lines joining these points on another set of subpictures beginning with a tag of 102 and ending with a tag of 1034. So, the user can OFF any of these subpictures from his calling program.

SUBROUTINE MODPAT (XOX, YOY, ZOZ, X1, Y1, Z1)

FUNCTION

patch, by means of a tracking object which can be used to drag one of the original points lying on the patch surface to another position.

- CALL MODPAT (XOX, YOY, ZOZ, X1, Y1, Z1).

PARAMETERS

X1 - Array of the original X coordinates.

Y1 - Array of the original Y coordinates.

Z1 - Array of the original Z coordinates.

XOX - Array of the modified X coordinates.

ZOZ - Array of the modified Z coordinates.

LANGUAGE

SUBROUTINE INTRSC (BX, BY, BZ)

FUNCTION

This SUBROUTINE draws the resultant continuous curve from the intersection of a plane and a patch defined by BX, BY, BZ vectors, on the CRT.

USAGE

CALL INTRSC (BX, BY, BZ).

PARAMETERS

BX - I/P BX vector of the patch.

BY - Input BY vector of the patch.

, BZ - Input BZ vector of the patch.

LANGUAGE

- FORTRAN

EXAMPLE

If a call is made to subroutine INTRSC, a representation of the patch on the U-W plane will be drawn on the CRT (See Figure I.1), also a tracking object will appear on the center of the CRT.

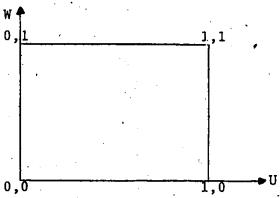


Figure (I.1) Representation of the patch on the U-W plane

The user is then asked to define the plane he wants to intersect the patch, since we are working on the U-W plane, the plane will be represented by a line. To define that line the user is asked to position the tracking object at two points on the borders of the patch (see Figure 1.2).

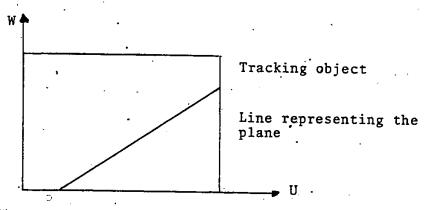


Figure (I.2) Patch and the interacting plane

The subroutine, then, draws automatically the resulting continuous curve.

This way the user is not asked to define his plane mathematically, only graphically.

SUBROUTINE BXBYBZ (BXI, BYI, BZI, BX, BY, BZ)

FUNCTION

This SUBROUTINE calculates the BX, BY, BZ vectors from the B matrices BXI, BYI, BZI of a patch. (refer to equation 13, S would be either one of BX. BY or BZ and B would be either one of BXI, BYI or BZI, e.g.

BX = [M][BXI][M]

definition of [M] matrix is given in equation (5).

USAGE

CALL BXBYBZ (BXI, BYI, BZI, BX, BY, BZ)

PARAMETERS

BXI- Input B matrix of the X coordinates.

BYI- Input B matrix of the Y coordinates.

BZI- Input B matrix of the Z coordinates.

BX - Output BX vector.

BY - Output BY vector.

BZ - Output BZ vector.

LANGUAGE

SUBROUTINE BETA1 (BX2, BX1, BY2, BY1, BZ2, BZ1, BX14, BY14, BZ14, J, K, L, M, R1, R2, R3, R4)

FUNCTION

This SUBROUTINE blends two
patches in the Y direction (refer
to the theory of blending patches
in this report).

USAGE

- CALL BETA1 (BX2, BX1, BY2, BY1, BZ2, BZ1, BX14, BY14, J, K, L, M, R1, R2, R3, R4).

PARAMETERS

- BX2 Output 'B' matrix of the blended patch of the X coordinates.
- BX1 Input 'B' matrix of the original patch, of the X coordinates, to the right of the blended patch.
- BX2 Output B matrix of the blended patch of the Y coordinates.
- BY1 Input 'B' matrix of the original patch, of the Y coordinates, to the right of the blended patch.
- BZ2 Output 'B' matrix of the blended patch of the Z coordinates.
- BZ1 Input 'B' matrix of the original patch, of the Z coordinates, to the right of the blended patch.

- BX14 Input 'B' matrix of the original patch, of the X coordinates, to the left of the blended patch.
- BY14 Input 'B' matrix of the original patch, of the Y coordinates, to the left of the blended patch.
- BX14 Input 'B' matrix of the original patch, of the Z coordinates, to the left of the blended patch.

J - set equal to 1.

K - set equal to 2.

L - set equal to 3.

M - set equal to 4.

R1 - a ratio value; input by the user equal to $\frac{\text{W1}}{\text{W2-W1}}$ (See Figure) or equal to 1.

R2 - a ratio value, input by the user equal to (1-W2)/(W2-W1) or equal to 1.

R3- - V1/(V2-V1) or 1.

R4- - (1-V2)/(V2-V1) or 1.

SUBROUTINE BETA (BXB, BX1B, BYB, BY1B, BZB, BZ1B, BXXB, BYYB, BZZB, R1, R2, R3, R4)

FUNCTION

same as SUBROUTINE BETA 1,
 except BETA blends two patches
 in the X direction.

USAGE

- CALL BETA (BXB, BX1B, BYB, BY1B, BZB, BZ1B, BXXB, BYYB, BZZB, R1, R2, R3, R4).

PARAMETERS

- same as BETA 1.

LANGUAGE

SUBROUTINE TOTA (BBM, BBMT)

FUNCTION

- This subroutine calculates the inverse of the matrix M and \mathbf{M}^{T} .

USAGE

- CALL TOTA (BBM, BBMT).

PARAMETERS

BBM - output matrix = M^{-1} .

BBMT - output matrix = $(M^T)^{-1}$.

LANGUAGE

- FORTRAN

SUBROUTINE MU (A, B, R, N, M, L)

FUNCTION

This SUBROUTINE multiplies two

general matrices.

USAGE

CALL MU (A, B, R, N, M, L)

PARAMETERS

A - Name of the first input matrix.

B - Name of the second input matrix.

R - Name of the resultant matrix.

N - Number of rows in A and columns of B.

M - Number of rows in B.

L - Number of rows in R.

LANGUAGE

SUBROUTINE MINV (A, N, D, L, M)

FUNCTION

- This SUBROUTINE inverts a general non-singular matrix.

USAGE

- CALL MINV (A, N, D, L, M).

PARAMETERS

A - Name of input matrix (destroyed)
when return A is the inverted
matrix.

N - Number of rows of A.

D - = 0 inversion correct

= 1 inversion incorrect

L - Working array dimensioned with N.

M - Working array dimensioned with N.

LANGUAGE

SUBROUTINE GEN (BX, BY, BZ, BGX, BGY, BGZ)

FUNCTION

This SUBROUTINE constructs
the 'B' matrix of a patch
generated by subdivision of
another given patch.

3)

USAGE

- CALL GEN (BX, BY, BZ, BGX, BGY, BGZ).

PARAMETERS

BX - Input BX vector of the given patch dimensioned with 16.

BY - Input BY vector of the given patch dimensioned with 16.

BZ - Input BZ vector of the given patch dimensioned with 16.

BGX - Output BX vector of the generated patch.

BGY - Output BY vector of the generated patch.

BGZ - Output BZ vector of the generated patch.

LANGUAGE

- FORTRAN

SUBROUTINE SDREV-(U, W, B, D)

FUNCTION

- This SUBROUTINE calculates the second drevative of V(U,W).

USAGE

CALL SDREV (U, W, B, D).

PARAMETERS

- U Input U value to define a certain point on the patch at which SDREV calculates the second drevative.
- W Input W value to define a certain point on the patch at which SDREV calculates the second drevative.
- B The 'B' matrix of the patch.
- D The output value of the drevative.

LANGUAGE

- FORTRAN

SUBROUTINE FDREV (U, W, B, D, KOK)

FUNCTION

USAGE

PARAMETERS.

This SUBROUTINE calculates
the first drevatives of
V(U, W) (refer to equation 10).

- CALL FDREV (U, W, B, D, KOK).
- U Input U value to define a certain point on the patch.
- W Input W value to define a certain point on the patch.
- B The input 'B' matrix of the patch.
- D The output value of the dreative.

KOK - A flag if = 1, D =
$$\frac{\partial V(U,W)}{\partial V}$$

if = 2, D = $\frac{\partial V(U,W)}{\partial W}$

LANGUAGE

FORTRAN -

SUBROUTINE POINT (U, W, B, BG)

FUNCTION

This SUBROUTINE calculates
the X or Y or Z coordinate
of a point lying on a given
patch at a corresponding
U and W values.

USAGE

- CALL POINT (U, W, B, BG).

PARAMETERS

U - Input U value.

W - Input W value.

B - Input 'B' vector defining the patch. (BX or BY or BZ).

BG - Output coordinates,

if B = BX, BG = X

, B = BY, BG = Y

, B = BZ, BG = Z

coordinate.

LANGUAGE

```
20%
FORTHAN IV
                  v. 2 1-1
                                                                      PAGE 221
             SUBROUTINE GENUAL (ax, ax, az, x, y, z)
1233
             DIMENSION PXC1);BYC1);BZC1);xC1);YC1);ZC1);
...3
             DIMENSION U(2) 44(2) 466(16) 444(16)
2234
             COMMON/OFILE/IBUF(1)
Ler5
             COMMUNISUPERIGRO, IPS
             SWERGUTINE GENUCL
       C
             FUNCTION:
      С
             GET, XIY Z COORD. OF THE PATCH GENERATED BY THE
           "SUBBIVE ION OF THE GIVIN PATCH
      C
             CALL IT XJYJZ
      C
             kRITE(5:3)
2227
      . 3
             FORMAT(14,32HENTER U(1), U(2), W(1), W(2) VALUES)
2223
             READ(5,*)U(1),U(2),W(1),k(2)
2239
            · LELU=(U(2)-U(1))/3.
2312
             DELW=(W(2)-W(1))/3.
2211 -
             UU(1)=U(1)
2212
             UU(2) = U(1) + DELU
             UU(3)=U(1)+2.*JELL
0213
2214
             UU(4)=U(1)+3.*DELU
             DC 1 I=1,4...
££15
U215
             WW(I)=W(J)
2217
             % % (3+4) ~ % (1)+DELW
             NH(1+8)=N(1)+2.*BELW
2213
2019
             #W(I+12)=W(1)+3.*DEL##
022U ·
             30 12 I=₹33
2821
             UU(1+4×I)=UU(1)
2222
             UUC2+4#I)=UUC2)
2223
             663+4×I)=88(3)
2224
       13
             55(4+4×1)=00(4)
2225
             CALL POINTS(UU)NW,9x,y,16)
Ø225
             CALL POINTS (UU) WW.BY, Y, 16).
2227
             CALL POINTS (UU, WW. BZ, Z, 16)
2228 -
             RETURN
2229
             END
```

```
21:
PAGE 221
```

```
FORTHAN IV
                  v22.1-1
              SUBRICUTINE INTACTIONAL STANDATCH ON PLOTELD
 6222
              GIMENSION X(1),Y(1),Z(1),XXX(32),XQ(2),YQ(2),ZQ(2)
 2023
              CCMMCN/DF.ILE/IBUF(1)
 2024
              COMMON/RECALL/IRECL
 0235
             COMMON/SUPER/GROJIP, IPO
       C
                SUBROUTINE INTACT
       С
             FUNCTION:
       C
             THIS SUBROUTINE RETURNES THE X.Y.Z COORDINATES OF A PATCH
       C
             TO THE MAIN PROGRAM USING INPUT FROM LIGHT PEN.
       С
               NPATCH=NUMBER OF PATCHES
                                           (I/P)
       С
             NPL=NUMBER OF PATCHES PER PLANE (I/P)
       С
             IFL=1 BEFINITION OF CURVED SURFACES
               IFL=2 DEFINITION OF SURFACES OF PEVOLUTIONS
                                                               (I/P)
0236
             WRITE(5)2)
2537
             FORMAT( PLEASE ENTER THE NUMBER OF PATCHES!)
25.33
             READ(5,3)NPATCH
22291
             hRITE(5,551)
2213
             FORMATC! PLEASE ENTER NUMBER OF PATCHES PER PLANE!)
JE 11
            · READ(5,3)NPL
3312
             FORMAT(12)
2613
             K = 3
       C
       С
             PLOT FRAME OF WORKING PLANES
       C
2214
             x.x =5 12 •
8215
             CALL SUBP(IP).
2215
             CALL APNT(XX3XX33-47)
2217
             CALL VECT(XX30.0) .
3218
             CALL APNT(970.3460.33-4)
2219
             CALL TEXT( YI)
             CALL ESUE
2023
2221
             CALL SUBP(IP+1)
2222
             CALL APNT(XX3XX33-4)
2223
             CALL VECT(2.2.XX)
2E 34
             CALL APNT(505.,1200.,,-4)
2225
           CALL TEXT(1 ZI)
66.39
           · CALL ESUE
UE 27
             CALL SUSP(IP+2)
2228
             CALL APNT(XX3XX33-4)
2229
            CALL VECT (-XX3-XX)
2232
             CALL APNT (52.35.)
2231 °
            CALL TEXT(1 X1)
2032
            CALL ESUB
2233
            II = 2
2234
            KK=2°.
            KKK=2
2235
            CALL SUBP (IP+4)
2237
            CALL GFF (IP+4)
0038
            GALL MENU(2.,502.,-102.,2923+1P, POSITION,, CONE)
2239
            CALL ESUE
            CALL SUBP(IP+3)
3643
```

```
211
   FORTHAN IV
                   v12.1-1
                                                                      PAGE 232
   0.41
               CALL (FF(IP+3)
  6.42
               CALL MENU(2.8,620.,-180.,2030+IP, DEFINE WORKING PLANE!,
              * IDUNEI)
  2243
               CALL ESUE
               DO 4 J=1.NPATCH/NPL
  c. 44
  6245
               WRITE(5 5)J
              FORMAT( DEFINE PATCHES IN PLANE NUMBER 1,12)
  3245
  2847
               wRITE(5)6)
  2243
              FORMAT(!----
         - 6
  EE 49
             IF (KKK.EG.5.OR.KKK.EC.9.OR.KKK.EC.17.CR.KKK.EC.17)KKK=KKK-1
              IF (KKK.EQ.21.0R.KKK.EQ.25.0R.KKK.EQ.29.0R.KKK.EQ.33)KKK=KKK-1
  ยีย51
  2253
              IF (KKK. EQ. 37. OR. KKK. EC. 41. OR. KKK. EQ. 45. CR. KKK. EQ. 49)KKK =KKK-1
  .255
              IF(KK+EC+4+0R+KK+EC+8+0R+KK+EC+12+0P+KK+EC+16)G8 TO 721
  2057
              IF(KK.E0.20.CR.KK.E0.24)50 T0 721
  2259
              GD TO 732
  5253
         7-1 DC 16 MEME=1 .KK
  6251
         16
              CALL OFF(IP+4+MEME)
  2352
         722 CALL ON(IP+3)
  2233
              CALL MENUHCIT, 2222+IP, 2001+IP)
  2254
              CALL OFF(IP+3)
  2235
              GC TG (22,132),IT
  2265
              CALL TRAK(XX,XX)
 26.57
              IF(II.E0.4)II=G
 28691
              KK = KK + 1
 3273
              II = II + 1
 2271
              WRITE(5,11)II
 2272
              FORMAT(1x, POSITION TRAK. 09J. TO DEFINE SEC. NUMBER (1)11)
 2273
              IF(IFL.E0.2)WRITE(5.12)
              FORMATCE YOU SHOULD DEFINE 6 POINTS IN THIS SECTION THE FIRST
 2275
         12
 2:75
              IF(IFL.EC.2) LRITE(5,14)
 2278
              FORMATO! TWO WILL DEFINE THE AXIS OF REVOLUTION!)
 2279
              READ(5,21)M
 2282
         21
              FORMAT(A2)
 2231
              CALL LPEN(HHITT1)
 2682
              IF(Ih.EC.@.OR.IT1.LT.121.OR.IT1.GT.123)GO TO 30
 2234
              CALL GRID(GRD, GRD)
 8835
             CALL TRAKXY(XØ,YØ)
 5235
             IT1=IT1-122
 8287
             GO TO (182,228,322),IT1
 8233
        102 CALL APRICKETYES
 2239
             CALL SUSP(IP+4+KK)
- 2293
             CALL OFF(IP+4+KK)
. 2291
             CALL VECT(-XX3-XX33II)
 2292
             CALL VECT(0.2.xx,,II)
 2293
             CALL VERTOXXEXXEEID
2094
             CALL VE T(0.23-XX, II)
2095
             CALL ES B
2296
             60 TO 422
2297
        202 CALL APRICKS, Y2)
2293
             CALL SUSP(IP+4+KK)
6899
             CALL OFF(IP+4+KK)
2132
             CALL VECT(-XX3-XX33II)
```

```
FURTRAN IV
                   V & 2 • 1 ~ 1
                                                                           PAGE 1223
   5131
                CALL VECT(XX,2.0,,II)
   2122
                CALL VECT(XX xX x x II)
   2133
                CALL VECT(-Xx,3,2,2,,11)
  12124
                CALL ESUE
   2125
                GO TO 430
  2135
                CALL APNT(XZ)YZ)
   2127
                CALL SUEF(IP+4+KK)
   ∟13ಕ
                CALL OFF(IP+4+KK)
  3129
                CALL VECT(XX,2,2,11)
  2113
                CALL VECT(8.2,xx,,II)
  0111
                CALL VECTC-XX,3,2,,11)
  1112
                CALL VECT(Z.Z.-XX.,II)
  2113
                CALL ESUB
  1114
                CALL CN(IP+4+KK)
 . .115
                I=1+K*NPL
  2113
                IF(II.E0.2) I=1+4*NPL+K*NPL
 ذ11 ن.
                IF(II.EC.3) I=1+8*NPL+K*NPL
  2122
                IF(II.EG.4)I=1+12*NPL+K*NPL
  2122
          1250 KKK=KKK+1
               CALL ON (IP+4)
  2123
               CALL ME UH (IT2,2932+IP,2931+IP) '
  2124
  2125
               CALL OF (IP+4)
  2126
               GO .TO (522,722),IT2
  2127
               CALL GRID(GRD, GRD)
  3128
               CALL TRAKXY(X(I);Y(I))
CALL SUBP(IPG+KKK+IRECL*16)
  2129-
  2132
               CALL OFF(IPO+KKK+IRECL *16)
  3131
               CALL APNT(X(I),Y(I),1,4)
  6132
               CALL ES B
  @133
               CALL ON (IPO+KKK+IRECL*16)
 2134
               GG TO (822,928,1020),IT1
         621
 2135
               \angle(I) = Y(I) - Y.C + X.Z - X(I)
 2136
               x(I)=(X2-x(I))*SQRT(2.)
 2137
               Y(I) = x3 512
 2133
               GO TO 1120
 €1.39
         928
               XXX(I) = I(I)
 2142
               x(I)=(YZ-Y(I))*SQRT(2.) /
 2141
               A(I) = XXX(I) - X9 + A0 - A(I)
 2142
               ¿(I) = YØ -512.
 1143
               GC TO 1102
 2144
         1002 Z(I)=Y(I)-Y0
 2145
               \mathbb{C}X - (I)X = (I)Y
 2146.
               X(I)=(512.-Xd)*SQRT(2.)
 2147
         1122 I=I+1
 2148
               60 TO 1222 -
 2149
              DG 1 IOI=1.KK
         133
 3152
             CALL ERAS(IP+4+101)
         1
 2151
               CALL CMPRS
 2152
               II=C
 2153
               K=16*J
 2154
               CONTINUE
. 2155
              RETURN
 2155
```

```
2621
             SUBROUTINE ROTATE(N) X , Y , Z , TN1 , TN2 , TN3 , T1)
1.12
             DIMENSI N U.C4,4); V(4,4); V(1); Y(1); Z(1); T(4,4)
       C
             SUBROUTINE ROTATE
       C
             FL(NCTION:
       C
             THREE DIMENSIONAL ROTATION & ANY APPLITARY AXIS
       C
             TN1, TN2, TN3 ARE THE DIRECTION COSINES OF THE AXIS
       C
             SF ROTATION.
                              (I/P)
      ·C
2233
             00 1 J= 34
2204
             00 1 I=1.N
2035
             U(I,J)=2.0
2226
             ړ ته =(لوI)۷
        1
2227
             3C 2 I=1 N
2633
             U(I+1)=\chi(I)
                                    PIN=NUMBER OF XJYJZ TRIPLETS
             U(I,2)=Y(I)
2010
             U(I \setminus I) = I(I)
4(11)
             v(I₃4)=1.3
5512
             DC 3 I= 34
2213
             30 3 J=1,4
0214
             T(I) J) = Z.2
                                     IT IS THE TRANSFORMATION MATRIX
2215
             T2=T1/5 .2957795
                                     IT1 IS ROT. ANGEL IN DEGREES.
8216
             T(4,4)=1.0
3217
             T(1_21) = N1*TN1+(1-TN1*TN1)*COS(T2)
0218
             T(1,2) = (N1*TN2*(1-COS(T2))+TN3*SIN(T2)
9219
             T(1,3)=TN1*TN3*(1-COS(T2))-TN2*SIN(T2)
65.23
             T(2)1) = N1*TN2*(1-COS(T2))-TN3*SIN(T2)
2221
             T(2,2)=TN2*TN2+(1-TN2*TN2)*COS(T2)
2422
             T(2,3)=TN2*TN3*(1-COS(T2))+TN1*SIN(T2)
8223
             T(3,1)=TN1*TN3*(1-COS(T2))+TK2*SIN(T2)
2E 24
             T(3,2)=TN2*TN3*(1-COS(T2))-TN1*SIN(T2)
₫£25
             T(3)3)=TN3+TN3+(1-TN3+TN3)*CGS(T2)
2559
             CALL MU(U)T)V)49494).
2227
             30 4 I= N
22.23
             X(I) = Y(I-1)
21.29
             Y(I)=V(1:2)
2232
             2(I)=V(I,3)
2231
             CONTINUE
Je 32
             RETURN
3233
             END
```

```
FASE 221
FORTRAN IV
                 VJ2.1-1
             SUBROUTINE SMOOTH(BXX) BYY) BZZ) NUMJIFTJBX 2, BY 2, BZZ) IFCL)
2221
             DIMENSION BX1(16);BY1(14);BZ1(16);EY2(16);BY2(16);
22 22
               BZ2(16), BXX(1), EYY(1), BZZ(1), IET(2), FBX1(4,4)
            * - FBY1(4,4),FBZ1(4,4),FBx2(4,4),FBY2(4,4),F&Z2(4,4)
2223
             COMMON/DFILE/IBUF(1)
             COMMON/ERS/IERAS
8.84
      C
            SMOOTH.SUB
      С
             PURPOSE:
      С
             THIS SURROUTINE SMOOTHES TWO ADJACENT PATCHES ALONG THE
      C
             COMMON BORDERS.
             ARGUMENTS:
             BXX=ARRAY CONTAINING ELEMENTS OF BX VECTOR BEFORE SMOOTHING
      Ç
      C
             EYY= oo
                                      9.5
                                              ∍ EY
                                                     ,
                                                             ,,
                                                                     . .
                           ,,
      C
             32Z= JJ
                                                 3 Z
             NUM=NUMBER OF DEFINED PATCHES.
      C
             IET = ARR Y CONTAINING THE TAGS OF THE TWO PATCHES TO BE SMOOT
      C
             EXZ=ARRAY CONTAINING ELEMENTS OF BX VECTOR AFTER SMOOTHING
      C
      C
                                               , 3Y
             5Y2= ,,
                            ,,
                                       ,,
                                                        ,,
       C
             222= 33
                                               , 3Z
                                       ,,
                                                        ,,
                            ,,
                          SMOOTHING ALONG LUNGITUDINAL BORDERS
       C
             IFOL=1
                                                                         (0/P)
                                       TRANSVERSE BORDERS
                                                                   (G/P)
       C
              . =2
                                 ,,
                       ,,
       C
             50 1 I= J2
66.35
             CALL IDNTFY(NUM; ITO)
2236
2237
             IET(I)=ITG
             DC 12 I=1,16
2228
             BX1(I) = 9xX(I + (IET(1) - 1) \times 16)
2639
             6Y1(I)=3YY(I+(IET(1)-1)*16)
6210
2211.
        16
             EZ1(I)=FZZ(I+(IET(1)-1)*16)
2212
             30 22 I=1,16
             EX2(I) = \lambda X(I + (IET(2) - 1) \times 16)
JE 13
             BY2(1)=BYY(I+(IET(2)-1)*16)
2214
0215
        22
             EZ2(I) = FZZ(I + (IEI(2) - 1) \times 16)
2316
             CALL GETE (BX1,BY1,BZ1,FBX1,F3Y1,FBZ1)
2217
             CALL GE B(BX2,BY2,BZ2,FBX2,FBY2,FBZ2)
0218
             WRITE(5)688)
             FORMAT(1x) IF YOU WANT TO SMOOTH ALONG LONG. BORDER TYPE 11:
2 6 1 9
2922
             WRITE(5 700)
             FORMAT(1x, ALONG TRANS - BORDER TYPE 21)
2221
        702
8622
             READ(5, 02)IFOL
2023
             FORMAT(12)
              IF(IFOL.Ed.1)G0 TO 922
22 24
2226
             00 5# I=1,4
2227
              F6X2(I)1)=F8x1(I)2)
              FBY2(I,1)=FBY1(I,2)
2223
              FBZ2(I,1)=FBZ1(I,2)
2229
       -50
```

0233

0031

2532 2533

2234

60

CO 50 I=1,4

GO TO 1 20

F8X2(I33)=F8X1(I34)
F8Y2(I33)=F8Y1(I34)

F5Z2(I)3)=F8Z1(I)4)

```
2235
        94.6
              CG 55 I 1,4
1135
              FEX2(1, )=F8x1(2,1)
              FEY2(1,1)=FBY1(2,1)
2235
        55
              FEZ2(1,1)=FBZ1(2,1)
2139
              00 65 1 1,4
              FEX2(3,1)=FBX1(4,1)
2248
2341
2342
              FEY2(3,1)=FBY1(4,1)
        65 F5Z2(3,1)=F8Z1(4,1)
1220 CALL GETBX(F8X2,F5Y2,F8Z2,8X2,8Y2,8Z2)
2243
2244
              RETURN (
2245
              еир
```

```
PAGE 201
```

```
SUBRICUTINE BLEND (EXPEBX) BY PRBY PRIPRE BX PRE YUFRZ)
CL21
            TIMENSION 8X(4,4),BBX(4,4),BY(4,4),28Y(4,4),5Z(4,4)
2222
             DIMENSI N BBZ(4,4),FBX(4,4),FEY(4,4),FBZ(4,4)
2223
             SUBROUTINE BLEND
      C
             FUNCTION:
      C
             THIS SUBROUTINE BLENDS TWO NON ADJACENT BATCHES
             INPUT VARIABLES:
             BXJBYJBZ ARE ELEMENTS OF THE BOUNDARY MATRICES OF THE
      C
             FIRST PATCH TO BE BLENDED WITH THE SECOND PATCH.
             BEXJEBY, BBZ ARE ELEMENTS OF THE BOUNDARY MATRICES OF THE
      C
             SECOND PATCH.
      C
             GUTPUT VARIABLES:
             FEX. FBY. FEZ ARE ELEMENTS OF THE INDETWEEN (GENRATED) PATCH
      C
             WRITE(5,38)
0024
             FORMAT(1x, ) IF YOU WANT BLENDING ALONG LONG DIRECTION ENTER
2:35
        34
               2 JRANS DIRECTION ENTER 11)
             READ(5)42) IB
3226
             FORMAT(I2)
2227
        42
             IF(IE.EC.1)GO TO 12
2233
             DO 1 1=1.4
2212
             FEX(1,I)=BX(2,I)
211 كان
             FBY(1,I)=BY(2,I)
2212
        1
             FEZ(1,I)=2Z(2,I)
2213
       C
             00 2 I=1,4
2214
             FBX(2,I)=BBX(1,I)
2215
             FBY(2,1 =BBY(1,1)
2216
        2
             F2Z(2,1)=8BZ(1,1)
2217
       C
             CO 3 I= 34
2213
             FEX(3) I =5X(4) I)
3019
              FEY(3,I)=BY(4,I)
0220
              FEZ(3,1)=3Z(4,1)
        3.
ZE 21
       Ċ
              50 4 I=1,4
 22 22
              FEX(4)I)=BBX(3)I)
2223
              FEY(4, I) = BBY(3, I)
 2224
              F92(4,1)=982(3,1)
 0225
       C
              GD TO 63
 2226
              DO 11 I 194
 2627
              FBx(I,1)=BX(I,2)
 BB 23
              FEY(I,1)=5Y(I,2)
 2229
              FBZ(I,1)=8Z(I,2)
 2233
        11
       C
              BC 21 I =1.4
 ë€ 31
              FBX(I,2)=BBX(I,1)
 2232
              FSY(I,2)=93Y(I,1)
 0033
              FEZ(I,2)=BBZ(I,1)
 2234
         21
        C
              DO 31 I=1.4
 2535
```

```
v62.1-1
FORTHAN IV
              F5((1,3)=EX(1,4)
F5Y(1,5)=5Y(1,4)
2236
2237
      3 1
C
2233
               FEZ(1,3)=5Z(1,4)
3139
               20 41 I=1,4
               FEXCIJAD=EBKCIJ3)
2242
               F8Y(1,4)=88Y(1,3)
2241
               F3Z(1,4)=EBZ(1,3)
2242
        41
       C
2243
2244
         62
               RETURN
               END
```

5

```
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```

```
2531
             SUBREUTINE CHSAVE (X,Y,Z)
2232
             DIMENSION X(1),Y(1),Z(1)
2223
             COMMON/DFILE/IBUF(1)
2234
             CCMMCN/SUPER/GRD
2225
             LOGICAL 1 FILE1(15), FILE2(15)
            SAVE THE DISPLAY
2225
            CALL INFILE(FILE1, FILE2, NPATCH)
2007
             CALL STOP
1233
             CALL ASSIGN(12,FILE2,2)
6229
             DO 1 I=1 NPATCH#16
1113
       1
             KRITE(123*)X(I)3Y(I)3Z(I)
                                             INRITE XYYZ CUCRD. IN DATA
1211
             CALL CLOSE(12)
                                             !FILE (FILE2.DAT)
2212
             CALL SAVE(FILE1)
2213
            CALL CONT
2214
            CALL LPEN(IHJIT)
JC15
            RETURN
2216
            END
```

```
3221
            SUBROUTINE GETSHP(X)Y)Z)
            DIMENSION X(1) (1) Z(1)
202
2223
            CCMMON/OFILE/IBUF(1)
            COMMON/RECALL/IRECL
2224
            LCGICAL*1 FILE1(15) FILE2(15)
3235
      C
            SUBROUTINE GETSHP
      Č
            FUNCTIO :
      C
            RECALL A PREVIOUSLY SAVED CISPLAY.
             CALL INFILE(FILE1, FILE2, NPATCH)
6685
             CALL STOP
2227
             CALL ASSIGN(13,FILE2,0)
ยียยี3
             DG 1 I=1, NPATCH*16
2229
             READ(12)*(I))Y(I)JZ(I)
2212
2.11
             CALL CL SE(10)
2212
             CALL INIT
             CALL RSTR(FILE1)
2013
2214
             CALL CONT
             CALL LPEN(IH;IT)
2215
             IRECL=IRECL+NPATCH
1216
2217
             DO 12 I=1,13
       12
             CALL ERAS(102+I)
2218
2219
             CALL CMPRS
2220
             RETURN
```

END

2221

```
FORTHAN IV
                  V42.1-1
                                                                       PAGE #01
0231
             SUBRICUTINE INTUOL (BY JBY JBZ JY JY JZ)
1732
             DIMENSION U(16), N(16), ((1), Y(1), Z(1)
                JBX(1), BY(1), BZ(1), DELX(15), DELY(15)
             COMMON/OFILE/IBUF(1)
2624
             COMMON/SUPER/GROJIP, IPO
J0 J5
             CGMMON/SEC/ISEC
      C
      С
             JCLTNI BNITUORBUE
      ¢
             FUNCTION:
      C
             AHIS SUBROUTINE DRAWS THE RESULTANT CONTINUOUS CURVE
      C
             FROM THE INTERSECTION OF A PLANE AND A PATCH
      C
      ¢
             INPUT VARIABLES:
             EXJEYJEZ ARE XJYJZ BOUNDARY VECTORS DEFINING THE PATCH
      C
      C
             CUTPUT VARIABLES:
      C
             X,Y,Z C ORDINATES OF THE CURVE OF INTERSECTION
             C = 1224 \bullet
2006
2627
            'CALL AP T(C/8 • 2) • × C/8 • • -4)
             CALL SUBP (61 42+1 SEC)
2238
2239
             CALL OFF(6142+ISEC)
3212
             CALL LVECT( . 25 * C . 0 . )
១៩11
             EALL LVECT(24, 25 *C)
2212
             CALL'
                  /LVECT(-.25*C,2.)
             CALL AVECT(0.,-.25*C)
0213
3614
             CALL APNT(+1*C+4+5*C/2+++-4)
2215
             CALL TEXT(10,01)
0216
             CALL APNT(+3×C+4+5×C/2+++-4)
2217
             CALL TEXT( 11201)
2213
             CALL APRIC+3*C++9*C++-4)
2219
             CALL TEXT('1,1')
8 L 20
             CALL APNT( 1 × C 1 + C 1 + C 1 - 4)
2221
             CALL TEXT( 13,11)
2022
             CALL ESUB
6823
             CALL ON 6142+ISEC)
      C
             POSITION TRACKING OBJECT ON THE POINTS OF INTRSECTION
22.24
             CALL TRAK(512.,512.)
2225
             KEMO=0
2226
       10
             hRITE(5,11)
2227
        11
             FORMAT(1X) POSITION TRAC+ OBJ++TYPE(CR> WHEN CONE+)
2223
             READ(5,20)I
WE 29
        24
             FCRMAT(A2)
2232
             IF(KEMO.GT.2)GO TO 30
2232
             CALL TRAKXY(XC,YC)
SE 33
             KEMO=KEMC+1
£634
             GC TO 14
2035
       32
             CALL TRAKXY(X01,Y01)
             CALCULATE(U1) ¥1) (U16) ¥16)
&£35
             CALL SUSP(6141+ISEC)
2237
             CALL APRICKO, YO)
```

```
36
37
39
             CALL LVECT((XC1-XC),(Y11-Y0))
             CALL ES &
2243
             U(1) = (X0 - (0/8*))/(*25*0)
             *(1)=(Y ~(5**C/8*))/(*25*C)
3241
             U(16)=(X01-(C/8.))/(.25xC)
2242
             w(16) = (YC1-(5.*C/8.))/(.25*C)
2643
6244
             DELTAU=(U(16)-U(1))/15.
2245
             SELTAK=(k(14)-4(1))/15.
2246
             00 43 1=2,15
2247
            .U(I)=U(I-1)+SELTAU
2248
       43
             w(I) = G(I-1)+DELTAG
      C
      С
             CRAW THE INTERSECTION CURVE
      C
2849
             CALL POINTS (UJWJBXJXJ14)
3252
             CALL POINTS (U) W) EY, Y, 16)
8251
             CALL POINTS (U, W, EZ, Z, 16)
0052
             DC 31 I=1,16
2853
             <(I) = x(I) * SQRT(*5)</pre>
2254
             Y(I)=Y(I)-X(I)+512
2255
             Z(I) = Z(I) - X(I) + 512
2256
       31
             CONTINUE
2557
             DG 33 I=1,15
2258
             DELX(I)=Y(I+1)-Y(I)
2959
        33
             DELY(I) = Z(I+1) - Z(I)
      C
2062
             CALL SUBP(6142+ISEC)
£261
             50 34 I=1,15
2646
             CALL APNT(Y(I),Z(I),,-4)
2E 63
       34
             CALL VECT(DELX(I),DELY(I),3)
2264
             CALL ESUS
2265
             RETURN
2255
             CKB
```

```
232
```

```
FERTHAN IV
                 V22.1-1
                                                                      PAGE 201
2231
             SUBROUTINE TEST1(X,Y,Z,BX,BY,5Z)
2002
             DIMENSION L(24) x(1) x(1) x(1) x(1) x(1) x(16) x(16) x(16) x(16)
                38X(1)38Y(1)3EZ(1)3C1(16316)3C2(16316)3C3(16316)
2023
             REAL L
      С
             SUBROUTINE TEST1
      С
             FUNCTION:
             THIS ROUTINE CALCULATES BX, BY, BZ BOUNDARY VECTORS OF
      C
             A PATCH DEFINED BY 16 X+Y+Z COGRDINATES.
             INPUT VARIABLES:
      C
      C
             X,Y,Z CCORDINATES DEFINING THE PATCH.
      C
             OUTPUT VARIABLES:
      C
             SAUBYUB DOUNDARY VECTORS DEFINING THE PATCH.
      С
      C
             GET THE LENGTH BETWEEN EACH SUCCESSIVE POINTS
             N=16
2254
2005
             CALL LENGTH(LJX,YJZ)
      C
             GET THE RATIO OF THE LENGTHES
2226,
             TL1=L(1)+L(2)+L(3)
2227
             TL2=L(4)+L(5)+L(6)
0228
             TL3=L(7)+L(8)+L(9)
2229
             TL4=L(12)+L(11)+L(12)
      C
             CALCULA E THE U VALUES
2612
             U(1)=Z.Ø
2211
             U(5)=0.0
2812
             L(9)=2.3
2213
             U(13)=0.0
2414
             L(4)=1.
2215
             U(8) = 1.
2216
             Ŭ(12)=1.
2217
             U(16)=1.
3.18
             U(2)≈L(1)/TL1
2319
             し(3)=(し(1)+し(2))/よし1
2222
             U(6)=L(4)/TL2
B221
             U(7) = (L(4) + L(5)) / TL2
2222
             U(12)=L(7)/TL3
2623
             む(11)=(も(7)+L(8))/TL3
             U(14)=L(10)/TL4
2224
2225
             U(15)=(L(10)+L(11))/TL4
      C
             W.R.T. W VALUES
      C
2226
             TL5=L(13)+L(14)+L(15)
2227
             TL6=L(16)+L(17)+L(18)
85 25
             TL7=L(19)+L(20)+L(21)
2229
             JL8=L(22)+L'(23)+L(24)
      ¢
      C
             CALCULATE THE W VALUES
      C
2232
             w(1)=0.0
4231
             a(2)=2.
0232
             k(3)=0.
3233
             H(4)=2.3
2 ม 34
             ¥(13)=1.
2835
             h(14)=1.
```

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FORTHAN IV
2635
             k(15)=1.∗
2237
             k(16)=1.
             h(5) =L(13)/TL5 *
3 L 38
             h(9)=(L(13)+L(14))/TL5
J£ 39
             k(6)=L(16)/TL6
2541
             ゅく12)=(しく16)+し(17))ノTL6
             h(7)=L(19)/FL7
w 2 4 2
2043
             ⋥(11)=(!(19)+Ľ(22))/TL?
2244
             %(8) =L(22)/TL8
2245
             h(12)=( (22)+L(23))/TL8
      C
            CONSTRUCT THE C MATRIX
             CALL CMATRY(U,W,C,N)
66.46
             DO 32 I=1 N
2047
1643
             DQ 32 J=13N
2249
       32
             (LeI) D= (LeI) 15
             CONSTRUCT THE B VECTORS
       C
1253
             DC 9 I= 3N
2251
             EX(I)=X(I)
2252
             (I)Y=(I)YE
2653
             &Z(I) =Z(I)
2254
             00 10 J=1.N
2255
             DG 10 I=1,N -
2256
             (LeI)12=(LeI)23
3657
             C3(I_JJ)=C1(I_JJ)
2258
        10
             CONTINUE
             SOLVE THE SYSTEM OF SIMULTANEOUS LINEAR EQUATIONS
             CALL SCLVE(BX)C1)NJ1J+222021,KS)
2459
             IF(KS.NE. D)STOP
2260
             CALL SOLVE(BY)CZ,N,1, 200001,KS)
2062
             IF (KS.NE.D)STOP
2263
             CALL SOLVE(BZ)C3)N313.3256019KS7
2065
             IF(KS.NE.C)STOP
2065
2603
             RETURN
3369
```

```
224
```

```
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                  v22.1-1
FORTRAN IV
2239
             SUBROCTINE DRWJGL(3x,8Y,8Z,1TGTO,MCN,NPL,x,Y,Z)
             DIMENSION BX(1), EY(1), BZ(1), X(1), Y(1), Z(1)
2222
             DIMENSION DELTAX(15),DELTAY(15),TOT(6),W(16),U(16).
2233
               JUL(16), WW(16), IDELX(15), IDELY(15)
             CCMMCN/DFILE/IBUF(1)
2:24
             COMMON/SUPER/GROJIP, IPO
0205
      C
      C
             DRWJCL.SUB
      ¢
             PURPOSE:
      C
             THIS SUPROUTINE DRAWS THE PATCH
      C
      C
      C
             ARGUMENTS:
       C
                   =ARRAY CONTAINING BX VECTOR OF THE PATCH
       C
             3 4
                                       BY
                                             , j.
       C
             ŝΥ
            ۵ī
                                       97
       С
             ITOTO = TAG OF THE SUBPICTURE CONTAINING PATCH
                                                                      (I/P)
       C
                   =<1 DRUSHP DRAWS LINES OF CONST. U.W PARAMETERS ONLY
       C
                                                                     AND PLOIS
                  =>2
       C
                         4 , ,
                                 ,,
                                                                      (I/P)
                   NEW CONTROL POINTS
       Ĉ
                                                                         (I/P)
       C
                     ENUBER OF PATCHES PER PLANE
       C
             ICN=2
1 98.92
             K = Ø
2027
              TOT(1)=2.0
2228
              TOT(2) = .2
2239
2213
              TOT(3)=+4
2011
              TOT(4) = 6
2212
              TOT(5) = 8
2213
              TCT(6)=1.2
       C
2214
              MEM = 0
8215
             . MEM1 =0
              DO 99 KOK=1,6
5616
              T=3.2
3217
              DO 40 I=1,16
2218
              W(I) = TOT(KOK)
2219
25 38
              U(I)=T
              T=T+1./15.
2221
        46
              CALL POINTS(U, W, 5x, X, 16)
6622
        122
              CALL POINTS(U,W,EY,Y,16)
2223
              CALL PCINTS(U,W,BZ,Z,16)
22.224
              DO 98 I=1,16
 2225
              x(1) =x(1)/SORT(2.)
 96.59
 2627
              Y(I) = Y(I) - X(I) + 512
              Z(I) = Z(I) - X(I) + 512
 2228
        92 .
              DC 91 I= 1515
 2229
              DELTAX(I)=Y(I+1)-Y(I)
 2633
              DELTAY(I)=Z(I#1)-Z(I)
 2231
              CONTINUE
 2832
              CALL APRICYCTOSZC10ss-40
 2233
                 92 I=1,15
 2234
```

PAGE L

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FORTRAN IV
                  VU2-1-1
2235
              XTIC+(I)XATJEC=(I)XATJEC
L236
              PRICH(I)YATJEC=(I)YATJEC
w437
              IDELX(I)=INT(DELTAX(I))
3233
              IDELY(I)=INT(DELTAY(I))
              CALL VECTODELTAXCID DELTAYCID 11
2239
6240
              DIFX = SELTAX(I) - FLOAT(IDELX(I))
2241
        9.2
              DIFY = DELTAY(I) - FLOAT(IDELY(I))
2042
             IF(MEM+GT+1)GO TO 999
2244
              00 121 I=1,16
             MEM=MEM+1
2245
2046
             MEM1 = MEM1+1
3647
             W(I)=U(I).
2248
        121
             U(I)=TOT(KOK)
5549
             GO TO 172
        999
3852
             MEM=Z
2251 ⊷
        99
              CONTINUE
2252
              IF (MON+NE+2)GD TO 1288
2354
             KW(1)=3.2
2255
              Wh(2)=+3333
2256
              WW(3)=+5666
2257
              hw(4)=1.2
2258
              DO 12 I=1,4
2259
             UU(I)=WW(I)
6263
              UU(I+4)=WW(I)
2261
              UU(I+8) =WW(I).
2662
             UU(I+12)=WW(I)
2263
             トト(【)=2.42
5554
              Wh(I+4)=+3333
2265
             ##(I+8)=+6666
2266
        10
              Ww(I+12)=1.0
2257
              CALL POINTS (UU WW BX X X 16)
22.68
             CALL POINTS (UU) WW, 3Y, Y, 16)
2269
              CALL: POINTS (UU) WW , BZ , Z , 16)
2273
             DO 25 I=1,16
2271
              X(I) = X(I) / SORT(2.)
$272
              Y_{\cdot}(I) = Y(I) - Y(I) + 512
2273
        20
              Z(I) = Z(I) - x(I) + 512.
              IPOPO=(ITOTO+1) x16+426.
2274
1275
              IF(NPL+GT+1)GO TO 1221
2277
             DC 52 I=1,16
             CALL SUBP(IPOPO+I)
2678
0279
             GALL APNT(Y(I),Z(I),1)
2282
        5 ż.
             CALL ESUB
2281
             GO TO 1000
        1221 DO 25 J=1,4
0082
2283
             DC 24 I=1.4
2084
             CALL SUBP(426+I+ION+(ITOTO-1)*4)
2635
             CALL APNT(Y(1+K))Z(1+K))1)
2236
        24
             CALL ESUB
2287
              K=K+4
2088
        25%
              ION=ION+8
        1000 RETURN
223<del>9</del>
2090
             ÈND
```

0

4

```
dir
```

```
PAGE 21
2221
             SUBROUTINE GETB(8x2,8y2,822,F8x,F8y,F8Z)
2022
             DIMENSION 8x2(1), 9x2(1), 9x2(1), Fex(4,4), Fey(4,4), Fez(4,4)
               38M(434)35MT(434)3FFX(434)3FFY(434)3FFZ(434)
      C
      C
             SUBROUTINE GETS
      ¢
             FUNCTION:
      ¢
             THIS SUBROUTINE CALCULATES THE 'B' MATRIX FROM THE 'S'
      ¢
      Ç
             VECTOR
                        S=(BM)(B)(BM)
                                           (EQUATION OF THE PARAMETRIC PC
      C
             INPUT VARIABLES:
      C
             BX2, BY2, BZ2 ARE ELEMENTS OF THE BOUNDARY VECTORS 'S'
      C
             GUTPUT VARIABLES:
      C
             FBX. FBY. FBZ ARE ELEMENTS OF THE BCUNDARY MATRICES &
      C
      C
             BM IS A COSTANT MATRIX, BMT IS THE TRANSPOSE OF EM MATRIX
      C
2223
             K = 1
2224
             CC 12 I=1.4
JE 25
             00 12 J=1,4
2225
             FBX(I_JJ)=BX2(K)
BL 27
             FEY([,J)=EY2(K)
2228
             FBZ(I_{J})=BZZ(K)
2029
        12
             K = K + 1
      C
      C
       Ċ
             DEFINE THE BM AND BM
                                     MATRICIES
2212
             5M(1,4)=1.
2211
             EM(2,1)=1.
2212
             BM(2,2)=1.
2213
             3×(2,3)=1...
6314
             EM(2,4)=1.
2215
             5M(3,3)=1.
2216
             3r(4,1),=3.
2217
             9M(4,2)=2.
2213
             EM(4,3)=1.
2019
             BMT(1,2)=1.
2220
             BMT(1,4)=3.
8821
             3KT(2,2)=1.
.Z022
             2MT(2,4)=2.
Ø223
             EMT(3,2)=1:
0024
             BMT(3,3)=1.
2025
             5MT(3,4)=1.
22 Zó.
             BHT(4,1)=1.
2227
             2MT(4,2)=1.
      C
      C
             GET THE B MATRICES(TENSOR) > CALL IT FBX > FBY > FBZ
      C
2823
             CALL MU(BM+FBX+FFX+4+4+4) o
2229
             CALL MU(FFX)BMT)FEX,454,4)
       Ċ
   ٠.
2232
             CALL MU(SMJFBYJFFYJ4J4J4)
32 31
             CALL MU(FFY)BMT,FBY,4,4,4)
```

FURTRAN IV

V22.1-1

FORTRAN IV

V22+1-1

PARE 23

0232 8233 С CALL MU(EM3FEZ3FFZ343434)
CALL MU(FFZ3EMT3FEZ343434)

8234 8235

(,

RETURN' END

```
SUPROUTINE POINTS(U) WIFFIXIN)
            DIMENSION UCNOWWOODECTADOXCHO
      C
      C
            SUBROUTINE POINTS
      С
            FUNCTION:
            CALCULATES THE XJOR YJOR 2 COORDINATES OF N FOINTS LYING ON
            THE PATCH SURFACE.
            UNW ARE INPUT ARRAYS DEFINING THE PARAMETRIC VALUES OF
            THE POINTS.
            B IS THE INPUT BOUNDARY MATRIX DEFINING THE PATCH.
            N=INPUT NUMBER OF POINTS.
            A=OUTPUT ARRAY CONTAINING COCRDINATES OF POINTS.
      C.
2223
            00 1 I=1,N
3234
            X(I)=(U(I)**3)*(w(I)**3)*2(1)+(U(I)**3)*(w(If)**2)
              *B(2)+(U(1)**3)*(W(1)*E(3))+(U(1)**5)*B(4)+(U(1)
              **2)*(V(I)**3)*B(5)+(U(I)**2)*(V(I)**2)*B(6)+(U(I
              )**2)*h(I)*B(7)+(U(I)**2)*5(8)+(U(I)*\[T])**3)*B(
                         9)+(U(I)*W(I)**2)*3(1@)+U(I)*W(I)*B(11)
              +U(I)*8(12)+(W(I)**3)*8(13)+(W(I)**2)*8(14)
              +W(I) #8(15) +8(16)
2225
            CONTINUE
2236
            RETURN
2227
            END
```

```
v 2 2 . 1 - 1
FORTRAN IV
             SUBROUTINE GETBX(FBX,FBY,FBZ,8X,BY,BZ)
0131
             SIMENSION BM(4,4),BMT(4,4),FEX(4,4),FBY(4,4),FBZ(4,4)
2222
                38x(16)38Y(14)38Z(16)3FFX(434)3FFY(434)3FF2(434)
       C
             SUBROUTINE GETBX
       C
             FUNCTION:
       C
             THIS SUBROUTINE CALCULATES THE 'BX' + 'BY + + 19 Z'
       C
             BOUNDARY VECTORS FROM THE FRX FRY FRZ (B MATRICIES)
       C
       C
             EM IS A CONSTANT MATRIX, SMT IS ITS TRANSPOSE
       C
                                       -1
       С
                                      MATRICIES
       С
              DEFINE THE SM AND SM
. 623
              EM(1,1)=2.
              EM(1,2)=-2.
2024
              EK(1,3)=1.
2225
              BM(134)=1.
2230
              5M(2,1)=-3.
2207
£ 233
              9M(2,2)=3.
              3K(2,3)=-2.
2639
              BM(2,4)=-1.
2212
              5K(3,3)=1.
£311.
              EM(4,1)=1.
2612
              5MT(1,1)=2.
2213
              SMT(2,2)=3. .
 2014
              BNT(3,3)=1.
 2215
              EMT(1,2)=-3.
 2216
              EMT(3,1)=1.
 2217
              EKT(1,4)=1.
 2013
              5MT(4+1)=1.
 3519
              5MT(4)2)=-1.
 3223
              SKT(3,3)=-2.
 JJ 21
              .5MT(2,1)=-2.
 2822
        C
               CALL MU(BM)FBX)FFX,4,4,4,4)
 2223
               CALL MU(FFX, BMT, FBX, 4, 4, 4)
 2224
               CALL MU(BM)FBY)FFY,4,4,4)
 2225
               CALL PUCFFY= 2MT = FBY =4 = 4 = 4 > 4 )
 3226
        C
               CALL MU(SMaFBZaFFZa4a4a4)
 3227
               CALL MU(FFZ,BMT,FBZ,4,4,4)
 2228
        C
               K = 1
 2029
               DO 12 I=1.4
 2233
```

DO 12 J=1,4 2231 SX(K)=FBX(I+J) 2832 BY(K)=FBY(I)J) 2233 EZ(K)=FBZ(IJJ) 2234 12 K = K + 12835 RETURN 2336 END

```
PAGE 021
FORTRAN IV
             · vaa.1-1
             SURROUTINE IDNTFY(NUM; ITO)
22 21
             COMMON/OFILE/IBUF(1)
2222
2233
             COMMON/SUPER/GRD
             COMMON/ERS/IERAS
2234
      C.
             SUBROUTINE IDNTFY
      C
      C
             FUNCTION:
             THIS SUBROUTINE IDENTIFY THE PATCH
             NUM=NUMBER OF DEFINED PATCHES(SUB-PICTURES) (I/P)
      C
                                                             (C/P)
             ITO=TAG OF THE IDENTIFIED PATCH
      C
             CALL ERASCIERAS)
2625
             CALL CMPRS
€686
             CALL SUBP(IERAS)
2227
             CALL APNT(2.0,2.2.2,,-4)
2228
             CALL ESUB
2229
             TURN LP SENSETIVTY ON FOR ALL DEFINED SUBPICTURES
       C
             DC 13 I=1 NUM
2212
            . CALL POINTR(9:1)
3811
             CALL SENSE(9,1)
        ں 1
12 £
             CALL TRAK(512.,512.)
2213
             TYPE *, IDENTIFY THE
                                      PATCH!
2214
·2 × 15
             kRITE(5,1)
               FORMATC TO DO SO POSITION THE TRACKING OSSET AT
2216
             WRITE(5,42)
2217
                FORMATCIOF THE PATCH AND TYPE CCRS WHEN
2218
        42
             READ(5,2)M
2019
2223
        2
             FORMAT(AZ)
             CALL MENUH(ITG: 1:NUM)
 2221
       C
       Ç
             TURN LP SENSETIVTY OFF FOR ALL DEFINED SUBPICTURES
              DO 32 I=1,NUM
 2222
              CALL POINTR(9)1)
 2223
              CALL SENSE(9,-1)
        32
 2424
            RETURN
 3225
 2226
              END
```

```
FORTRAN IV
                V22•1-1
                                                                     PAGE 221
2031
             SUBROUTINE INFILE(FILE1) FILE 2, NPATCH)
2222
             LOGICAL*1 FILE1(15) FILE2(15) JOSP(5) JOAT(5)
             DATA DSP/LetalotDistStelPletZ/CAT/LetalDistRichAlotTisZ/
8223
      Ç
      C
             SUBROUTINE INFILE
      C
             FUNCTION:
      C
             INPUT FILE NAME
      C
             NPATCH=NUMBER OF PATCHES TO BE SAVED OR RECALLED
      C
             FILE  IS THE DISPLAY FILE NAME
             FILE2 IS THE DATA FILE NAME
uC 34
             WRITE(5,30)
2225
             FORMATO! ENTER NUMBER OF PATCHES JOR SECTIONS;1)
2224
             READ (5,3020) NPATCH
2237
        3222 FORMATCIES
2205
             WRITE(5,11)
       1
2229
        11
             FORMAT( ! TYPE FILENAME: !)
2312
             READ(5,22)N,(FILE1(I), I=1,N)
111
             IF(N+EQ+0)GO TO 1
0213
        28
             FORMAT(Q36 A1)
2214
             DG 198 I=1.N
             FILE2(I)=FILE1(I)
2515
             SO 222 I=1.5
3816
             FILE1(I+N) = DSP(I)
2217
8155
        223
             FILE2(I+N)=DAT(I)
2219
             RETURN
2222.
             END
```

```
FORTRAN IV
```

```
SUBROUTINE CMATRX (Ush ) CON)
35 J1
             SIMENSION UC10 +WC10 +C(N+N)
2232
      C
             SUBROUTINE CMATRX
      C
             FUNCTION:
      C
             THIS SUBROUTINE CALCULATES THE C MATRIXCONTAINES U.W.
      C
             PARAMETRIC PROUDUCTS DUSING THE PARAMETRIC PC EXPANDED EQUATI
      C
                     U.W PARAMETRIC VALUES
                                                (I/P)
             N=NUMBER OF DATA PINTS (USUALLY 16) (I/P)
      C
       C
             00 8 I=1 N
3223
             C(I_11) = (U(I) + +3) + k(I) + +3
6024
             C(I_{3}2) = (U(I)**3)**(I)**2
DE 35
             (I)#<(5**(I)U)=(&(I).
2225
             C(I_34)=U(I)**3
2227
             C(I)5)=(U(I)**2)*W(I)**3
2223
              C(I,6)=(U(I)**2)*W(I)**2
8839
              C(I,7) = (U(I) * * 2) * W(I)
0213
              C(1,8)=U(1)**2
2211
              C(I,9)=U(I)*H(I)**3
2212
              C(I,10)=U(I)*W(I)*#2]
a213
              C(I:11)=U(I)*W(I)
2214
              C(I,12)=U(I)
2215
              C(I_{1}3) = W(I) **3
2216
              C(I)14)=W(I)**2
2217
              C(I,15)=w(I)
2318
              C(1,16)=1.
2219
              CONTINUE
B & 23
        3
              RETURN
 3621
 2222
              END
```

```
SUBROUTINE LENGTH(L, X, Y, Z)
2 - 21
2295
            REAL L
            DIMENSION L(1), x(1), y(1), Z(1)
2223
      C .
      C
            SUBROUTINE LENGTH
      C
            FUNCTION:
             THIS SUBROUTINE CALCULATES THE LENGTH BETWEEN
      C
             EACH SUCCESSIVE POINTS
      C
      С
             INPUT VARIABLES:
             X, Y, Z COCRDINATES OF INPUT POINTS
      C
      C
             OUTPUT VARIABLES:
             L IS AN ARRAY CONTAINING LENGTH BETWEEN EACH, SUCCESSIVE POINT
      C
DE 24
             00 1 I=1,3
             L(I) = SQRT((X(I+1)-X(I))**2+(Y(I+1)-Y(I))**2+
JL 95
            * ((Z(I+1)-Z(I))**2))
             CONTINUE
2526
             CC 2 I=1,3
EE37
             L(I+3)=SQRT((X(I+5)-X(I+4))**2+(Y(I+5)-Y(I+4))**2
5523
            # +(Z(I+5)-Z(I+4))##2)
82.29
             CONTINUE
2613
             DG 3 I=1.3
             L(I+5)=SORT((X(I+9)-X(I+3))**2+(Y(I+9)-Y(I+3))**2
3611
            * +(2(1+9)-Z(1+8))**2)
2212
             CONTINUE
             DO 4 I=1.3
2213
             L(I+9)=SGRT((X(I+13)-X(I+12))**2+(Y(I+13)-Y(I+12))**2
2214
            * +(Z(I+13)-Z(I+12))**2)
2515
             CONTINUE
2016
             J = 13
             DG 5 I=1.4
2217
             L(J)=SORT((X(I+4)-X(I))**2+(Y(I+4)-Y(I))**2
2218
            * +(Z(I+4)-Z(I))**2)
2019
             J=J+3
3222
             CONTINUE
2221
             J=14
2222
             DO 6 1=1 . À
             L(J) = SDRT&(X(I+3)-X(I+4))**Z+(Y(I+8)-Y(I+4)) **Z
0023
            * +(Z(I+8)-Z(I+4))**2)
2224
             J=J+3
             CONTINUE
Z & 25
             J=15
2226
            EG 7 I=1,4
2227
             L(J)=SQRT((x(I+12)-x(I+8))**2+(Y(I+12)-Y(I+8))**2+
6×28
            * (Z(I+12)-Z(I+8))**2)
2029
             J=J+3
2233
             CCNTINUE
             RETURN
2231
2232
             END
```

```
SUBROUTINE SOLVE(RJAJMINJEPSIJER)
2231
2222
            CIMENSION A(1) R(1)
      C
      C
            SUBROUTINE SOLVE
            FUNCTION:
            THIS SUBROUTINE SOLVES A SYSTEM OF LINEAR EQUATIONS
            USING GAUSS ELIMINATION.
            R IS THE M BY N MATRIX OF THE RIGHT HAND SIDE (DESTOYED)
            ON RETURN R CONTAINS THE SOLUTION OF THE EQUATIONS.
            A IS THE M BY M COEFFICIENT MATRIX(CESTOYED)
            M IS THE NUMBER OF EQUATIONS IN THE SYSTEM.
            N IS THE NUMBER OF RIGHT HAND SIDE VECTORS.
      C
            EPS IS AN INPUT CONSTANT WHICH IS USED AS RELATIVE TOLERANCE
            FCR TEST ON LOSS OF SIGNIFICANCE.
      С
      С
             IER IS THE RESULTING ERROR PARAMETER CODED AS:
             IER=2
      C
                           NO ERRGR
      C
                            NO RESULT
             IER = - 1 OR N
      C
      C
            *COHTAM
             SOLUTION IS DONE BY MEANS OF GAUSS ELIMINATION WITHE
            COMPLETE PIVOTING.
      C
2223
             IF(M)23,23,1
             IER=6
3844
2235
             PIV=2.0
2226
             M M = M + M
2227
             M = N = M
02.28
             DG 3 L=1 MM
2239
             TE=ABS(A(L))
2212
             IF(TE-PIV)3,3,2
2211
             PIV=TB
       2
2212
             I=L
2013
             CONTINUE
2214
             TOL = EPS * PIV
8215
             LST=1
2215
             CC 17 K=1.H
2017
             IF(PIV)23,23,4
2613
             1F(IER)7,5,7
2019
             IR(PIV-TCL)5,6,7
2222
             IER=K-1
       6
0021
             PI/VI=1./A(I)
3222
             J = (I - 1) / M
             I = I - J * M - K
0023
2224
             J=J+1-K
2025
             DO'8 L=KJNMJM
2226
             LL=L+I
8027 ·
             TB=PIVI*R(LL)
2223
             R(LL)=R(L)
2629 -
             R(L) = TB
0630
             IF(K-M)9,18,18
2631
             LEND=LST+M-K
2032
             IF(J)12,12,12
2233
             M×L=II
2234
             DG 11 L=LST,LEND
```

23

2631

2082

8833

6234

RETURN

IER =-1

RETURN

END

PAGE 222

ATRICIES .

S OF B.

K

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```
2021
            SUBREUTINE DATFIL (NPATCH)X, Y, Z)
21.32
             CIMENSION X(1) Y(1) Z(1)
             SUBRUTINE DATEFIL
             FUNCTION:
      ¢
             THIS SUBROUTINE RETURNES X . Y . Z COORDINATES OF A PATCH
      ¢
             OR A' GROUP OF PATCHES TO THE MAIN PROGRAM USING DATA FILES
      C
             APATCH=NUMBER OF PATCHES
                                           (I/P)
      C
      C
             SEFINE NUMBER OF DATA FILES (=NUMBER OF PATCHES)
      C
2003
             SYTE FILE(14)
2224
             WRITE(5)1)
2225
             FORMATCE
                         PLEASE ENTER THE NUMBER OF DATA FILES!)
        1
8626
             READ(5,2)NPATCH
       z^z
2027
             FORMAT(12)
      C
2233
             K = 2
2229
             DC 12 I=1, NPATCH
             56 20 J=1,14
2212
2211
       23
             FILE(J)=! !
2612
            TYPE 4,1
0213
             FORMAT( * ENTER THE NAME OF FILE NUMBER
2214
             ACCERT 3:FILE
0215
       . 3
             FORMAT(14A1)
2316
             J=I+9
2217
             CALL ASSIGN(J.FILE.0)
2213
             READ(J;*)(X(II+K);Y(II+K);Z(II+K);II=1;16)
2019
             K=16 # I
22.22
            · CALL, CLOSE(J)
             CANTINUE
2021
        12
2222
             RETURN
2323
             END
```

```
2221
            SUBROUTINE KEYBRD(NPATCH, X, Y, Z)
200Z
            DIMENSION X(1) Y(1) Z(1)
      C
      C
            SUBROUTINE KEYBRD
      Ç
            FUNCTIO:
      C
            THIS SUBROUTINE RETURNS THE X,Y,Z COORDINATES OF A PATCH
      C
            OR A GROUP OF PATCHES TO THE MAIN PROGRAM VIA THE KEY ROARD
            NPATCH IS THE NUMBER OF PATCHES (I/P)
2833
            wRITE(5,1)
2/34
            FCRMAT( PLEASE ENTER THE NUMBER OF PATCHES!)
       1
2225
            READ(5,2)NPATCH
દે\૨૩ ઇ
       2
            FCRMAT(12)
      C
            DEFINE X,Y,Z COORDINATES OF EACH PATCH
2227
            K = 3
2208
            DO 3 J=1, NPATCH
2229
            WRITE(5 4)J
            FORMATC! ENTER 16 XJYJZ COORDINATES OF PATCH, NUMBER 1312)
2213
       4
2011
            READ(5,5)(X(I+K),Y(I+K),Z(I+K),I=1,16)
C212
       5.
            FORMAT(3F10.4)
2213
             K=16*J
2214
       3
            CONTINUE
2215
            RETURN
2216
             CMB
```

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FORTHAN IV
                 v52•1-1
                                                                      PAGE 021
13 ي ۽
             SUBRULTINE INTREV(X,Y,Z)
2022
             DIMENSION X(1) + Y(1) + Z(1) + X X X (32) + X C(2) + Y O(2) + Z O(2)
8633
             CGMMCN/DFILE/ISUF(1)
2024
             COMMON/RECALL/IRECL
2135
             COMMON/SUPER/GROWIP JIPO
      C
      C
             SUBROUTINE INTREV
      C
             FUNCTION:
      C
             THIS SUBROUTINE RETURNES THE X3Y3Z COORDINATES OF A PATCH
      C
             TO THE MAIN PROGRAM USING LIGHT PEN
      C
             K = 2
2526
2227
             xx = 512.
2280
             CALL SUBF(IP)
2239
             CALL APNT(XXxXXxx-4)
2.12
             CALL VECT(XX,2.3)
3311
             CALL APNT(970.3460.33-4)
2212
             CALL TEXT(' Y')
8213
             CALL ESUB
2314
             CALL SUBP(IP+1)
2315
             CALL APNT(XX,XX,,-4)
2016
             CALL VECT(Z.Z.XX)
2217
             CALL APNT(505.)1220.)
2218
             CALL TEXT( Z')
2019
             CALL ESUB
2023
             CALL SUBP(IP+2)
2821
             CALL APNT(XX,XX,,-4)
2222
             CALL VECT(-XX3-XX)
6323
             CALL APNI(52.35.35.3-4)
2224
             CALL TE T( X )
3225
             CALL ESUB
22.26
             II=2
00.27
             KK=0
2323
             KKK = Z
2229
             CALL SUBP(IP+4)
2333
             CALL OFF (IP+4)
2231
             CALL ME 600.j850.j=102.j2920+iPj1PCSITION.j1DGNE1)
2232
             CALL ESUB
0033
             CALL SUBP(IP+3)
0034
             CALL OFF(IP+3)
223S
             CALL MENU(2.0)1220.,-122.,2000+1P, DEFINE WORKING PLANE!,
            * IDGNEI)
             CALL ES B
2236
       C
2237
        722
             CALL ON IP+3)
2238
             CALL MENUH(IT,2022+IP,2701+IP)
2239
             CALL OFF(IP+3)
2642
             GO TO (22,132), IT
2241
        22
             CALL TRAK(XX3XX)
```

IF(II.EO.4)II=0

```
PAGE 222
2244
             KK = KK + 1
JL 45
             II = II + 1
1.46
             wRITE(5:11)II
3247
       11
             FORMAT(1x, POSITION TRAK. OBJ. TO DEFINE SEC. NUMBER (1)11)
0.45
             WRITE(5,12)
2249
       12
            FURMATC' YOU SHOULD DEFINE 2 FOINTS IN THIS SECTION THEY!)
2253
             kRITE(5,14)
2351
             FCRMATC! WILL DEFINE THE AXIS OF REVOLUTION!)
        14
2252
             READ(5)21)M
0253
        21
             FORMAT(A2)
2254
             CALL LPEN(IH: IT1)
            IF(Ih.EQ.2.0R.IT1.LT.IP.OR.IT1.GT.IP+2)G0 T0 32
C 2 5 5
2.57
            CALL GRID(GRE, GRE)
27.53
             CALL TRAKXY(XE)YE)
. . 59
            IT1=IT1-IP+1
تدغك
             GC TC (182,282,320),IT1
E 2 61
      120 CALL APRICATIVE
2352
             CALL SUEP(IP+4+KK)
2263
             CALL OFF(IP+4+KK)
22.64
             CALL VECT(-XX3-XX33II)
3265
             CALL VECT(2.8,XX,,II)
2200
             GALL VE T(XX,XX,,II)
2267
             CALL VECT(2.23-XX3311)
8869
             CALL ESUS
2269
             GO TG 422
2070
       220 CALL APNT(X0,Y2)
SE 71
             CALL SUBF(IP+4+KK)
6272
            CALL OFF(IP+4+KK)
2273
            CALL VECT(-XX3-XX33II)
2274
            CALL VECT(XX)2003,11)
2275
           CALL VECT(XX,XX,,II)
2275
            CALL VECT(-XX,3.d,,II)
            CALL ESUB
2077
22.78
            GO TO 422
2279
       328 CALL APRICKE, YES
2032
            CALL SUBP(IP+4+KK)
Se 81
            CALL OFF(IP+4+KK)
2632
            CALL VECT(XX,E,Z,,II)
2283
            CALL VECT(2.2.XX...II)
3284 .
            CALL VE T(-XX,0,0,0,,11)
2235
            CALL VECT(2.23-XX3311)
2636
            CALL ESUE
2287
           CALL ON(IP+4+KK)
6233
            I=1+K
2639
            IF(II.EQ.2)I=5+K .
.2291
            IF(II.E3.3)I=9+K
2293
            IF(II.EC.4) I=13+K
2295
       1222 KKK=KKK+1
2296
            CALL CN (IP+4)
2297
            CALL MENUH (ITZ)2928+19,2931+1P)
2298
            CALL OFF (IP+4)
2299
            GO TO (522,762), IT2
2130
            CALL GRID(GRD,GRD)
```

FORTHAN IV

RETURN

E·ND

2127

GT OFF RU SHIP2

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244
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FURTRAN IV
                V62-1-1
                                                                  PAGE 621
20 21
            DIMENSION X(32),Y(32),Z(32),EX(16),B1(16),3Z(16)
2002
            DIMENSION TEX(16) TEY(16) TEZ(162 JEXG(64) JEYC(64)
              3BZC(64)3BXZ(16)3BYZ(16)3BZZ(16)3IET(4)38XXX(16)
              #3ZZZ(16)#TEX1(32)#TEY1(32)#TEZ1(32)#BYYY(16)#XOY(32)
            JBYSPLT(16)JBZSPLT(16)
2224
            REAL LAMALLIALL2
26.35
            COMMON/DFILE/IBUF(3002)
2006
            CCMMON/RECALL/IRECL
0037
            COMMON/SUPER/GRD JP JIPG
0043
            CCMMCN/SEC/ISEC
      C
            PROGRAM SHIP2.FOR
      С
      C
              PURPOSE:
      C
            THIS PROGRAM AIDS IN THE PRELIMINARY DESIGN OF SHIPS HULLS
      C
     ·C
              UTILIZATION:
      C
            START UP THE COMPUTER
      C
            TYPE RU:SHIP2
      С
      C
            ALY A. BADAWY
      C
            UNIV OF MCMASTER
      C
            MECH's ENGG.
           11 JUNE 1979
2229
            GRD=1.
3212
            NDEF = 2
2811
            NUM≃©
3012
            IP=121
2013
            IP0=426
0214
            IYOA=1
3615
            KEMO = Ø
2216
            IRECL=3
2217
           ISEC=2
      С
      C
            GET THE MAJOR DIMENSIONS OF THE SHIP
      C
3155
            WRITE(5,345)
2319
       345. FORMATC! PROGRAM SHIPZ WILL AID IN THE DESIGN OF SHIP HULLS!:
0422
            WRITE(5,33)
2621
       33
            FORMAT(1x, ENTER LENGTH, BREADTH, DRAFT, F, M, A, S+)
0022
            READ(5,44)L,B,D,F,M,A,S
ØØ23
       44
            FORMAT(7F7.2)
      C
      C
            BEGIN DESIGNING THE SHIP HULL
      C
ಪ೬24
            WRITE(5,120)
3025
            FORMATO X3/11
       120
                           NOW YOU ARE
                                              TO DESIGN SHIP HULL!)
      C
      C.
            SET UP DESIGN MENU
```

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115
```

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FORTERN IV
                                             V22.1-1
                                                                                                                                                                                    PAGE 222
0225
                                 CALL INIT(3200)
.227
                                 CALL SCAL (0.2,0.0,1+B,1+2)
2228
                                  CALL SUBP(11200)
2829
                                 CALL GFF(11222)
2233
                                 CALL MENUCE. BULL + BUL
                                       FORWARD BODY', DESIGN AFT BODY', DESIGN STERN', SMOOTH', ISA
                                       JIRECALL!J!REFLECT!)
2231
                                 CALL MENU(0.2)3,-6/5.,12013, MODIFY', ERASE', INTRSC',
                                     'SPLIT','DONE')
2032
                                 CALL ES E
2633
                   132
                                 CALL ON (11002)
                C.
                C
                                  WAIT FOR A MENU HIT AND BRANCH TO SERVE IT
                C
2234
                                  CALL MENUH(IT)12218,12822)
3 × 35
                                  CALL OFF(11000)
                 C
                 C
                                 MIDDLE BCDY
                 C
2236
                                  IF(IT.EQ.1)LL1=M
2238
                                  IF(IT.EO.1)LL2=F
                 C
                 C
                                  FORWARD BODY
2042
                                  IF(IT+EQ+2)LL1=F
0242
                                  IF(IT.E. .2)LL2=2.2
                 C
                 C
                                  AFT BODY
2244
                                  IF(IT.EG.3)LL1=A
2046
                                  IF(IT .E T. 3)LL2=F+M
                 C
                 C
                                  STERN BODY
2248
                                  IF(IT.EQ.4)LL1=S
2252
                                  IF(IT+E3+4)LL2=F+M+A
                 C
2252
                                  IF(IT.E0.5)GO TO 444
                                                                                                                                        ISMOOTH TWO PATCHES
2254
                                  IF(IT.E0.6)GO TO 544
                                                                                                                                        ISAVE DISPLAY
Ø656
                                  IF(IT.EC.7)GO TO 554
                                                                                                                                        IRECALL DISPLAY
2258
                                  IF(IT.EQ.8)GO TO 655
                                                                                                                                        IREFLCT DISPLAY
2263
                                IF(IT.EQ.9)GO TO 755
                                                                                                                                        IMODIFY A PATCH
2262
                                  IF(IT.EQ.10)GO TO 855
                                                                                                                                        IERAS A PATCH
Z£ 64
                                  IF(IT+EQ+11)GO TO 555
                                                                                                                                        ! INTERSECT
                                  IF(IT.EQ.12)G0 TO 955;
2266
                                                                                                                             !SPLIT
2268
                                  IF(IT.E0.13)GO TO 999
                                                                                                                             !EXIT
                 C
                                  BEGIN DESIGNING THE SHIP
2073
                                  CALL SHIPS1(X)Y>ZJNPATCHJNPLJLL1JLL2JLJBJD>KGXJYGY;
2071
                                  IP=IP+43
2272
                                  IPO=IPO -42
2273
                    5200 NOEF=NDEF+1
```

```
241
```

```
0274
               NUMBR(NDEF) = NPATCH
        C
        С
               CRAW THE SURFACE
        С
 2275
               IF(NFL.LT.2)GC TO 522
        C
 3277
              DC 221 NN=1, NPATCH
 @£78
              CC 222 K=1,4)
 6279
              K1=(K-1) +4 + NPATCH+4 + (NN-1)
 2583
              KK1=K1+NUM×16
 2351
              DO 222 J=1,4
 2352
              XREAL(KK1+J)=X(K1+J)
 2283
              YREAL(KK1+J)=Y(K1+J)
 2254
              ZREAL (KK1+J) =Z(K1+J)
 2205
              TEX(J+(Y-1)*4)=x(K1+J)
 2286
              JEY(J+( -1)#4)=Y(K1+J)
 2637
              TEZ(J+(K-1)*4)=Z(K1+J)
         222
 2233
              CALL TESTICTEX TEY TEZ . BX . BY . BZ)
 2 L 89
              DO 77 IEN0=1,15
 2293
              SXO(IENC+KEMO) = 9 X (IENO)
 ₹91
              BYO(IENO+KEMO) =8Y(IENO)
2692
        77
              SZOCIENO+KEMO) =SZCIENO)
2293
              KEMO=KEMO+14
2094
              CALL SUBF(IYOA+IRECL)
6295
              CALL DRWSHP(BX,BY,BZ,L,ITCTO,2,2)
3576
              CALL ESUB
3097
              IYOA=IYOA+1. .
2298
        241
              CONTINUE
2299
              NUM=NUM+NUMBR(NDEF)
3138
              GO TO 132
2121
        5 🕁 🕉
              J=NUK*16
2122
              30 11 I=1,1K
2123
             AREAL(I+J)=X(I)
2134
             (I)Y=(L+I)JA3FY
2125
             ZREAL(I+J)=Z(I)
2126
             TEX(I)=x(I)
2107
             TEY(I)=Y(I)
0128
        11
             TEZ(1)=2(1)
       С
3129
             CALL TEST1(TEX+TEY+TEZ+34,3Y,3Z)
        5 8 5
3112
             CC 88 IENO=1,16
2111
            -9XO(IENO+KEMO) =BX(IENO)
2112
             EYO(IENO+KEMG) =BY(IENO)
6113
             BZO(IENC+KEMO)=BZ(IENG)
2114
             KEMO=KEMC+16
2115
             CALL SUBP(TYOA+IRECL)
2116
             CALL DRUSHP(BX38Y3BZ,L,ITOTO,230)
2117
             CALL ESUB
2118
             IYOA = IY A+1
3119
       220
             CONTINUE
6123
             NUM=NUM NUMBR(NDEF)
2121
             GG TG 132
      C
```

FORTRAN IV

V22-1-1

- v22.1-1

```
SMOOTH TWO PATCHES AT A COMMON BORDER
      C
      C
      C
            . CALL SMOSHP(5x0)EYO)BZO)NPATCH, IET, BX2,BY2,B22,L, IFOL)
2122
             IF(IEOL+EO+1)NPL=2
£123
0125
             IF(IFQL+EG+2)NPL=1
             NP30 = IP0 - 48
3127
d123
             NP0=48
             NPLA = (IET(2) - 1) \times 16
2129
2132
             DC 144 I=1,16
             EXB(I+NPLA)=BX2(I)
J131
             EYO(I+NPLA)=EY2(I)
ũ132
3133
        144
             BZC(I+NPLA)=BZZ(I)
             CALL ERAS(IET(2)+IRECL)
∂134.
2135
             CALL CMPRS
             IF(NPL+GT+1)G0 TO 12
2135
0133
             DO 28 I=1,16
             CALL ERAS(NPOO+I)
2139
        23
2142
             CALL CMPRS
             GC TO 22
2141
             DO 25 I=1,4
        13
2142
             CALL ERAS(IPO-NPO+I+(IET(2)-1)+4)
2143
             CALL ERAS(IPO-NPO+I+8+(IET(2)-1)*4)
2144
             CALL ERAS(IPO-NPO+I+16+(IET(2)-1)*4)
2145
             CALL ERAS(IPO-NPO+I+24+(IET(2)-1)*4)
        25
2146
             CALL CMPRS
£147
             CALL SUBP(IET(2)+IRECL)
∕ ⊌143∙
              CALL DRWSHP(BX2,BY2,BZ2,L,IET(2),2,NFL)
2149
2152
              CALL ESUR
             GO TO 132
2151
       C
              SAVE THE DISPLAY AND CREATE A DATA FILE
       C
       Ċ
       C
 2152
        544
              CALL GSSAVE(XREAL)YREAL)ZREAL)
              GC TQ 132
 2153
       C
       C
              RECALL THE DISPLAY
       C
 2154
        554
              CALL GETSHP(XREAL)YREAL)ZREAL)
              GO TO 130
 0155
       C
              REFLECT THE DISPLAY ABOUT X-Y OR Y-Z OR Z-X PLANE
       C
        655
              ROL=LL2
 2156
 2157
              IS0=4
              IF(NPL.EG.2)ISO=8
 2158
 2158
              FACO=1.
              TEX1(ISD)=XOX(ISO)
 2161
              DC 1323 I=1, ISO
 3162
              FA=2.*((B/2.)-(YOX(I)-((L*SQRT(2.)-ROL)*SQRT(.5))))
 2163
        1000 XOX(I)=XOX(I)+FA
 2164
 3165
              SEKA=-TEX1(ISO)+XOX(ISO)
```

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PAGE 105
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FORTRAN IV
            ROL1=ROL+(LL1/3.)*SCRT(2.)
3155
            S6 2222 I=1;IS0
J157
             YCK(I)=XCX(I)-SEKA
2165
            FA=2.*((E/2.)-(YOX(I+1S0)-((L*SORT(2.)-ROL1)*SORT(.5))))
2169
       2007 XCX(I+ISO) =XOX(I+ISO)+FA-SEKA
2172
             ROL2=ROL+(2.*LL1/3.)*SGRT(2.)
2171
             SC 3100 I=1.ISO
1172
             FA=2.*((B/2.)-(X0X(I+2*ISC)+(C(L*SORT(2.)-R0L2)*SORT(.5))))
£173
        3220 X0X(1+2+180)=x0x(1+2+180)+FA-SEKA
2174
             ROL3=RCL+LL1*SQRT(2.)
2175
             EG 4022 I=1.ISO
J176
             FA=2.*((E/2.)-(XOX(I+3*ISC)-((L*SQRT(2.)-ROLE)*SQRT(.5)))
£177
        4220 X0X(I+3#ISO) =X0X(I+3*ISO)+FA-SEKA
2178 -
                          I=1,16*NPL
             58 5220
J179
             CALL APNT(XOX(I),YOY(I))
2182
             TEZ1(I)=YCY(I)-L+LL2*SQRT(.5)
Ø131
             TEY1(I)=XCX(I)-L+LL2*SQRT(.5)
3182
             TEX1(I) =(LL2*SGRT(*5))*SGRT(2*)
2133
             \chi(I) = TEX1(I)
2134
             Y(I) = TEY1(I)
 2185
        5230 Z(I)=TEZ1(I)
. 3135
             GO TO 6 ZØ
 원137
             MGDIFY PATCH
       C
             FIRST IDENTIFY THE PATCH TO BE MODIFIED
       C
             WRITE(5,762)
        755
 2150
             FORMAT(1x, POSITION TRACK OBJ. AT ANY PT. OF THE PATCH)
        762
 21.59
              WRITE(5,765)
 2193
              FORMAT( ' TO BE MODIFIED, TYPE (CR) WHEN DONE !)
 2191
        765
              CALL LPEN(IH; ITOTO)
         770
 2192
              IF(IH-E0.2.0R.ITOTO.LT.1.0R.ITOTO.GT.16)G0 TO 778
 2193
              READ(5,775) IY
 2195
              FORMAT(A2)
 2196
         775
              WRITE(5 777)ITOTO
 2197
              FORMATCIX. YOU HAVE JUST POINTED AT PATCH NUMBER
 £190
              CALL TRAK(LJL)
 2199
              WRITE(5 783)
 3222
              FORMATCIX, POSITION TRAK. OBJ. AT NEW PT.TYPE CCR. WHEN BONE
         762
 231
              READ(5,775)10
 2222
              CALL TRAKXY(X00,Y00)
 2223
              IPOS=1PO-4Z
 2234
              IPOP0=IP00+128
  2225
              CALL LPENCIK, ITT)
 0206
         785
              IF(IK.E0.Z.OR.ITT.LT.IPOD.OR.ITT.GT.IPOPO)GQ TC 785
  2207
              PX=ITT-IPOD
  2229
              WRITE(5,790) MX
  2212
              FORMAT(1x . YOU HAVE POINTED AT PT. NUMBER 1.12)
  2211
         792
              XOX(FX)=XCO
  8212
               Y0Y( PX ) = Y00
  2213
               Y(MX)=XCC+(X(MX)/SORT(2.))-L
  2214
               YREAL(MX+(ITOTO-NUMBR(NDEF)) #16) =XOC+(X(MX)/SCRT(2+))-L
  2215
               Z(MX)=Y00+(X(MX)/SQRT(2.))-L
```

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FURTRAN IV
                 VU2.1-1
                                                                     PAGE 276
2217
             ZREAL(MX+(ITCTO-NUMBR(NDEF))*16)=Y0C+(X(MX)/SCRT(2.))-L
2213
             IF(NPL.GT.1)GO TO 795
2222
            . MCNA=16*(ITOTO-1)
2221
             DC 822 I=1,16
6222
             MCNA1=MONA+I
J 223
             TEX(I)=XREAL(MONA1)
2224
             TEY(I)=YREAL(MONA1)
J225
       322
             TEZ(I)=ZREAL(MONA1)
2226
             GO TO 825
2227
       795
             00 810 K=1,4
2223
             K1=(K-1)*4*NPATCH+4*(ITOTC-NUM+NUMER(NDEF)-1)
2229
             00 812 J=1,4
2232
             TE < (J+(K-1)*4) = X(K1+J)
2231
             TEY(J+(Y-1)*4)=Y(K1+J)
             TEZ(J+(K-1) +4) =Z(K1+J)
2232
       811
3233
             CALL TESTICTEX, TEY, TEZ, 3x0x, 8Y0Y, 8Z0Z)
       325
2234
             ICK=(ITOTO-1)*15
235 كان
             DO 82 I K=1,1名
3236
             BXO(IAK+ICK) =BXCX(IAK)
2237
             SYO(IYK IOK)=BYOY(IAK)
2233
       δŴ
             BZO(IYK+ICK)=BZOZ(IYK)
2239
             CALL ERAS(ITOTO)
2242
             IF(NPL+GT+1)GO TC 9520
2242
             CO 932 I=1,16
2243
       936
             CALL ER S(IPOD+I)
2244
             GO TC 9522
2245
       9602 DO 9202 I=1,4
2246
             CALL ERAS(IPOD+I+(ITOTO-1)*4)
2247
             CALL ERAS(IPGD+I+8+(ITOTO-1)*4)
2243
             CALL ERAS(IPCD+I+16+(ITOTC-1)×4)
2249
       9200 CALL ERAS(IPGD+I+24+(ITOTO-1)+4)
2252
       9500 CALL CMPRS
2251
             CALL SUBP(ITOTO)
             CALL OR SHP(BX0X)BY0Y)BZCZ)L,ITOTO,2,NPL)
2252
2253
             CALL ESUB
2254
             GO TO 130
      C
             ERASE A PREDEFINED PATCH
      C
€255
       855
             WRITE(5,760)
2255
             WRITE(5,860)
2257
             FORMAT( TO BE ERASED TYPE (CR) WHEN DONE !)
       360
₽25ā∙
       37U
             CALL LPEN(IH; IERAS)
2259
             IF (IH.EO.Z.OR.IERAS.LT.1.OR.IERAS.GT.16)GO TO 372
2261
             IPOD=IPO-42
2252
             READ(5,775)1E
2263
             WRITE(5,777) IERAS
2264
            ·CALL ERAS(IERAS).
2265
             IF(NPL+GT+1)GC TO 9612
2257
             DO 852 I=1,16
2263
       362
             CALL ERAS(IPOD+I+(IERAS-1)*16)
2259
             30 TO 9515
2270
       9612 DC 9122 I=1.4
```

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FORTRAN IV
                  VU2.1-1
                                                                     PAGE CZ7
 2271
              CALL ERAS(IPO-42+I+(IERAS-1)#4)
 2272
              CALL ER S(IPC-43+I+8+(IERAS-1)*4)
 2273
              CALL ERASCIPC-42+1+16+(IERAS-1)*4)
 3274
        9133 CALL ERAS(IPO-40+I+24+(IERAS-1)+4)
 2275
       . 9512 CALL CMPRS
 £276
              GO TO 1.32
       C
             GET INTRSECTION OF A PATCH WITH A PLANE
 2277
        555
             WRITE(5,55)
2273
        55
             FORMAT( * POSITION TRACK . CBJ. AT PATCHITYPE CR > WHEN DONE !)
€279
        56
             CALL LPEN(IHO, IHE)
 2230
             IF(IHO.EC.2.OR.IHE.GT.16.0R.IHE.LT.1)GO TO 56
 2232
             READ(5-775) 1B
2233
             KRITE(5,777) IHE.
2234
             WRITE(5,5550)
        5552 FORMAT(1x) ENTER NUMBER OF SECTIONS!)
2235
2235
             READ(5,5562)NSEC
2287
        556% FORMAT(12)
2283 🗬
             DO 5660 I=1.NSEC
6553
             CALL INTRSC(BXO,BYO,BZO,L,B,IHE)
2290
             CALL ERAS(2142+ISEC)
2291
             CALL CMPRS
2292
        5660 ISEC=ISEC+7
€293
             GO TO 130
       C
             SPLIT OPTION
       C
. 2294
             WRITE(5,131)
        955
2295
             FORMAT(* POINT WITH THE L.P. AT THE PATCH TO BE SPLITTED!)
      131
2296
        132
             CALL LPENCIH'TTOTO
             IF(IH-E0-E-OR-ITOTO-LT-1-OR-ITOTO-GT-16)GO TO 132
6297
2299
             READ(5,775)IS
232Z
             HRITE(5,777)ITOTO
0321
             JSPLT=(ITOTO-1)*16
€332
             DO 133 I=1,16
2363
             BXSPLT(I)=BXO(JSPLT+I)
€324
             SYSPLT(I)=BYO(JSPLT+I)
2325
       133 333ZSPLT(Ib=BZO(JSPLT+I)
      C
            ►CALL SPLITTING ROPUTINE
      C
2326
             CALL GEN(BXSPLT)BYSPLT, BZSPLT, X, Y, Z, L, LL 2)
2327
             GO TO 6 22
      C
      C
             DONE OPTION
2333
       999.
            PAUSE
2329
             CALL FREE
2312
             END
```

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FORTRAN IV
                  V22 -1-1
                                                                     PAGE 221.
3 L 2 1.
             SUBROUTINE SHOSHP(BXX)BYY)BZZ,NPATCH, IET, EX?, EY2, EZ2, L, IFOL)
2 ت یا ت
             DIMENSION 8x1(16),8y1(15),9Z1(16),8x2(16),8y2(16),
               BZ2(16), BXX(1), BYY(1), EZZ(1), IET(2), FBx1(4,4)
               >FBY1(4,4),FBZ1(4,4),FBX2(4,4),FBY2(4,4),FBZ2(4,4)
2203
             SMOSHP . SUB
             PURPOSE:
             THIS SU, ROUTINE SMOOTHES TWO ADJACENT PATCHES ALONG
             COMMON BORDERS.
      C
      C
             ARGUMENTS:
      C
             BXX = ARRAY CONTAINING ELEMENTS OF BX VECTOR BEFORE SMCOTHING
      C
             êYY= ,,
                                                                    ,,
      C
             ني = 225
             NPATCH=NUMBER OF PATCHES.
      C
             IET = ARR AY CONTAINING THE TAGS OF THE TWO PATCHES TO BE SMOOT!
             BAZ=ARRAY CONTAINING ELEMENTS OF BX VECTOR AFTER SMOOTHING
      C
      ·? C
2234
                                               , 8Y
      K
      C
             BZ2= ,,
      ¢
             L=EENGTH OF THE SHIP
      C
                          SMOOTHING ALONG LONGITUDINAL BORDERS
                                                                       (0/P)
                                      TRANSVERSE BORDERS
                                                                  (0/P)
             DG 1 I=1,2
2636
             CALL IDNSHP(NPATCH; ITO; L)
2237
             IET(I)=ITO.
             DC 10 I=1,16
             BX1(I)= XX(I+(IET(1)-1)*16)
2229
2210
             PY1(1) = BYY(1+(IET(1)-1) *16)
2211:
             BZ1(I)=9ZZ(I+(IET(1)-1)*16)
2212
             DO 20 I=1,16
ZZ13.
           № 5X2(I)=BxX(I+(IET(2)-1)*16)
             2Y2(1)=BYY(I+(IET(2)-1)+16)
2215
             BZE(1) #BZZ(I+(IET(2)-1)*16)
       22
2216
             CALL GETB(BX1)9Y1)BZ1,F3X1,F3Y1,F8Z1)
2217
             CALL GETE(BX2,BY2,BZ2,FBX2,FBY2,FBZ2)
2218
             WRITE(5,600):
0±19
       .6:2
             FORMATCIX, IF YOU WANT TO SMOOTH ALONG LONG. BORDER TYPE
2623
             WRITE(5 700)
2221
       733
             FORMATCIX, ALONG TRANS. BORDER TYPE 215
2222
             READ(5,200)IFOL
2623
       352
             FORMAT(12)
2224
             IF()FCL.EO.1)GO TO 922
2026
             00 50 I=1,4
0e 27
             FBX2(I)1)=FBX1(I)2)
2223
             F5Y2(I,1)=F8Y1(I,2)
2229
             FEZZ(I,1)=FEZ1(I,2)
0230
             00 62 I=1.4
2231
             F2X2(1,3)=FBX1(1,4)
            #F245(1)3)=F841(1)4)
```

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FORTRAN IV
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22.33
        62
              FBZ2(I+3)=FBZ1(I+4)
2634
              GC TO 1022
              DC 55 1 /1.4
FEX2(1,1)=F9x1(2,1)
2535
        930
5235
2237
              FEY2(1, ) = FEY1(2,1)
              FEZZ(131) =FEZ1(2,1)
DO 65 1 1.4
2235
        55
2239
              FBX2(3, )=FBX1(4,1)
2242
              FEY2(3,1)=FBY1(4,1)
2241
8642
        65
              FEZ2(3,1)=FBZ1(4,1)
2243
        1002 CALL GETEX(FBX2)F3Y2)F3Z2)8X2)8Y2)8Z2)
2244
              RETURN
2845
              END
```

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FORTRAN IV
                   V22.1-1
                                                                      PAGE 221
              SUBROUTINE SHIPS1(X,Y,Z,NPATCH,NPL,1,LZ,L,E,S,XCX,YCY)
  -261
  2222
              REAL LIL11L2
  2233
              COMMON/RECALL/IRECL
  6034
              CIMENSION X(1) = Y(1) = Z(1) = XXX(32) = XC(2) = YO(3) = ZC(2) = XCX(1)
                       >Y3Y(1)
  2225
              CCMMON/DFILE/IBUF(1)
  4605
              CCMMCN/ UPER/GRD, IP, IPO
        C
        С
              SHIPS1.SUE
        С
        С
                PURPOSE:
        C.
              THIS SUBROUTINE AIDS IN THE DESIGN OF SHIP HULL
        C
              IT SETS THE WORKING PLANES WITHIN THE PRESCRIPED LENGTHES
        C
              OF EACH SECTION OF A SHIP .
        Ç
              IT RETURNES THE X3Y3Z COORDINATES OF A PATCH
        C
              TO THE MAIN PROGRAM USING LIGHT PEN
        C
        C
              ARGUMENTS:
                X=ARRAY CONTAINING THE X COORD . OF THE 16 POINTS DEFINIG
        C
                   THE PATCH
                                  (0/P)
                Y=0/P Y COORD. OF PATCH
                                                 (0/P)
              Z=0/P Z COORD. OF PATCH
                                               (C/P)
              NPATCH=NUMBER OF PATCHES
        C
              NPL=NUM ER OF PATCHES PER PLANE
        C
              L=LENGTH OF A SHIP
        С
              L1=LENGTH OF THE SECTION TO BE DESIGNED
        C
              L2=LENGTH FROM THE ORIGIN OF THE AXIS TO THE SECTION
        C
              BEBREADTH OF THE SHIP 5
        C
              D-DEPTH OF THE SHIP
                         CONTAINIG THE CORRESPONDING X COORD. OF THE PATCH
              X O X = ARRAY
                 ON THE TRT X-Y PLANE
                                            (C/P)
              YGY = ARRAY CONTAINIS THE CORRESPONDING Y'COORD. OF THE PATCH
        C
                 ON THE CRT X-Y PLANE
              35=CB/2.)
 6227
 2223 .
              00 = D
 2229
              FACTOR= 2.0 .
              IRECL=IRECL+IRECL #40
 6210
 0211
              WRITE(5,2)
 2012
              FORMATO! PLEASE ENTER THE NUMBER OF PATCHES!)
 2413
              READ(5/3)NPATCH
 2214
              WRITE(5,551)
        551
. 2215
              FORMAT( PLEASE ENTER NUMBER OF PATCHES PER PLANE!)
 2216
              READ(5,3)NPL
.-2017
         3
              FORMAT(12)
 0013
             .K = Z
       C.
 2019
              CALL SCAL (0.0,0,0,L+B,L+B)
 2223
              XX=L ·
 2221
              CALL SUBP(IP+IRECL)
 2022
              CALL APNT(XX,XX,,-4)
 2223
              CALL VECT(XX,2.0)
 2224
              CALL ESUB
```

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254
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FORTRAN IV
            CALL SUBP( fp+1+ IRECL)
2225
            CALL APMT (XX x X x x x - 4 )
6225
2227
            CALL VECT(2.8)XXX
0023
            CALL ESUB
2229
            CALL SUBP(IP+2+IRECL)
DJ 33
            CALL APNT(XXxXXxx-4)
2531
            CALL VECT(-XX)-XX)
2832
            CALL ESUB
            II = Z
21.33
8234
            kK = 2
£ 235
            KKK = Z
JL 36
            CALL SUBP(IP+4+IRECL)
            CALL OFF (IP+4+IRECL)
EL37
2233
            CALL MENU(0.,L,-100.,2900+IP+IRECL, PCSITION (,100NE))
             CALL ESUB
8239
            CALL SUBP(IP+3+IRECL)
2643
             CALL OFF(IP+3+IRECL)
£241
             CALL MENU(2.0)Ly-100.,2202+1P+1RECLy DEFINE WORKING PLANE'S
2242
            * IDONE!)
             CALL ES B
8843
2244
            DO 4 J=1, NPATCH/NFL 1
2245
            WRITE(5 5)J
            FORMAT( DEFINE PATCHES IN PLANE NUMBER 1,12)
2246
2247
             WRITE(5,6)
       6,
            FORMATC! --
2046
      C
           IF(KKK.EC.5.0R.KKK.EQ.9.0R.KKK.EQ.13.0R.KKK.EQ.17)KKK=KKK-1
2649
      7.2
             `IF(KKK•EQ.21.6R.KKK+EQ.25.0R.KKK+EQ.29.0R.KKK+EQ.37)KKK=KKK-
2851
             IF(KK.E0.4.0R.KK.E0.8.0R.KK.EG.12.0R.KK.E0.16)GO TO 721
2253
             IF(KK+E0+20+OR+KK+E0+24)GO TO 701
685S
             60 TC 702
2057
             DC 16 MEME=1 KK
       761
2258
             CALL OFF(IP+4+MEME+IRECL)
2659
       16
       732
             CALL ON (IP+3+IRECL)
2663
             CALL MENUHCIT, 2200+IP+IRECL, 2001+IP+IRECL)
2261
             CALL GFF(IP+3+IRECL)
£262
2863
             GG TC (20,130), IT
             CALL: TRAK(XX3XX)
2264
       22
             IF(II.EQ.4)II=2
2265
             KK=KK+1
2267
             II = II + 1
£03G
             WRITE(5)11)II
6669
8272
             FORMAT(1x) POSITION TRAK. OBJ. TO DEFINE SEC. NUMBER
3271
             READ(5,21)M -
2272
        21
             FCRMAT(A2)
             CALL LP:N(IH;FT1)
.2373
             IF(IH-E0-2-OR-IT1-LT-101-OR-TT1-GT-123)GO TO 32
2274
2076
             171=171-102
2277
             GO TO (120,200,300),IT1
             GO TO 328 *
2278
        120
                 GO TO 300
        226
2679
```

```
6283
        323
             CALL AP T(L-L2-FACTOR)L-L2-FACTOR),-4)
2231
            ( CALL SUBF(IP+4+KK+IRECL)
6232
             CALL OFF(IP+4+KK+IRECL)
2233
             CALL VECT(89,2.2) II)
             KALL VECT(2.2.00.11)
8884
             CALL VECT(-BBJ2+DJJII)
2 2 5 5
2236
             CALL VECT(D.C)-DD()II)
2337
             CALL VE T(83/12.30.00.03.3-4)
             CALL VECT(0.0,DD),II)
2235
2239
             CALL VECT(38/12.,2.2,,-4)
2292
             CALL VECT(0.2,-DD,,II)
             CALL VECT(38/12.32.2.2.3.3-4).
2291
2292
             CALL VECT(@.0)DDssII)
2293
             CALL VECT(83/12.30.2.1-4)
2294
             (IIieGC-c0+0)TOBV JAC
2295
             CALL VECT(88/10.,2.0,,-4)
2295
             CALL VECT(0.2,3D,,II)
2297
             CALL VECT(BB/10.32.03,3-4)
2598
             CALL VECT(2.2,-DD,,II)
2299
             CALL VECT(88/10.00.20.20.4)
2102
             CALL VECT(2.2.2DJ.11)
             CALL VECT(38/12.50.23.5-4)
2131
2102
             CALL VECT(3.0)-DD;;[1)
             CALL VECT(BB/10.20.20.20.4)
£133
2104
             CALL VECT(2.2.DD.,II)
             CALL VECT(88/10+92+03)-4)
8135
L146
             CALL VECTO2.2,-DD/10.,,-4)
             CILCONSCES (TOBY "JAD
.2107
             CALL VECT(@.2)-DD/12.j-4)
2138
2139
             CALL VECT(BB, Z, Z, J) II)
2110
             CALL, VECT(Z.2,-DD/12.,,-4)
2111
             CALL VECT(-BB-3.0.11)
             CALL VECT (0.03-DD/10.3-4)
2112
£113
             CALL VECT(BB,2.2,,II)
2114
             CALL VECT(2.0,-DD/10.,-4)
2115
             CALL VECT(-88,0.2,,II)
             CALL VECT(@.DJ-DD/10.JJ-4)
2116 ··
2117
             CALL VECT(BB,Z,Z,J,II)
2118
             CALL VECT(Z.Z.-DD/10...-4)
2119
             CALL VECT(-BB,0.2,,II)
2.1 2 2
             CALL VECT(9.2,-DD/12.,,-4)
6121
             CALL VECT(38,2.2.2.11)
             CALL VECT(2.2,-00/12.,,-4)
2122
2123
             CALL VECT(-38,0.0,11)
2124
             CALL ESUB
2125
        4 & C
             CALL ON IP+4+KK+IRECL)
             FACTOR=FACTOR+L1/3.
0126
2127
             I=1+K*NPL
2128
             IF(II.E0.2) I=1+4*NPL+K*NPL
2132
             IF(II.EQ.3)I=1+8*NPL+K*NPL
             IF(II.EG.4) I=1+12*NPL+K*NPL
0132
2134
        1.200 KKK=KKK+1
 ö135
             CALL ON (IP+4+IRECL)
```

```
v22.1-1
                                                                       PAGE 231
             SUBROUTINE INTRSC(9x)8y,8Z,L,8,1HE)
C & 21
             DIMENSION UC20) H(22) HXINTR(22) HYINTR(20) HXTR(20)
2232
               #BX(1) #EY(1) #BZ(1) #DEL x(19) #DEL Y(19) #BEX(16) #BY(16)
                JBBZ(16).
2223
             COMMGN/DFILE/IBUF(1)
2524
             COMMON/SUPER/GROJIP, IPC
≥025
             COMMON/SEC/ISEC
0226
             REAL L
       C
             THIS SUBROUTINE DRAWS THE RESULTANT CONTINUOUS CURVE
       C
             FROM THE INTERSECTION OF A PLANE AND A PATCH
2227
             C=L+8
2223
             CALL APRICC/8.JS.*C/8.JJ-4)
6229
             CALL SUBP(4140+ISEC)
8612
             CALL OFF(6140+ISEC)
ψί.11.
             CALL LVECT( . 25 * C . Ø . )
⊎212
             CALL LVECT(2.,.25*C)
2213
             CALL LVECT(-.25*C.0.)
JE14
             CALL LVECT(0.,-.25*C)
2215
             CALL .AP T(+4*C+5+*C/8+++-4)
             CALL TE TO UT)
2216
2217
            ™ CALL APYT(C/9+3+95*C33-4)
2213
             CALL TEXT(1-W1)
2619
             CALL APNT( •1 *C • 4 • 5 * C / 8 • • • - 4 )
ខន្តន្តខ្
             CALL TEXT('0,0') -
0 É 21
             CALL APNT(+3*C+4+5*C/2+++-4)
2622
             CALL TEXT('1)0')
2623
            · CALL APNT(+3×C++9×C+++4)
2024
             CALL TEXT(11:11)
2025
             CALL APNT(.1*C.99*C.)
2226
             CALL TEXT( 12,11)
2227
             CALL ES B
2228
             CALL GN(6147+ISEC)
       C
       C
             POSITION TRACKING OBJECT ON THE POINTS OF INTRSECTION
             CALL TRAK(L)L)
22.29
023B
             KEMO=2
2231
        1 .
             hRITE(5,11)
2332
        11.
             FORMAT(1x, POSITION TRAC. 08J., TYPE CR> WHEN CONE !)
2233
             READ(5,20)I
2034
        22
             FCRMAT(A2)
Ø 235
             IF(KEMO+GT+2)G0 TO 32
0£37
             CALL TRAKXY(XO,YO)
2233
             KEM0=KE 70+1
2239
             GO . TO 13
2843
        30
             CALL TRAKXY(XO1:YO1)
       C
      C
             CALCULATE(U1+W1)+(U20+W2d)
             CALA SUBP (5141+ISEC)
3241
2242
             CALL APNT(X0,Y00"
3243
             CALL LVECT((X01-X0))(Y01
```

```
FORTRAN IV
               * Va2.1-1
1244
             CALL ESUE
11.45
             L(1)=((S-(C/8.))/(.25*C)
2246
             L(1)=(Y0-(5.*C/8.))/(.25*C)
2647 .
             U(20)=(X01-(C/8.))/(.25×C)
3048
             k(22)=(YC1-(5**C/&*))/(*25*C)
2249
             DELTAU=(U(22)-U(1))/19.
2150
             DELTAH=(W(22)-W(1))/19+
J251
             0C 42 I=2,19
2252
             UCI) = UCI-1) + DELTAU
             W(I)=W(I-1)+DELTAW
2253.
        42
       C
      C
             SRAW THE INTERSECTION CURVE
2054
             DC 55 I=1,16
2255
             BEX(1)=BX(I+(IHE-1)*16)
2356
             68Y(I)=BY(I+(IHE-1)*16)
2257
        55
             EBZ(I)=BZ(I+(IHE-1)×16)
      ď
Z258
             CALL POYNTS (UJWJEBXJXINTRJ20)
0059
             CALL POINTS (U) W/EBY/YINTR/22)
0263
             CALL POUNTS(U,W,BBZ,ZINTR,20)
2861
             00 31/1=1,22
26.62
             XINTACID=XINTACID*SORT(.5)
2563
             YINTR(I)=YINTR(I)-XINTR(I)+L
             ZINTROID = ZINTROID - XINTROID+L
2264
6265
        31
             CONTINUE
2866
             DO 33 I=1,19
2267
             DELX(I)=YINTR(I+1)-YINTR(I)
2368
      . . 33. .
             DELY(I)=ZINTR(I+1)-ZINTR(I)
      C
2069
             CALL SUBP (6142+ISEC)
6273
             CALL APNT(YINTR(1) = ZINTR(1) = 3-4)
2271
             DG 34 ,I=1,19
             CALL EVECTODELY(I),DELY(I),3)
:2272
        34
2273
             CALL ESUB
```

RETURN

END

2274

```
6231
            SUBROUTINE GEN(BX)BY)BZ)X)Y)Z)L)L2)
2222
            2623
            DIMENSIÓN 8(2) 9 (2) 9 UU (16) 9 W # (16)
2234
            REAL LoL2
22.35
            COMMON/OFILE/IBUF(1) -
2266
            COMMON/SUPER/GRDJIPJIPO
            GET X,Y,Z COORD. OF THE PATCH GENERATED BY THE
             SUBDIVISION OF THE GIVIN PATCH
      C
            CALL IT X,Y,Z
             wRITE(5:3)
2637
             FORMAT(1x) 22HENTER U(1) JU(2) JH(1) JH(2) VALUES)
2223
2229
             READ(5)*)U(1);U(2);W(1);W(2)
      C
2212
             SELU=(U(2)-U(1))/3.
11 تا تا تا
             DELW=(W(2)-W(1))/3.
2212
             しし(1)=8(1)
             UUC2)=UC1)+DELU
2513
             UU(3)=U(1)+2**DELU
⊿214
2015
             UU(4)=U(1)+3.*DELU
2216
             30 1 I=1,4
             wk(I)=W(1)
2217.
             WW(I+4)=W(1)+DELW
2618.
             ##(I+3)=k(1)+2•*DELW
2219
             WW(I+12)=W(1)+3.*BELW
8223
       1
2021
               12 I=153
2222
             ひい(1+4×1)=UU(1)
             UU(2+4* )=UU(2)
2223
0224
             บบ(3+4×I)=00(3).
2225
       16
             じじ(4+4×I)=じじ(4)
2226
             CALL POINTS (UU) WW BX x X x
BE27
             CALL POINTS (UUJWWJBYJYJ16)
             CALL PRINTS (UU + W + BZ + Z + 16)
22.28
2229.
             DC 23 I=1,16
2.30
             X(I)=X(I)/SORT(2.)
Z 2 31
             Y(I) = Y(I) - X(I) + L
3832
       22
             Z(I)=Z()-X(I)+L
                                              11111
2233
             IPOPC=IPO-43
34 نىڭ
             DO 30 I=1,16
             CALL ER S(IPCPO+I)
2235
             CALL SU P(IPOPO+I)
2236
             CALL APNT(Y(I),Z(1),1,8)
6237
             CALL ES'B
0238
        34
2239.
             CALL CYPRS
2243
             DC 48 I=1,16
             2(I)=Z(I)-L+L2*SGR·T(+5)
2541
             Y(I)=Y(I)-L+L2*SQRT(*5)
2242
             x(I)=(L *SORT(*5))*SORT(2))
2.043
        4 2
2244
             RETURN
2045
             END
```

```
PAGE D2
FCRTHAN IV
                 V22.1-1
2001
             SUBROUTINE IDNSHP(NPATCH; ITO; L)
            REAL L
2233
            COMMON/DFILE/IBUF(1)
2234
             CCMMCN/ UPER/GRD
      C
             IDNSHP. JUB
      C
      C
             PURPOSE:
      C
             THIS SU ROUTINE IDENTIFY THE FATCH
      C
      C
             ARGUMENTS
             NPATCH=NUMBER OF PATCHES
      C
                  =TAG OF THE SUBPICTURE OF THE HITTEN PATCH
                                                                   (0/P)
      Ċ
                   =LENGTH OF THE SHIP
2025
             DG 12 I 1,2
             CALL POINTR(9,1)
2625
             CALL SENSE(9,1)
2307
8528
             TYPE **! IDENTIFY THE
             CALL TR K(L)L)
2639
             WRITE(5,1)
2213
               FORMA ( TO DO SO POSITION THE TRACKING SEJECT AT ')
2611
             WRITE(5 42)
2212
              FORMATCIANY PART OF THE PATCH AND TYPE (CR) WHEN DONE!)
2213
       48
             READ(5,2)M
0214
2015
            FORMAT(A2)
             CALL ME UHCITO, 1, 20)
2216
             00 30 I=1.2
2317
2218
             CALL POINTR(9)1)
             CALL SENSE(9,-1)
8219
       3₽
2322
             RETURN
0021
             END
```

SUBROUTINE SOREV(U,W,5,D). 5521 2532 DIMENSION B(16) THIS SUB. CALCULATES D**2V/DUDK 2233 D=(3.*U**2)*(3.*N**2)*8(1)*(3.*U**2)*(2.*N)*D(2)+ (3.*U**2)*B(3)+2.*U*(3.*W**2)*B(5)+(2.*U)*(2.*W)*B(6) +2.*U*B(7)+(3.*k**2)*B(9)+2.*k*B(12)+B(11) 2234 RETURN 6225 CN3

FAGE

0221 ⊵ ∄ 2 C C C С C _ C

SUBROUTINE CURVPT(U) W. P. JUP. X. N.) DIMENSION U(N), W(N), E(16,3), X(N)

SUBROUTINE CURVET

FUNCTI NT. CALCULATES THE X3OR Y3OR Z CCORDINATES OF N FOINTS LYING UN THE PATCH SURFACE. USW ARE INPUT ARRAYS DEFINING THE PARAMETRIC VALUES OF

THE POINTS. B IS THE INPUT BOUNDARY MATRIX DEFINING THE COEFFICIENTS OF N=INPUT NUMBER OF POINTS &

K=CUTPUT ARRAY CONTAINING CCCRDINATES OF POINTS.

ALY SADAKY

OCT 1379

00 1 I= 1x1 $\chi(I) = (U(I) + 3) * (u(I) * 3) * E(1 * 3) + (U(I) * 3) * (u(I) * 3)$ (19U)+(9U e4)8K(5**K(19U)+((9Ue5)8*(19W)*(5#*(19U)+(9Ue5)6* I)+(4U.c)+(4U.c)+(5×4(1))+(5×4(1))+(5×4(5×4(1))+(5×6×4(1))+(5×4×4(3(9)JP)+(U(I)+k(I)+k2)+5(10)JP)+U(I)+k(I)+3(11)JP) +U(I)*E(12;JP)+(W(I)**3)*E(13;JP)+(W(I)**2)*E(14;JP)

+W(I) #E(15% JP) +B(16% JP)

CONTINUE RETURN END

```
PAGE 721
                  V22.1-1
FORTRAN IV
             SUBROUTINE DRAWF(EXJEYJPZ)
2001
8232
             ##ENSIGN #X(16)35Y(16)35Z(1 )3X(16)3Y(15)3Z(16)
             DIMENSION DELTAX(15) JUELTAY(15) JTOT(11) JR(15) JU(16)
2003
4004
              COMMON/DFILE/IBUF(1)
      С
      C
              THIS SUBROUTINE DRAWS THE PATCH
       С
4235
              TOT(1)=::02
3325
              TCT(2)=1
3227
              FCT(3)=02
1223
              TCT(4)=.3
2.39
              TCT(5)=.4
              TOT(6) = 5
              TOT(7) = \epsilon
              TOT(E) = 7
L E 12
2243
              TOT(9) = 8
              TOT(10)=+9
2214
2815
              TCT(-11) =1.2
JJ16
              MEM=2 .
              MEM1 =2
0117 ·
              DO 99 KOK=1,11
4813
2219
              7=3.2
              30 40 I=1.15
8223
2221
              #(I)=LCL(KØK)
              U(I)=T
2322
2323
              T=T+1./15.
        40
             CALL POINTS (U.W.BX.X.16)
2224
        1.3
              CALL POINTS.CU.W.BY.Y.15)
2225
              CALL POINTS (U.W.BZ.) 16)
DO 90 I=1:16
-2625
2227
2228
              x(I) = x(I)/SGRT(2 \cdot )
2229
              Y(I) = Y(I) - X(I)
2239
             \cdot Z(I) = Z(.) - X(I)
        92
              CC 91 I=1,15
2...31.
2232
              DELTAX(I)=Y(I+10-Y(I)
0233
              DELTAY(I)=Z(I+1)-Z(I)
              CONTINUE
2234
        91
0235
              Y(1) = Y(1) + 512
              Z(1)=Z(1)+512.
0036 ·
              CALL APRICY(1), Z(1), 3-4)
2237
2033
              DC 92 I=1,15
              CALL VECTOBELTAX(I) CELTAY(I))
2039
         92
              IF (MEM . GT . 1) GC TC 999 .
2242
              DC 121 =1-16
2242.
              REMEMEM+1
2243
              PEM1=MEM1+1.
8244.
              h(I)=U(I)<
2245
              n(I) = LOT(KOK)
2245 .
         121
              GC TG 122-
 2847
         999
2843
              MEMES
 2249
         99
              CONTINUE
 3150
              RETURN
 2251
              END
```